

August 19, 1957

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AVIATION CALENDAR

Aug. 19-23—Dynamic Engineering Conference, Boulder (Cells) Laboratories, National Bureau of Standards

Aug. 18-22—Brazilian Semiotic International
Semiotic Conference, Albany, N. Y.

Aug. 18-21—Western Electrical Works & Convention, Cow Palace, San Francisco, Calif.

Aug. 18-20—Gas Diamane Symposium
Transport Properties in Gases at High
Temperatures and Pressures, Techni-
cal Institute, Northwestern Univer-
sity, Evanston, Ill.

Aug. 18-20—Pacific General Meeting, Assoc. for Institute of Electric Engineers, Pacific Div.

Aug. 13-Sept. 1—Midget Asphalt Run, Ft. Wayne, Ind.

Sept. 1-15—Sixth International Astronomical Conference. Royal Astronomical Society and Institute of the Astronomical Sciences, Edinburgh and London, England.

Sept 18-2017 Flang Daplan, Society of
Swedish Account Coordinators, Farnborough,
UK, England

Sept. 1984-11th General Assembly, International Union of Geodesy and Geophysics, in conjunction with International Geophysical Year, University of Toronto, Canada.

Sept. 7—Second Annual Convocation of the OCS Club, Hotel Phillips, Kansas City, Mo.

Sept. 8-13—Second Annual Course on Investment Castings. Massachusetts Institute of Technology, Cambridge, Mass.

Sept. 9: Annual General Meeting, International Audubon, Cleveland, Ohio.

Sept. 21—Third Purple Arrow National Meeting, American Society for Testing Materials, Houston, Texas. Held at the Sheraton Hotel.

Sept. 14-15—Meeting of Antique Argyle

Sept. 18-1957 Caden Pave and Flying Display, Royal Aeronautical Society, W'n

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AVIATION WEEK • AUGUST 19, 1997

Vol. 47, No. 3

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Published: June and July 2004 in Dallas, Texas.

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(Continued from page 5)

Sept. 19-23—Fall Meeting American Society of Mechanical Engineers, Steiner Hotel, Hartford, Conn.

Sept. 15—Fleet & Whitney Aircraft Engine Maintenance and Operation Forum, sponsored by Southern Aircraft Co., Midvale Hotel, Dallas, Tex.

Sept. 16-27—19th Nucleonics Acoustics Conference, jointly sponsored by University of Michigan, Transportation Research Institute, Wayne State University, and The Aero Club of Michigan, Agona, Mich.

Sept. 20-23—North Central Regional Civil Aeronautics Council Air Panel, Toms River Hotel, Kansas City, Mo.

Sept. 30-Oct. 5—Nucleonics Acoustics Meeting, Aircraft Production Forum & Aircraft Engineering Display, Society of Automotive Engineers, Hotel Ambria del Las Vegas.

Oct. 1-4—19th Annual Meeting and Forum, National Business Aircraft Association, Comstock Hotel, Denver, Colo.

Oct. 1-7—First Annual Aircraft Symposium, Society of Experimental Test Pilots, Beverly Hilton Hotel, Los Angeles.

Oct. 7-10th Annual National Electronics Conference, Chicago, Ill.

Oct. 7-10th—19th Symposium, Langley Research Laboratory, Cleveland.

Oct. 7-10—Light Aircraft Congress, International Association of Manufacturers, Boston, Mass. For details write: IAT, 31 Lovell Rd., Concord, Mass.

Oct. 8-10th Annual Support Development and Operations Conference, Chesham Hotel, Stevens, N. Y.

Oct. 9-11—National FID Convention Section for Experimental Stress Analysis II, Castle, Hawthorn, San Diego, Calif.

Oct. 10-11—National Noise-Mitigation Symposium, Sheraton Hotel, Chicago, Ill.

Oct. 11-21—Canadian Aeronautical Institute Institute of the Aeronautical Sciences Meeting, Montreal, Canada.

Oct. 21-25—Conference on new development in the field of power, American Society of Mechanical Engineers, American Hotel, Allentown, Pa.

Oct. 24-25—Fourteenth Annual Display, Air Force Technical Association, Sheraton Hotel, Seattle, Wash. Pacific Auditorium.

Oct. 26-31—Second Winter Meeting, American Nuclear Society, Sheraton Hotel, New York, N. Y.

Oct. 20-23—Third Annual Meeting, Association of the U. S. Army, Sheraton Park Hotel, Washington, D. C.

Oct. 20-23—Annual East Coast Conference on Aeronautical and Nonaeronautical Structures, Fifth Regiment Armory, Baltimore, Md.

Oct. 26-30—National Industrial Packaging & Handling Exposition, Atlantic City Convention Hall, N. J.

Oct. 30—Aviation Electrical Equipment Display, U. S. Coast Hotel, San Diego, Calif.

Nov. 5-9—John M. White Industry Guided Missile Reliability Symposium, limited to those with score security clearance, Naval Air Station Test Center, Ft. Meade, Md.

Nov. 6-8—Third Annual Symposium on Aeronautical Communications, Sheraton Hotel, Dallas, Texas, N. Y.

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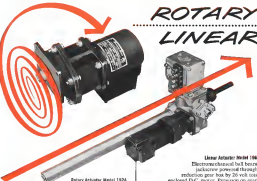
Allis-Chalmers also builds transformers, control, switching, motors, and amplifier generators for use in wind-tunnel and test facilities. For information, contact your nearby A-C office or write Allis-Chalmers, Industrial Equipment Division, Milwaukee 1, Wisconsin.

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Cable load structured to a 40 ohm level • Voltage
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Connectors

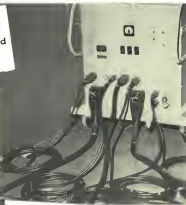
All type A and B
Genie DPST
21 Model 107 attachment
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Each with mechanical
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Shield cables: Four three cable diameter, retaining
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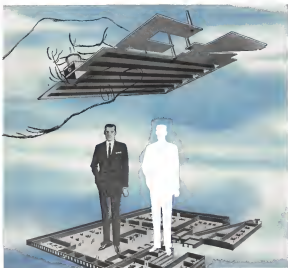
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the circuitry requirements, pre-
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EDITORIAL Cleaning Up the Pentagon

There is a new crop of jokes circulating in Washington these days on the fact that one of the nation's leading soap manufacturers, Neil McElroy, president of Procter & Gamble, is succeeding Charles E. Wilson as Secretary of Defense. Main point of the gags is whether an Ivey Soap salesman will be able to "clean up" the current Pentagon mess. There is little doubt among experienced Pentagon observers that Mr. Wilson is leaving a mess in the Pentagon that would tax the talents of the readiest "clean up" man who could possibly succeed him. And in evaluating Mr. McElroy's performance in the years ahead, it will be necessary to bear in mind the ratatouille and the misgratule of the legacy left by Mr. Wilson.

There have been significant changes in the Pentagon during Mr. Wilson's regime—the longest of any Secretary of Defense since the post was created in 1947. Most significant has been the changing mix of American armor versus Soviet armor. At the beginning of Mr. Wilson's regime, U.S. armorers enjoyed an unquestionably significant superiority over Soviet armorers and as a result was a major support of U.S. foreign policy.

Today, top Pentagon brass admit that the Soviets have surpassed U.S. armorers in the quantity of operational jet aircraft and only agree about the narrowing degree of U.S. qualitative superiority in aerial weapon systems. Mr. Wilson and his policies consist of course to be blamed for the entire state of this affair as much of it is accounted for by the tremendous drive the Soviets have made in acquiring their armor potential across the board, from helicopters and transports to intercontinental bombers and ballistic missiles. However, in the face of overwhelming concrete evidence of this Russian escalation, Mr. Wilson has administered policies that have had the effect of reducing both the quantity and quality of U.S. armorers, both now and for years to come. Even if armadillo and complete antibiotics were administered to the Wilson policies it would still take several years of (impractical) hot time for armorers to fully recover from their effects.

While Harold Stassen, the President's special agent on disarmament, and Secretary of State John Foster Dulles have been negotiating with Soviet delegations on disarmament, the Defense Department under Mr. Wilson has been engaged in unilateral disarmament that will, unless it is quickly checked, steadily deteriorate the U.S. position at the international bargaining tables.

The cause for this unilateral disarmament that is materially reducing both the size and effectiveness of U.S. armorers is financial. The Defense Department policies during Mr. Wilson's regime have increasingly placed more emphasis on field savings and less emphasis on forces sufficient to meet U.S. requirements. Mr. Wilson himself admitted this with his characteristic candor at a recent Pentagon press conference.

However, one of the major causes of the mounting defense expenditures with little commensurate increase in military effectiveness has been the tripling of weapons development, particularly in the missile field, that gave to the line of a major crisis during the tenure of Mr. Wilson's Pentagon stewardship. Congressman George Mahon, of Texas, who heads the military subcommittee of the House Appropriations Committee, put

his finger on the pulse of this problem when he recently said, "We can't afford any longer to equip all the services to fight the whole war by themselves." No responsible defense expert in Congress or the Pentagon believes that the answer to an adequate modern defense system is simple more money. It is becoming increasingly clear that the answer is to get \$38 billion more of genuine defense capability from the \$38 billion that is being spent already. Few would seriously claim that the American people are getting their money's worth in defense now.

The growth of tripling in weapons development has swollen the defense budget abnormally and the stop and start procurement and development programs that result from the frantic efforts to keep under a fiscal ceiling have wasted billions of the defense budget without producing any usable weapons. Boats and mowers of the three military services are essentially exchanged for those scrapped at the Key West conference in 1968 although none of the services have read more than lip service to these definitions. The most an economic mechanism forced by the immediate fiscal problems are also producing a major imbalance in the armorers program. For example, the intercontinental ballistic missile program is still going full blast with top priority, overtime and other measures that have run its cost to nearly double the original estimates in order to get an operational weapon at the earliest possible date. At the same time, the development of an aerial logistics system, vital to the successful war and peacetime operation of ballistic missiles, has been completely abandoned and all effort in this field stopped on the grounds of insufficient funds. Many Pentagon missile experts are still trying to figure out how they can field, operate and support a ballistic missile weapons system without the necessary aerial logistic system.

In a related armor field connected to the problems cited above, Mr. Wilson's regime has contributed to "backsliding" the Pentagon record without much results to show for the additional money. The office of the Secretary of Defense has grown in true bureaucratic fashion during the last four years from a small, advance type group to a gigantic super wall lined with dozens of redundant secretaries of defense who now constitute one of the major bottlenecks in producing a Pentagon decision.

As a result of the financial problems created by the tripling of weapons development and overhead growth, the armorers program has been cut back and stretched out with the stop and starts of research and development significantly retarded. For more money we will get fewer aerial weapon systems later than the normal pace of technological development could provide them.

There are the kind of problems Mr. McElroy will face in becoming the new Secretary of Defense. He will need all of the help he can get from the Congress, the military, the executive branch of the government and the American people if he is to effectively tackle and solve them. Much of the future safety of this country will depend on how successfully he makes this effort and the degree of support he gets from all concerned.

—Robert Holt

THERMISTOR SYSTEMS PLAYING BIG ROLE IN AIRCRAFT APPLICATIONS

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**CONTROLS TEMPERATURE
...PRECISELY**

WHO'S WHERE

In the Front Office

Alan Joseph J. Clark (USN, ret.), a director, **Alida Airlines, Inc.**, 1000 E. 9th, Ott., president, **Jet-O-Vac Co.**, Melrose, Wis. Mr. Ott includes **Donald W. Tyrrell** who continues as board chairman.

Fred Hensley, president, and **William J. McGowan**, general manager, **Pratt & Whitney Corp.**, West Coast Division, P.O. Box 1000, Los Angeles, Calif. **Li-Don Fung**, W. Tschelchke (USA), (ret.), corporate vice president, **Northrop Aircraft, Inc.**, Hawthorne, Calif. **Arthur H. Jones**, vice president defense planning, **Armstrong Manufacturing Corp.**, New York, N.Y.

Donald L. Dossel, vice president-coordinator and sales, **Tremont Engineering Corp.**, Tulsa, Okla.

Donald J. McGinnis, vice president-coordinator and sales, **Thompson Corp.**, Detroit, Mich. **Robert J. Lang**, a vice president, **Aircraft Engineering & Maintenance Co.**, Oakland, Calif.

William F. Hinchman, vice president, newly established **Advanced Products Division**, **ACP Industries, Inc.**, Milford, Pa. **Capt. Hamilton G. Hawks**, assistant, **Guided Missile Division**, **Naval Air Station**, **USN**, **Naval Air Station**, **D. C. Capt. Harold**, records, **Col. Robert E. Gale** (USMC), record.

Honors and Elections

Edward F. Corle, honor award prize dental assistant in charge of the **Aircraft Facilities Planning Group** has been named 3d DDC of 1957.

The **512th Fighter Interceptor Squadron**, being out of **RAF** station **Wormsley**, **Great Britain**, has been awarded the **Outstanding Squadron** by the **RAF**. **Tom and Gert Thomas**, **D. White**, **USAF**, **Chief of Staff** presented the **Wingard Achievement Award** to the squadron commander, **Lt. Col. John I. Rostrom**, at the seventh annual session of the **RAF Fighter Association** in **Washington, D. C.**

Changes

Donald B. Ott, director of planning, **William J. McGowan**, director, **International Business Machines Corp.**, New York, N.Y. **Charles B. Dossel**, deputy group executive, **Defense Products Corp.**, **Armstrong & Family Co.**, **Alameda, Va.**

E. E. Hensley, assistant chief engineer, **administration**, **Division of General Dynamics Corp.**, **New Design**, **Col. Richard F. White**, records, **Mr. Hensley** as chief design engineer, **Alan L. E. Ott**, **3-107** stress project engineer.

Henry M. Stupich, manager of advanced sales and marketing services, **Henry M. Stupich Engineering Dept.**, **General Electric Co.**, **Sturtevant, N. Y.** **Frank L. Mosier**, director of engineering, **Armstrong Industries, Inc.**, **Detroit, Mich.**

INDUSTRY OBSERVER

►North American's high-speed, high-altitude X-15 research vehicle will test the hobbled upward systems not developed by Convair as part of industry's Crew Escape Systems Committee effort (AW May 6, p. 94).

►Allison's Model 550 twin turbo-prop engine is now running on the test stand at the Indianapolis, Ind., plant. Model 550 delivers about 5,000 r.p.m. and was developed mainly by General Motors Corp. private funds in the new Allison jet development laboratory facilities.

►Navy plans to expend \$15 million in fiscal 1958 on new test and support facilities for its Polaris fast ballistic missile, including a new research and development facility in California's Santa Barbara area. Funds also guarantee the acquisition of present facilities at Patrick AFB, Fla. Naval Ordnance Laboratory, White Oak, Md., and industrial plant at Acmept General in Sacramento. Lockheed will send 200 acres to the government for the new facility.

►Minimum gross weight for the Bumblebee Model 18 Mustang turbo-prop aircraft is about 127,600 lb.—about the same as the Douglas DC-7B and approximately 14,000 lb. heavier than the Lockheed Electra turbo-prop transport.

►Mass differences between the Tu-110 four-jet Russian transport and its earlier twin jet predecessor, the Tu-104, is an additional 10,000 lb. added to the fuselage and about eight feet added to the wing span to accommodate the additional engines. Tu-110 has a maximum gross weight about 22,000 lb. heavier than the Tu-104. Redesigning jets also have been incorporated in the Tu-110 to get slower landing speeds and shorter landing runs.

►Some industry sources believe Gen. D. H. Baker, former director of procurement and production for the Air Materiel Command, was overly optimistic in predicting that there will be 57 million sq ft in surplus facilities by the end of 1961 (AW July 1, p. 28). Reasoning is that while space needed to house actual waste assumes a set as great as that required for additional aircraft much more space is needed for supporting missile manufacturing equipment. Such equipment, the sources say, will consume a large amount of the 57 million sq ft.

►Studies have been conducted at Holloman AFB, N. M., of special electronic film processing for photographing dark areas subject to connection with television systems and nuclear reactor systems. Projects have been developed by Harold E. Edgerton, professor of electrical engineering at Massachusetts Institute of Technology. Largest area covered thus far is 4 x 6 ft.

►Navy hopes to complete first design of nuclear surface airframes during the current fiscal year when blocked by present expenditures. Beginning of nuclear development is scheduled for fiscal 1959.

►Growth Landing is conducting a series of tests VTOL and STOL developments for the National Aviation Committee for Aeronautics. NACA will use the survey in planning future VTOL-STOL research programs.

►Navy may be forced to develop a means of automatically controlling flight of a catapulted aircraft for several seconds after launch. Point may not be reached where catapult launching high-performance carrier-based aircraft will have to crush speeds that will cost enough Gs on the pilot to cause necessary mistakes.

►Air Force, Navy, Army, Navy will be supplied by Pentagon decision to cut fiscal 1955 petroleum purchases between 10 and 15% in current fiscal year. Highest cuts will be made in aviation fuels and Navy special fuel oil.

►Mechanical translation of Russian into English will be demonstrated at Rome Air Development Center by late September. Mechanical language translation system was developed for Rome's Directorate of Intelligent Electronic Warfare by International Telemetering Corp.



The high frontier

Far to the north of seaplanes, roads and railroads, DGV line outposts stand guard over America. They were built with the help of Fairchild C-123 transports, which flew in the men and the construction material—even the radar antennae which now scan the arctic skies.

The C-123 commuter service to the DGV line goes on at the rate of up to eight tons of cargo per plane—

and at the end of the line the landing strips have proven too rough, too hardscore for any other heavy-duty transport now in service.

The members of this polar bear club have come to depend on the C-123—just as the U. S. Armed Forces everywhere. The C-123 shares these qualities with other Fairchild aircraft: ruggedness, reliability, and ever increasing utility.

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AIRCRAFT DIVISION • BAEYNSBORO 15, MASSACHUSETTS
A DIVISION OF FORTRESS INDUSTRIES
WHERE THE FUTURE IS MANUFACTURED IN 1970-1980

Engine Profits

House Armed Services Investigating Subcommittee's investigation of military aircraft engine profits and procurement policies officially got under way last week, as predicted by Aviation Week (Aug. 31, p. 21). And it is doubtful if they will be slowed down when the color information the subcommittee staff has collected on 17 engine manufacturers for the twelve period 1951 through 1976 is all be culled. Members of the subcommittee headed by Rep. Edward Herbert (D-La.) must to have Washington Area Congress airports in a week or so. Only one manufacturer has testified—Ray T. Harter, president of Curtis Wright Corp. (AW July 28, p. 38).

GM Under Fire

However, the subcommittee last week did press charges that General Motors Corp.'s Buick-Gilbertville Pontiac Division failed to furnish USAF negotiation contract cost data on a \$175 million contract for 995 F4H aircraft and that USAF negotiators were too lax in price reduction negotiation even on the basis of the inadequate cost data available. Testimony by the subcommittee staff included:

- In a subcommittee bill forward pricing on 300 aircraft a unit price of \$275,000 was negotiated. This was a comparison between General Motors' calculation of \$255,000 and USAF's estimate of \$262,000.
- Subcommittee staff's calculation based on information also available to USAF—was \$277,644 compared to \$254,075 estimate.
- Post audit by General Accounting Office based on "actual costs" and the correct unit price should have been \$150,000.

Rep. Edward Herbert (D-La.), subcommittee chairman, charged that USAF "conspired" to General Motors. "An F-4 was not adequate," he pointed, "in meeting its price it considered fair and reasonable."

MATS Probe

Senate Commerce Committee will shortly establish a special subcommittee to make a thorough study of the operations of the Military Air Transport Service as well as MATS. Sen. Warren Magnuson (D-Wash.) said the subcommittee will determine "whether these services are operating beyond the scope intended by Congress." Emphasis also will be placed on the study on the cost and efficiency of MATS compared with commercial airlines. House and Senate Appropriations Committee already have directed Defense Department to channel a greater portion of transportation from MATS to commercial carriers.

Military Pay Raise

Ralph J. Conner, president of General Electric Co., will make another bid this week for his proposed to charge private military pay scales in testimony before a Senate Armed Services Subcommittee headed by Sen. John Stennis (D-Miss.). Conner headed the Defense Advisory Committee on Professional and Technical Compensation which concluded that, by boosting pay for skilled employees, Defense Department could make tremendous savings by reducing uncompetitive salaries (AW

Feb. 4, p. 37). Conner estimated that if his group's recommendations were implemented the annual defense budget could be reduced by \$3 billion. Other members of the subcommittee are Sen. Hiram Boren (D-Okla.), Sen. Stuart Symington (D-Md.), Sen. Everett S. Sells (D-Iowa), and Sen. Frank B. Rowan (D-Vt.).

Longer Hitch

As F-15s, in another effort to stem the flow of costly technical personnel back to higher-paying civilian life, says before order pilots will have to sign for a longer hitch rather than the present three years. An Air Force spokesman said that, under the three-year plan, a pilot spent three-quarters of his hitch working to become combat ready. At the end of three years, two-thirds of the pilots returned to civilian life.

Missile Information

USAF's Office of Information Services hopes to have a conference on schedules of information policy regarding missile forces with outgoing Secretary of Defense Charles E. Wilson before he leaves, which will probably be around Sept. 1. Air Force Gen. William H. Tunner, on which to have a decision. Maria S. Sauer, Assistant Secretary of Defense for Public Affairs, has refused to allow newspaper authorities under pending laws the House Government Information Subcommittee, headed by Rep. John Moss (D-Calif.) (AW Aug. 5, p. 24). A memo by Sauer prohibits the disclosure of the details except consideration of the fact that a test firing has occurred and "a brief statement" as to any results.

CAA Fore Deadline

Following a Civil Aeronautics Board report that the General Passenger Air Regulations be revised, the committee Ralph E. Wiley last week set procedural dates and estimated that strict adherence to the schedule is necessary. Deadline for exchange of revised exhibits is Sept. 15, rebuttal exhibits, Oct. 11, written testimony of all parties except Bureau of Civil Aeronautics be submitted by Oct. 25, written testimony by Bureau of Civil Aeronautics, Nov. 1, submission of final briefs, Nov. 8, hearing, Nov. 15.

Airways Modernization Board

Legislation providing for the establishment of an Airways Modernization Board to monitor U.S. airways development over the next three years, became final last week, with the signature of President Eisenhower. At the same time, the President sent to the Senate the name of former USAF Lt. Gen. Elwood Quesada for confirmation as head of the new board. Quesada's appointment was first announced in MIA (AW May 27, p. 25).

Floberg to AEC

John F. Floberg, chairman of the Conference of Local Airlines, was recommended for confirmation as a member of the Atomic Energy Commission last week, by Senate members of the Joint Committee on Atomic Energy. —Washington staff

AMC Develops Missile Support System

Air Materiel Areas are given support responsibility in first step of plan to revamp AMC logistics system.

By Claude Witte

Wright-Patterson AFB, Ohio—USAF's Air Materiel Command has assigned support responsibility for missiles to five Air Materiel Areas.

This is the first step in a 10-year program to revolutionize the AMC logistics and communications system to meet the demands of missile combat capability. The Air Materiel Area will evolve into the first of a new USAF administrative unit, the Weapons Support Manager (WSM).

Initial assignments of missile support responsibility have been made by four groups. They are:

- GAR-1 Falcon and GAR-6 Sidewinder air-to-air missiles in Mediterranean, IV, Air Materiel Area.
- GAM-65 Basilisk and GAM-72 Green Quad in Oklahoma City, AMIA.
- TM-61 Matador surface-to-surface missiles in Warner Robins, Ga., AMIA.
- IN-49B Bomarc surface-to-air missile to Ogden, Utah, AMIA.
- SM-62B, SM-61F Noodin and SM-73 Bull Goose to Ogden AMIA.
- SM-65 Atlas, SM-68 Titan and SM-75 Thor ballistic missiles to San Francisco, Calif., AMIA.

Responsibility for AMC's Directorate of Plans and Programs also will move. Those who anticipate such responsibility for Air Materiel Areas will form the "lead team" of combat unit missile support in less than 10 years. Their new Weapons Support Manager commands will take on responsibility for the Weapons Support Officer (WSO) phase out at the missile development and production activity.

Missile System Support

Wright-Patterson AFB, Ohio—At present, there are only two staffs left outside the responsibility of the Air Materiel Area as an Air Materiel Area support manager for a particular missile family.

They are vehicles and propulsion units. Later, they develop that other item will be added to the list, but not until some sound money appears for the equipment.

The new Weapons Support Manager is responsible for all other items, from the lead team to the assigned support. This responsibility covers maintenance, with minimum exceptions of field level repair and personnel.

In addition, support of the missile support system will include the provision of missile maintenance teams available for worldwide assignment.

Air Materiel Command has selected the design and development of missile support teams to be assigned worldwide from the inventory of skilled personnel.

New Support Teams, which can run as high as 200 to 300 men, are available to meet a crisis at any world command installation. A hundred experts can be sent to a base in need of a supply unit, fuel unit, maintenance unit or to new stock levels of a unit in need of a supply unit, fuel unit, maintenance unit or to new stock levels.

and the missile enters active inventory. The process will be similar to that which the WSM, when AMC takes over control from the Air Research and Development Command.

AMC places strong emphasis on the fact that it is not making overnight introductions of new equipment and methods in the logistics process.

Factor Comparison Needed

Since the equipment and methods still do not exist. Still greater networks of electronic communications must be provided and the business machine manufacturers have a new demand for larger and faster computing devices. Over, already called by AMC demands, possibly in only a grade product of the machine that it needs in the next few years.

As these are introduced, all in support of missile combat capability and USAF economy, they will bring improvements in transportation techniques applicable to material, aircraft and applied at the airport in these weapons.

The first already has been made. Unleashed of the Air Materiel Area as a weapon system, change into the commander of a system, the support manager actually is getting a tryout with conventional aircraft.

Such units, with concentrated management authority, are in all areas and maintenance capability, have been established for the Century series fighters and the Boeing B-52 long-range jet bomber. Experience will continue.

This concentration of responsibility, maintenance and support teams that can keep an aircraft MAF at a combat base in need represented by the next

commander. From now on, he will rely on a single installation within the U.S. for all his equipment, spare and maintenance support. Parts, skills, management responsibility and activities control will be centered at one point.

Establishment of weapon system storage sites in case some of the material along with improved inventory controls are expected to reduce the size of AMC's physical structure in the next 10 years. The combat unit no longer will go to specialized depots for special components, such as the electronics rapid depot host at Galesburg AFB.

In addition, concentrated supply and maintenance activity will phase out of the inventory depot structure. The main support functions of both the U.S. specialized storage depot and the overseas depot will be assigned to the Weapons Support Manager. Personnel for the next part will be used where the skills are in demand, at the remaining Air Materiel Areas, the weapons system storage sites or in mobile maintenance units.

Change of the storage site and home of the WSM will not depend on which phase of the transition is reached. With the growing introduction of missiles, AMC expects the most efficient plan will be along the "lead group" line, used in the initial assignments. In this case, one type of missile will be sent to Middletown, another to Oklahoma City. The ballistic types are centered in California and the cruise missiles at Ogden, Utah.

Expanding Standardization

The headend will center parts in skills in handling and maintenance of a particular type of missile at one base. It will facilitate use of interchangeable parts when this is possible. It also will expedite the constant usage for new developments.

In assigning management responsibility for the first missile to specific Air Materiel Areas, it was made clear that AMC will continue to provide its own support for first-line weapons. Combat support support will be used only in research and development stage and in weapon system development.

Prime contractors will find two new conditions facing them when missiles are produced in the USAF inventory. There will be fewer stocks of parts, made up in the form of a new complete and accurate information about demand. On top of this the industry will be forced to help supply materials to make this possible.

For foundations, AMC is looking for results by 1960 from a new Division of Logistics Research in the Directorate of Plans and Programs.



How B-47 Carries Rascal

Mounting of Bell GAM-65 Basilisk on belly of B-47 bomber on a base with wing is shown in photo above. Expected, 32 B-47s, now in inventory, will become operational with SAC B-47s this year (AW Aug. 12, p. 31).

AEC Begins Nuclear Ramjet Study

Washington—Atomic Energy Commission has initiated feasibility studies on the application of nuclear power to aircraft and rocket engines.

First official statement that AEC is interested in the possibility of applying nuclear power to engines is contained in the commission's 22nd semi-annual report to Congress, released the first six months of the year.

Last September, AEC reported that it was "concerned with research and development of nuclear rocket propulsion" (AW Sept. 17, p. 23) at the N. D. Brown at Los Alamos, New Mexico Laboratory, Los Alamos, N. M. and at the Lawrence Livermore Laboratory in California. The announcement was a continuing effort aimed at letting scientists and engineers know the work aimed.

The current semi-annual report said AEC "continued studies" and "continued feasibility studies" relating to applications of nuclear power to aircraft and rockets. The work is being done in AEC laboratories and will be industrial or security application.

The report also contained the first AEC statement that atomic batteries are expected in the future to power

"equipment in guided missiles and space craft."

The type of small experimental atomic batteries have been announced so far. They consist of nuclear reactors to electric drives. Useful life depends upon the radioactive fuel, but AEC said lifetime can be "many years, and the batteries are a virtually maintenance-free and rugged, reliable."

Latest batteries announced use the radioactive perchlorate 147 to power a small switch. It produces 20 micro-watts, but a useful life of more than 5 years and is smaller in diameter than a dime. AEC also said:

• **Revisions and additions to laboratory facilities at Los Alamos.** "Increased by nuclear and thermonuclear nuclear program and the nuclear propulsion program and the weapons program."

• **Reactor experiments** relative to nuclear operation of turbojet engines" contained under General Electric Co. at the General Electric Turbine Division at Idaho Falls, Idaho. First power of a turbojet is being done in a nuclear test, phase three in 1956 (AW Feb. 4, p. 29). GE and Pratt & Whitney Div-

ision of United Aircraft Corp. are engaged in studies under Air Force's air craft nuclear propulsion program. Construction of a Low-Power Test Facility also was begun at Idaho Falls.

• **Construction of facilities at Pratt & Whitney's nuclear engine laboratory in Middletown, Conn.** "provided safety features and some portions of this complex were completed" during the month period. USAF is building the plant. Pratt & Whitney is named to operate the nuclear reactor to test USAF and AEC.

• **AEC contacted** an option to buy an electronic equipment production plant known as the Pacific Northwest Plant, which General Electric recently completed for it at Richland, Wash. and General Electric, Wash. D.C.

• **Development continued** in "a program exploring new design principles which can be used more effectively for defense purposes" and "methods of reducing the radioactive contamination resulting from weapons destruction."

• **USA's nuclear** (SSN-571), successor of Nautilus, nuclear carrying nuclear powered submarines, traveled 62,740 mi without being refueled. Of this, it traveled 50,495 mi fully submerged.



ERGONOMICS of new flight suit and boots is demonstrated in a San Diego swimming pool.

Safer, More Comfortable Suit Developed for Supersonic Flight

Light weight flying suit, based on the present concept and designed to cover the flight qualities of USAF's supersonic fighters, interception and bombers, has been developed for the Air Force by a subcontractor of the Institute, Com. Aerospace Systems Center. New flight gear will give superior pilots greater safety, more mobility and increased comfort.

Convair-Sun Diego's Human Engineering Group functions as chairman of the ICESG subcommittee on development of personal equipment. ICESG is composed of 15 major air craft companies including responsibility on pilot cockpit station problems (AW May 6, p. 94).

The new garment, which is being developed by Convair's Human Engineering Group under guidance of the Aero Medical Laboratory, Wright Air Development Center and the ICESG subcommittee, is made up of an inner padded jacket, boots and gloves which are integrated into a flight suit.

Convair attributes three advantages to the new flight suit:

- Combined into one three piece outfit (the five separate garments now required for survival, water-repelling, thermal heat for survival, sustaining suit for cooling, partial pressure suit, and undergarment).
- Provides up to 10 seconds additional protection against burns.
- Flashes a pilot face-to-face of an

accident without full gear and in cold areas without undergarment.

- Decreases bulk of the equipment going at greater comfort and reducing drag during a supersonic cruise.
- Reduces by about one third the time a man takes a pilot to don the full complement of personal equipment, an important consideration in case of a sudden return to an attack.

Here is what Convair engineers had to give a pilot once he is in the air while his body is under pressure and more care is put during the 90% of the time he is not in a cockpit.

- Reduced has a larger five piece for increased mobility, is structurally stronger to withstand supersonic operation loads, on a low bulk and is lighter in weight than the M-2 and M-3 types in use.
- Soak at the neck incorporates a built-in hearing between helmet and neck allowing a pilot to turn his head back at 90 deg. in either direction. All connectors, cuffs, cuffs, cuffs and cuffs are built into the suit.

Other features include a built-in hearing between helmet and neck allowing a pilot to turn his head back at 90 deg. in either direction. All connectors, cuffs, cuffs, cuffs and cuffs are built into the suit.

- Glove are made in two parts. One part is fingerless and penetrates the back of the hand. Second part covers the fingers but without penetrations. Convair says tests have shown that if the fingers are not enclosed in the glove, the hands do not lose mobility and dexterity as happens with present equipment.

Boots are made in the pilot's foot with a low bulk and are made in a single piece. To use them on, a pilot steps into the boots to see and feel them close. Fasten at the back of the foot, not at the front, which helps a pilot to adjust and move and makes it unnecessary for him to take them off in the water.

New suit has been tested extensively for pressure in an altitude chamber at Randolph Air Force Base, Tex. for test flight and declassification studies at USAF's SMIARF track at Hurler Mountain, Utah. For the meantime at Convair-Sun Diego, and for engineers in a San Diego swimming pool.

Convair concludes that the new suit is "not so much a new design as a redesign, and it is still based on the present garment concept."

Other companies working on the ICESG personal equipment subcommittee with Convair include Boeing Aerospace Co., Convair-Fairchild, Lockheed Aircraft Corp., North American Aviation, Inc. and Republic Aviation Corp. Naval aircraft-related officers from North Island Naval Air Station, San Diego, are in an advisory

Greater Human Factors Emphasis Asked by Navy Safety Expert

By Richard Secor

San Diego—Cockpit instrumentation today is designed for displaying the man who is flying the plane, a Naval safety expert told Institute of Technical Services members meeting here last week.

Accidents often attributed to pilot error only are failures of instruments, according to Capt. J. Stankovic, chief safety officer of staff commander, Air Force Pacific Fleet.

Capt. Stankovic presented a paper at one of two open sessions among the five of the IAS National Naval Aviation meeting. The session was highlighted by discussion of the need for instrument failures considered as one of the parts of weapon system throughout their evolution and use.

Cockpit Problems

The safety officer said these problems in present cockpit instrumentation:

- Today's attitude indicator is small, yet this is the most vital instrument and should be at the center of attention and control in read, with all other gauges and designed to be complex.

How about a rectangular presentation with paper color for each and sky for red?

In the display from many sources has been developed, but the layout and readout are not, which through the layout, still is an existing factor. He added it might not be better to delay some more studies in this area and concentrate on solving this problem.

- Altitude at present, with all these hands presents a problem. Missing is a frequent and dangerous, and is necessary to be well recognized.

• In personal equipment, a part of today's high performance airplane cockpit environment the present suit is highly uncomfortable and almost unbearable pilot and crew.

He said, however, that while an emergency demands it use in present development.

Kind of today's instrumentation displays and cockpit environment is that accident frequently attributed to pilot error really are failures of instruments, Stankovic declared. Pilot problems resulting from today's cockpits and presentations are much more than just purely physical ones.

Other Republic of the aerospace system which need closer human engineering from the very beginning, the paper indicated, are those parts which affect

instrumentation and navigation. He should be used to what some personnel can maintain without strain, which in time leads to maintenance problems and sometimes to accidents.

System read design and engineering aimed at instruments to display of complete advantage to be taken of the advanced capabilities then made, according to the paper.

Also, Stankovic said, experience has shown that, while it is the object, experience has shown that the value of new aircraft is lost, it is the younger, less mature pilots who fly them and incur high accident rates.

In these cases, the paper pointed out, the older, more mature pilots have learned to live with the consequences of the undesirable characteristics of airplanes, to compensate for them intensively, and hence those characteristics which are harder for experience to change pilot into through test phases without correction, showing up in operational accident rates.

Another factor bearing on the situation is that techniques once accepted operationally tend to be used again. Stankovic said, this is especially true in pilot training. The present methods of standard are supposedly adapted to newer planes and perhaps the new basic investigation leading toward improvements in the cockpit and cockpit techniques which are not necessarily the best ones can be needed out and new ones substituted.

Workload Analysis

Concerning workload problem analysis, a paper prepared and delivered by F. Ciesler, director of the Air Warfare Research Department, Air Development and Materiel Command, said one of the most serious problems of the work is being sensitive to changes in original assumptions upon which weapon systems for a given combat condition are based.

For best results, analysis should always be willing to revise original assumptions and change them according to new developments on new information. Accents of assumption should be questioned every time they are copied in subsequent facts, the paper said.

Another factor bearing close scrutiny is analysis of weapon system requirements in terms of time rather than cost. It is an original analysis which called for development of a very sophisticated weapon system to meet the conditions, it is a waste of time to specify a degree of sophistication which

can be reached eventually. The paper said such capabilities are desirable in case of both with effects on the unit, base period in which the system cost is the most important system cost in function.

Superfluous abilities leading operational capability are production rates in an absolute reference are to be modified, the paper indicated.

Fallible Failures

Judging of safe service life in high performance aircraft is a series of possible fatigue failures calls for accurate determination of load and structural analysis as early as the plane's life is possible, according to a paper prepared and presented by Dr. J. H. Stankovic, Design Division of Navy's Bureau of Aeronautics.

Today's structures of heavy planes and fighters have indicated, Stankovic said, the complete failure of the structure after 100,000 cycles, making it as possible that accurate as possible knowledge be obtained as to when the plane's service life is nearly over and its structure is nearing the time period when people or loads may cause sudden and complete failure.

He cautioned, however, that too great a conservatism in forecasting a current service life for a design can cause trouble in the too early retirement of very expensive weapon systems while operational capabilities remain within safe operating margins.

Characteristics of today's loads, Stankovic's paper noted, are low cyclic, and high peak loads.

Affecting this in turn are changes in mission the plane was not originally designed to accomplish, unexpected stresses which induce loads near the limit load factor means aircraft must be anticipated in original design specifications.

Navy Progress

In its current program to move an operational jet into service life, the Navy has high-speed analysis graphs and statistical accelerometers mounted in at least 150 of the 20 latest types aircraft in use.

But of flight safety engineering as a discipline about every aspect of aircraft design and test, C. G. Miller, Chief of Flight Safety, recently reported, and in his paper incorporation of safety engineering into the design of the plane's development cycle, from design through test flight and operations.

Miller said flight safety engineering should have a high position within the command concept framework, with its activities spread throughout the various working levels and groups with an engineering flight test and flight test.



COMFORTNESS of new suit is evident here.



Convair has built back into new suit (right).

Pratt & Whitney Says Cutbacks Peril Nuclear Engine Project

Windsorham—Pratt & Whitney Division of United Aircraft Corp. told its stockholders last week it is "probable" that its nuclear engine project "will be drastically reduced in scope or even possibly eliminated entirely."

An officer asked to say what its interests in the Pratt & Whitney project is thus far but confessed that the aircraft nuclear propulsion program as a whole is under intensive review, and said a decision is imminent on how much development to continue to support.

One likelihood is that USAF will cancel its support of the Pratt & Whitney program, but that the Atomic Energy Commission will continue to support it. Air Force at Langley, the Convair Aircraft Nuclear Engine Laboratory at Middletown, Pratt & Whitney is scheduled to see it for aircraft development make contact with USAF and AEC.

But if USAF, the major backer in terms of dollars, should decide to abandon this project and continue to support only the General Electric Co. project at Fernald, Ohio and Idaho Falls, Idaho, Pratt & Whitney might conceivably decide not to continue the work at such a low level of support.

Pratt & Whitney pointed out that any change in the status of its nuclear project, "if it comes, will have little effect on current operating results or those for the future, as it has no or been counted on internally in our financial forecasting due to its highly experi-

mental nature and production potential. All capital assets devoted to this particular project have been factored by the government."

Stopping of development work on the T37 turboprop engine also was reported by P&W, due to Air Force's cancellation of the Douglas C-119 cargo project. The company and work on the T37 "could be quickly resumed if a requirement for such an engine developed in the future," because development work on the T37 turboprop engine is continuing.

The company said possibility of a change in the status of the nuclear engine project was due to "the Defense Department's intense drive to combine new war, the rapid advance being made in other forms of aerial propulsion."

Future of atomic propulsion for the nuclear powered aircraft also are in doubt. Lockheed Aircraft Corp. this year has carried on its weapons system contract on the nuclear plane with about one third as many engineers as last year. Simmonds, notes that the laboratory being built in North Carolina would come into operation some 18 months later than it could have. Lockheed and Pratt & Whitney are teamed on one atomic-engine concept and Convair Division of General Dynamics Corp. is teamed with General Electric on a second concept.

The Convair GE program also is continuing at a lower dollar volume and longer time phasing.

Smith Leaves Slick, Replaced by Rentzel

Dallas—Robert J. Smith resigned as president of Slick Airways last week, with Dixon W. Rentzel again becoming operating head of the cargo airline.

Smith resigned his position of president, chairman and manager of Slick's executive committee, severing all relations with the airline. Reason cited by Rentzel in announcing the change was that Smith is engaged in a number of other activities and finds he cannot do justice to them and to Slick.

With Smith's resignation, Rentzel assumes the duties of president he dropped when Smith joined the airline in May, 1956. Rentzel joined Slick as chairman of the board of directors in November, 1954, and served in chairman and president during the period when the carrier was reeling after the planned merger with Flying Tiger fell through. Rentzel remained as

chairman when Smith took over the presidency last year.

Before joining Slick, Smith was involved in a number of activities, including duties as board chairman of the Elavich Federal Reserve Bank and president of Pioneer Aeronautical Services and Pioneer Air Lines. He continued his association with the Federal Reserve Bank and Pioneer Aeronautical during his tenure with Slick. Smith is also a USAF Reserve brigadier general.

Rentzel said Smith came to Slick with the understanding that he would continue active participation in all these affairs. After trying it for a year, Rentzel said, Smith finds "it is not feasible for me to run to the police in all of these important activities."

"While we understand his devotion to these other interests," Rentzel said, "we deeply regret that he has elected to resign from Slick. Under his direction, the company was successful in obtaining additional financing needed to continue the expansion of its aircraft fleet. His appointment allowed the company to maintain its international operations and to support its domestic carrier service."

News Digest

Boeing Aircraft Co. has been awarded a \$174,315,444 Air Force contract for production of the IM-99 Bomarc interceptor missile. USAF will not be Boeing for area defense.

Donald G. Moore, Naval Research Laboratory, specialist in design and development rocket technology, was named last week as manager of the Vanguard Overseas Group at Patrick, AF, Va. Moore, an electronics engineer with NRL since 1946, will supervise the Navy's part of the operations at Patrick for the earth satellite program.

Brig Gen Victor Hagen has been named assistant deputy commander of the Air Research and Development Command for Weapons Systems with headquarters at Wright Air Development Center, Dayton. He succeeds Maj Gen Harold Ellis who moves to the Pentagon as assistant USAF chief of staff for air defense matters.

Boeing L-25 helicopters in being demonstrated in Gulf Coast oil companies in a 4,000 sq. mile. Daily spare parts carried on spool plans.

Bell Aircraft Corp. organized Burns Manufacturing Co. of Buffalo, N. Y., a producer of modified Fibergrip propellers. Transactions, including stock and shares of Bell stock, makes Burns Bell's seventh wholly owned subsidiary.



GLOBAL GUARDIAN—Nowhow can an aggressive take refuge from America's mighty instrument of retaliation, the Northrop B-2 Spirit? world's first intercontinental guided missile, now scheduled for delivery to the Strategic Air Command of the U.S. Air Force. These platoon bombers are designed to carry nuclear warheads with steering accuracy over intercontinental ranges. B-2 Spirit missiles are launched in mass from into thousands of miles from their targets. This powerful deterrent to attack in its earliest product at Northrop's "years ahead" engineering. Bombs, cruise, supersonic air attack, ground support and movement equipment, integrated weapon systems and other vital national defense elements are being developed and produced at Northrop, where simplicity and economy are vital elements in the continuing search for weapon systems to guard tomorrow's day America.



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AIR TRANSPORT CAA Says Jet Airliner Problems Ease

Second jet age planning report is issued by CAA in effort to put jet problems in proper perspective.

By Ford Eastman

Washington—Problems expected through the use of jet aircraft in civil air transportation will not be as serious as first thought, the Civil Aeronautics Administration reported today.

In its second jet age planning program report, the CAA and industry study of jet operational requirements has reduced many jet problems to their proper perspective, dispelling much of the earlier concern. The agency's first jet program report was issued last year (AW Feb. 24, p. 27).

The CAA said that, in the past, "various quarters have expressed concern over such questions as runway length and strength requirements, foreign object damage, jet fuel reserves, fuel reserves to delay flights and noise. Now, runway problems concerning length and strength appear to be of relatively minor concern where the jets will fly, the main noise structures in today's larger passenger aircraft. Runway length at most, however, probably will be required for larger aircraft that will fly in stages in areas of cities, presently aerodynamically, the CAA said.

Foreign Object Damage

Foreign object damage of jet engines has been a serious problem in past military operations, the report said, but added that improvements in contemporary and engine designs at civil jet transports, plus potential developments such as the automatic service and good knowledge of airports and structures are now considered sufficient to handle this problem.

In the case of fuel reserves, the report pointed out experience with military jet operations where short fuel reserves were encountered was amply demonstrated when applied to civil jet operations. The civil jet manufacturers it said, have used great care in designing the jet transports to have holding fuel reserves and general terminal performance comparable to present civil practice.

As for reserves of fuel stored in progress, the CAA said the low speed and variable altitude jet engines have shown the ability to have reserves reserves comparable to present, large piston engines.

Another question expressed with the jet engines was that of engine thrust

for holding. Here again, the CAA said, the problem has been reduced through the development of reserve thrust devices.

Because of the public concern over jet aircraft noise, the CAA placed special emphasis upon the noise problem. Public concern was based primarily upon the noise of military aircraft—the same boom, jet afterburners and noise as military.

Still, civil jet transports will not go faster than the speed of sound, the main boom will be an exhaust plus the noise of the engine. CAA also pointed out that the jet aircraft also is an excellent military feature which will not be used on civil jet transports. As for taking noise the agency said, all civil jet transports are to be equipped with suppressors mounted on the engine to reduce noise levels by 50 percent. Such suppressors will reduce the noise of the jet transports to 20 percent that of present, large piston-engine aircraft.

The CAA jet planning program was initiated one and a half years ago to provide the necessary information possible to ease the transition from piston engine aircraft to jets. Both government and industry are participating. The program has been aided substantially by the CAA said, by manufacturers of jet and turbo-prop aircraft through special jet planning conferences and provision of considerable technical information. The report also has been contributed through providing background on jet operational experience.

Report Highlights

Here outlined for the current report was provided by the CAA offices of Flight Operations and Aerodynamics, Airports, Air Traffic Control, Air Navigation Facilities and Planning Development. Other highlights are:

- **U.S. Airline Transport Production.** Current schedule calls for two large jet transports in Jan. 1, 1969, 31 in July 1, 1970, 125 in Jan. 1, 1971, 161 in Jan. 1, 1972, 161 in Jan. 1, 1973, 161 in Jan. 1, 1974, 161 in Jan. 1, 1975, 161 in Jan. 1, 1976, 161 in Jan. 1, 1977, 161 in Jan. 1, 1978, 161 in Jan. 1, 1979, 161 in Jan. 1, 1980, 161 in Jan. 1, 1981, 161 in Jan. 1, 1982, 161 in Jan. 1, 1983, 161 in Jan. 1, 1984, 161 in Jan. 1, 1985, 161 in Jan. 1, 1986, 161 in Jan. 1, 1987, 161 in Jan. 1, 1988, 161 in Jan. 1, 1989, 161 in Jan. 1, 1990, 161 in Jan. 1, 1991, 161 in Jan. 1, 1992, 161 in Jan. 1, 1993, 161 in Jan. 1, 1994, 161 in Jan. 1, 1995, 161 in Jan. 1, 1996, 161 in Jan. 1, 1997, 161 in Jan. 1, 1998, 161 in Jan. 1, 1999, 161 in Jan. 1, 2000, 161 in Jan. 1, 2001, 161 in Jan. 1, 2002, 161 in Jan. 1, 2003, 161 in Jan. 1, 2004, 161 in Jan. 1, 2005, 161 in Jan. 1, 2006, 161 in Jan. 1, 2007, 161 in Jan. 1, 2008, 161 in Jan. 1, 2009, 161 in Jan. 1, 2010, 161 in Jan. 1, 2011, 161 in 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Eastern Wins Nonstop Route Between New York, Mexico City

Washington—Designation of Eastern Airlines as the U.S. carrier to operate nonstop service between New York, Washington and Mexico City was approved last week by President Eisenhower. The Civil Aeronautics Board had recommended Eastern over Pan American World Airways by a three to two vote.

Under the bilateral air agreement with Mexico, Eastern will commence flights on June 30, 1959.

Eastern hopes to begin service within the next 30 days if the agreement is negotiated satisfactorily as expected. Board Chairman E. V. Rickenbacker said.

Rickenbacker praised the Board's "total lack of concept of the issue involved" and its "searching desire of the public interest and needs."

The airline route to Mexico City, "Golden Falcon," aircraft for first class service on the route, also provide cargo service primarily with Super C Constellations.

Mexico has designated Aerolineas de Mexico to compete with the U.S. carrier on the New York, Washington-Mexico City route. Its application for a foreign air service permit has been approved by CAB Executive Francis W. Brown, leaving only the Board and the President to sign the final order.

Designation of Eastern and Aerolineas de Mexico by the two countries is a double blow to Pan American. Not only did the airline lose out to its own carrier, but Mexico City, New York and Washington, but it now must discontinue all of its stock interest in Aerolineas de Mexico. Pan American interest in Aerolineas was purchased by the Civil Aeronautics Board in 1954.

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The CAB majority said the evidence in the case indicated that Pan American's proposed operation would not be a profitable one, much less produce sufficient profits to offset delays. The three members and Pan American proposed only through flights between Mexico City and Europe via Washington and New York, and that, while these schedules would be attractive to travelers between Mexico and Europe, these flights would account for only 3.5% of the total service over the Mexico City-New York route. At the same time, the bulk of the traffic would be along the line connecting Mexico City and Europe, rather than that which Eastern proposed.

The minority, members Chas. Carey and Louis J. Horst, said the market in question is more largely served by foreign operators and that U.S. air-

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Russia Turns Attention to Feederlines

Moscow—Russia has begun to pay more heed to feeder lines in its neglected hinterland.

With the USSR's growing bulk of new helicopter and turboprop aircraft for its hinterland and domestic feeder lines (AW July 22, p. 75, Aug. 12, p. 40), Russia's vast air work of local routes has been doing business as usual with the world's largest fleet of subsonic commercial aircraft.

But things should begin improving in the near future, for the Soviet leaders need airways.

Most likely is that G. K. Antonov, designer of the new Ukrainian turbo-prop transport, is once again given his chance in the field of the local carrier. His Polikarpov (L-40) turboprop transport is expected to be ready for service short haul airline service by next summer. One version of this feeder line will carry six passengers in two tandem rows.

The An-40 Little Bear, with its 141 mph top speed, will be better than one of the Soviet aircraft now used on the local carrier. Its ability to take off and land in 335 ft. will be valuable for the operators of the airfields that mark such all-weather air points on Russia's feeder routes.

Antonov's management, who is now in charge of the local plane industry and local leaders for the local airline An-40. Poor communications and navigational aids limit service regularly scheduled flights, except for "unofficial" flights to the major feeder centers, which are used as bases.

An overhaul of the local route structure is also in the works. Aviation officials that the network has grown too large and too complex to be managed by the state.

The time has come for local and enterprise airlines to develop and build up their own local routes. Through stages of regional economies, a network of "regional airlines" will be developed, which will be the mainstay of the Soviet feeder system.

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Convair-Eland Sale

London—Convair of this City has 140 aircraft with Super Eland turboprop engine has been ordered by REAL Airlines of Great Britain to be used by Pacific Expressways Corp., of South Mexico, Calif.



Flameouts Trouble RAF Comets; Modification Program Under Way

By William J. Coughlin

London—The Hamilton Comets in its operations, several have recently had a series of flameouts of their Rolls-Royce Avon Mk. 117 engines under tropical flight conditions.

Early difficulties were traced to icing on the fuel nozzles, now cured by a change in operating procedure. Since then, however, there have been flameouts in turbulence similar to those encountered by early versions of the Hawker Hunter when from high gear.

A modification program, including both ground and flight testing, is under way at Rolls-Royce and de Havilland in an effort to solve the trouble. Flameouts and consequent runway landings were considered by a General II of Royal Air Force Transport Command on a flight through a severe tropical storm above the Bay of Bengal.

Transport Command, which has taken delivery of ten Comet II's, currently is operating on a schedule between England and Australia, through Tokyo, Saigon, Colombo and Singapore.

Turbulent Area

The latest trouble occurred on a Comet which entered an area of severe turbulence above 40,000 ft. Two of its engines flamed out while en route, the remaining two dropped off due to engine failure. The aircraft descended to 15,000 ft above, as the third at impact, the pilot succeeded in sighting the two engines while still flying in clouds.

The trouble is attributed to sharply disturbed flow across the intake at high incidence, resulting from the peculiarly rough conditions of the international fleet.

Following the loss of Bengal incident, de Havilland test pilot John Cooney had a Comet II fly to simulate over England and put it through a series of violent maneuvers in an effort to reproduce the pattern of the flameouts. They did not occur.

A Comet II, with de Havilland and Rolls-Royce engineers aboard, then was flown to the area where the trouble first was encountered. The aircraft deliberately seeking to duplicate the most conditions of the last incident.

The experiment was successful—about 300 ft. more, according to technicians who went through the violent gusts which at times exceeded 40 ft.

The result was even flammier and a serious case of rapping an another en-

gine. It was determined that the difficulty occurred above 40,000 ft. at 575 mph, at Mach 75—slightly above the Mach 75 normal cruising speed for the Comet II.

Transport Command is operating the Comets under a full and complete airworthiness certificate up to 43,000 ft. If it should lose its civil certificate in any event, the Comet would not suddenly be withdrawn from passenger service by the RAF.

Rolls-Royce reports that the trouble can be modified by reducing rpm to 7,000 and speed to Mach 65. This is its recommendation to Comet pilots entering such weather fronts.

The immediate danger, as involved by following these operating conditions, was a possibility for the engine fires, and the next step is to see what can be done to cure the trouble.

Modification now being flight tested involves raising the setting of the variable-incidence compressor inlet guide vanes.

The guide vanes were designed for maximum incidence at low engine speeds and low incidence at high engine speeds, producing a maximum of thrust. They are operated by a van on the lower right-hand side of the compression casing.

Within the operating range of the guide vane, the van portion is shifted to engine speed by a governor pump. The van moves in the way in which the engine and opens the vane to reduce the stall and increase the airflow. The modification, raising the opening angle of the vane, reduces mass flow slightly but it is hoped

that it will ease the flameout trouble. The modification has been flight tested in England but not in the Bay of Bengal area as yet.

Flight tests are under way to test the de Havilland Comet II's in the latter tropical limit to determine whether similar difficulties occur with its Rolls-Royce R.A. 19 Avon engines. British Overseas Airways Corp. has taken delivery of the first of its Comet II's which are to be used to gain experience on the engine which will power the Comet IV.

New Series

On the IIE, two of the R.A. 29s are installed in the outboard position with intakes modified to provide sufficient flow for the more powerful engine. Rolls-Royce engineers do not expect a duplication of the flameouts, noting that the 200 units is considerably more advanced than the more 100 series Avon.

The flameouts occurring in the outboard first were the second series of flameouts encountered after the de Havilland Comets were placed in service by Transport Command.

When the aircraft began operating on the route to Australia, wing problems developed. On occasion, two engines at once flamed out—sometimes in clear weather. In all cases, the engines were sighted successfully at altitude and an attempt had to land with a dead engine.

The trouble was traced to fuel filter icing, producing flameouts from extremely low fuel temperatures. This was solved by changing the position in use of the filter heater. Previously, the switch actuated from the cockpit, burning was noted, was spring-loaded to shut off after a minute. A red warning light indicated when flow through the

filter was reduced. Now, when fuel temperatures reach zero, the heater is turned on and left on. The engine no longer had temperature before have been shifted closer to the engine at the low pressure filter valve.

To abate the slight increase in fuel consumption, caused by leaving the heater on, Transport Command has increased its payload fuel storage from 6,575 to 8,575.

Transport Command states that from the immediate standpoint it is well pleased with the performance of the Rolls-Royce engines. In more 24,000 engine hours, there have been only four recorded engine changes.

The RAF Comets are fitted with Elex zero valve and the converted engine that is "absolutely certain" to operate on the Australian route in view of the trouble encountered. Simplest solution is the flameout problem in keeping out of the severe and unusual weather conditions where they occur and the short-warning radio makes this possible.

The modification program being carried out by Rolls-Royce and de Havilland is not considered extremely urgent since RAF has no further Comet II's on order and a limited number of engines therefore are involved. The engine staff, of course, is one of the nation's.

CAB Fare Probe to Investigate Earnings vs. Jet Financing Needs

Washington—Effect of current low levels and earnings upon airlines' ability to attract financing for new jets will become primary when the General Program Fare Investigation is resumed on Nov. 18, according to the Civil Aeronautics Board.

The decision to move airlines' re-equipment and financing program one step further after the Board had turned down a bid by airlines for a 6% per single fare increase in an emergency measure to meet rising costs and declining profits (AW Aug. 12, p. 55).

The CAB said that, while it had voted to deny an immediate 6% increase in fares, it recognized the importance of only one of the domestic airlines' fare level in the entire domestic airline system and not merely for the seven carriers that sought the emergency fare boost. As a result, the CAB ruled that the General Program Fare Investigation be postponed for hearing and decision as late as possible within the limits of due process and fair hearing.

Testimony and data introduced in the 6% case will be consolidated into

the General Fare Case. The Board said trends of the data would avoid needless duplication of time and effort. It added, however, that it may be necessary to provide for additional cross-examination and for submission of evidence by parties who did not participate in the 6% case.

Testimony Consolidated

The CAB and the hearing in the 6% case did not preclude interested parties a significant relationship between the Board's rate-making function and the new equipment and financing plans of the domestic airline carriers. It said it believed, therefore, that the general case should include a full report upon the equipment and financing plans of the carriers in a full fare case period.

As a result, each airline carrier was ordered to produce in a future five-volume report its equipment purchase plans, traffic forecasts, financing plans, equipment retirement plans and evidence as to the long-run effect of its future equipment plans and financing program upon the general rate level and program form. Airlines were asked to present



BOAC Britannia 312 Flies

Second production Boeing 312 of 18 on order by British Overseas Airways Corp. under its first flight of Filton. Howard Hughes reportedly is negotiating to buy Britannia, but BOAC has failed to going up with its 312 order as Hughes is seeking. Total Hughes order for TWA would be 21.35 aircraft.



CAB Maps Near Collisions

These 331 near collisions were reported to the Civil Aeronautics Board in the first quarter of 1957, about the same rate as the last quarter of 1956. Map shows locations, with numbers in heavy dots indicating reports at places with too many to show by individual dots.



Vapor trails high in the sky will remind you that the 707 is flying its proving runs. Soon these fine planes, the first American jet transports, will come off production. American Airlines will be first to offer jet travel in the U. S. A. Early in 1959 American will use the 707's on transcontinental Mercury service. **AMERICAN AIRLINES**
America's Leading Airline

North Atlantic Fares

Washington-Civil Aeronautics Board has tentatively set stand against approving a 19% passenger fare increase across the North Atlantic on the basis that no showing had been made as to the consumer need of the increase.

After the CAB had heard down the proposed increase (AW June 10, p. 45) European governments asked for cross-industry through the State Dept. award.

Recommendation this was acted by its individual carriers—Air France, British Overseas Airways Corp., Lufthansa, KLM, Royal Dutch Airlines, Sabena and Swissair.

At the same time, CAB extended its approval of an international Air Transport Assn. agreement providing for a new low-fare, high-density service across the Atlantic by March 31, 1955.

CAB approval of such service was reportedly scheduled to expire on Sept. 30. Implementation of the plan has not yet begun.

this information at an early stage after hearings began.

"Closely related to the carrier's to co-operate program," the Board said, "is their ability to present evidence. In the light of this fact and the clear ability of having full information on the aspect of the general fare problem, we will direct that the evidence as to rate of return be given first priority in the order of evidence to be received at the hearing.

The CAB said the new program came out its pledge in the earlier an endorsement of its vote in the Standard Fare Case "in maximum cost-making surveillance" of airline fares in announcing its own order, the Board said the companies' relief to quoted by the seven carriers was not justified on the basis of the record before it.

Japan Studies Use Of Route Over Pole

Tokyo-Japan Air Lines hopes to open an international airline service across the North Pole by 1961 under a new 10-year aviation cooperation plan now being studied by the Transportation Ministry.

The plan contemplates an air route between Tokyo and Copenhagen identical to the one inaugurated by the Scandinavian Airlines Service in February. The route may be extended to London later.

In the hope of the transatlantic one-way service in SAG's trans-pole service JAL feels there is room for another airline along this route.

AIRLINE OBSERVER

Aeroflot now is now being checked out to four U-2's turbo-prop transports prior to beginning regular service with the new Antonov An-10 transport on Aeroflot's domestic routes. Aeroflot has not yet taken delivery of its Illyushin designed Moscow transports, and indications are that only the prototype is now flying.

Continued engineering on Fieschi's B-27 turbo-prop transport is progressing slower than expected, and the company now estimates the first delivery to West Coast Airlines in March, 1956, about 90 days later than the first delivery date scheduled. Part of the reason is due to the fact in December.

American Airlines' public offering of surplus C-46's to the corporate aircraft market is primarily aimed at establishing a better price than the current market, usually from below the price of the current "old" used transport market, and to attract attention. The economic plane prospects would be more likely to purchase aircraft at a price nearer American's figure, giving this way for a better bargaining position on future sales.

Vang Airlines has completed a mutual traffic agreement with Japan Air Lines to facilitate travel between Tokyo, New York and Seoul. Under the agreement, JAL can transfer passengers and cargo to Vang Airlines, and Vang has opened a ticket office in Tokyo. JAL has been authorized to fly between Tokyo and Seoul but has not yet inaugurated service on that route. As a result of the agreement, passengers from Tokyo will be rerouted to San Francisco via JAL, transferred to a U.S. carrier to New York, and then via Vang to Seoul.

Air Transport Assn. has introduced of Civil Air Regulations to require installation of flight recorders on all aircraft flying above 25,000 ft. It is a "lifeline expense." Amendment is scheduled to become effective on Sept. 9. Purpose of the recording, the Civil Aeronautics Board says, is to make certain accidents and incidents are not entirely avoidable due to turbulence and air accident investigation purposes. ATA estimates the cost of the recorder at \$2,500 to \$3,000 each and one must, if not all, information obtained can be gathered from other sources. Since most airlines powered aircraft of the future will fly above 25,000 feet, installation of recorders would be mandatory under the CAB's rule change. Most of the present equipment operated in aircraft would be exempted.

West Berlin Research Committee reports that East Lufthansa expects a deficit for the current year of about \$15 million at the East German and about \$3 million at the West German concession rate. The deficit is based upon the gap between operating costs and income. East Lufthansa figures cost per flight hour of the B-44 at \$120 as those against an income of \$55 an hour. East Lufthansa owns a total of 65 planes of the B-16 type with 20 seats. Fifteen were supplied by Soviet Russia; one was built in Donau, Germany. Nine more aircraft of the same type are scheduled to be completed in Dresden by the end of this year and will be added to East Lufthansa's fleet.

Efforts to streamline international aviation operations to keep pace with jet airplanes will be made at the 14th Congress of the Universal Postal Union in Ottawa. The meeting, which began last week, is expected to continue in September. Delegates from 96 member countries will attend.

White House reports widespread interest regarding availability of aviation facilities planning reports are available at the Government Printing Office in Washington at nominal cost. They include Aviation Facilities Planning, Research and Development, National Requirements for Aviation Facilities, 1956-1975, Air Traffic Volume, Aircraft Characteristics, Forecast of Aviation Activity and Modernizing the National System of Aviation Facilities.

Northwest Airlines plans to begin DC-7C service on its domestic routes between Seattle, Chicago and New York by Sept. 1. Northwest has taken delivery on seven of 11 DC-7Cs on order.



CAPT. THOMAS BARTLEY FLIES 27,000 MILES A MONTH WITH SINCLAIR GASOLINE AND OIL

When Captain Bartley takes off from New York on the Houston run, his Eastern Air Lines plane is doubly protected by Sinclair aviation products. Sinclair Aviation Gasoline and Aircraft Oil are teamed to give the mighty engines maximum power and superior lubrication. Sinclair is proud that Eastern places such confidence in Sinclair aviation products.

SINCLAIR

AIRCRAFT FUELS AND LUBRICANTS

Sinclair Refining Company, Aviation Sales, 600 Fifth Avenue, New York 20, New York

Captain Thomas Bartley of Eastern Air Lines made his first solo flight 26 years ago. Since then, counting military service with the Tropic Carrier Command in Europe during World War II, Capt. Bartley has flown over three million miles. Today on the Eastern Air Lines' New York-Houston run from Idlewild Airport, he flies over 27,000 miles a month (at distance under 2,000 miles greater than that around the earth) every month. Strong Sinclair products he uses on his own run.

SHORTLINES

► **Thera, Airline of Spain**, will begin New York-London service on Sept. 4 using Lockheed Super G Constellation. Enrichment flights will leave New York on Wednesdays for London, Madrid and Rome. Westbound, the new service will depart Rome on Thursdays for Madrid, London and, at the pilot's discretion, bypass Santa Maria, Azores, and fly nonstop to New York.

► **Aeromex Airlines** flew more than 700,000 passengers, 47.5 million passenger miles during July, to set new July records for the airline. Passengers carried increased 9% while passenger-miles rose 8%. The airline also carried more than 7.7 million pounds of air cargo—this added to and over capacity totaled 9,447,000 pounds, an 18.4% increase over July, 1956.

► **British Overseas Airways Corp.** will introduce the Bristol Britannia 400 to its Middle East and Far East routes on Aug. 22 and 23. The turboprop must push will replace Lockheed Constellation on the London-Moscow flight on Aug. 22 on a one-flight-a-week basis and on the London-Cairo route on Aug. 23, also on a one-flight-a-week basis. On Dec. 18, BOAC plans to begin daily service from New York to Nassau using a Vickers Viscount on tourist flights and a Douglas DC-7C on deluxe and executive flights. BOAC reports that bookings for July on the Bermuda, Bahamas and Caribbean routes were 17% above last year and August bookings are up 52%. BOAC's transatlantic bookings are up 53% on tourist and 55% on deluxe and first class flights.

► **Hawaiian Airlines** reports \$1,717 per engine earned during July, a 5.2% increase over July, 1956. Total passenger miles rose 6.9% to 7,256,000. Cargo passenger carried by Hawaiian climbed 20% over July of last year.

► **Seaboard & Western Airlines** will increase all-cargo service between the U. S. and Western Europe to seven flights a week in each direction on Sept. 3. Daily Monday through Sunday service will use Lockheed Super Constellation capable of an 18-ton payload. The Seaboard flight will be made with a Douglas DC-4 which has a payload of one ton. Seaboard & Western reports operating earnings increased 59% in revenues, averaged 11% for the first six months of 1957. Total revenues rose to \$18,591,000 against \$6,148,000 last year. Out of this figure, earnings for the first six months were \$661,779 compared with \$415,977 in 1956.

COCKPIT VIEWPOINT

By Capt. R. C. Robson



Zero-Zero Landings—III

Instrument approaches in the experimental runway lighting project at Andrews Air Force Base during actual zero-zero conditions brought out many interesting facts concerning pilot-traffic operations. We not only learned a great deal about the lights but also several things relating to crew coordination and the need for visual training aids.

As previously stated the weather on the night of April 26 was a pea-soup condition—low fog. We have not yet had the opportunity to fly what Captain Peter Dineen of BEA calls the middle soup run. Admittedly, this was the most interesting—especially when you run into a "mud" at 50 ft.

On our flights the first indication of lighting was from the continuous discharge flashers. They appeared in pairs through the fog at about 250 ft. This provided directional information and an indication of some degree of displacement as well as serving as a "tone out" to the pilot.

More Lights Apparent

At about 200 ft the glow of the fixed burning lamps became noticeable and between 100 and 150 ft the individual barrettes usually came into view. As mentioned in an earlier column we always had a good full view of the runway lights. Generally we began to make the transition from instrument to visual flight in the neighborhood of the 1,000 ft but on the approach lights and at the runway threshold we were definitely "off the page." The 60 ft runway edge runway lights provided everything needed for touch-down—depth perception, attitude information, speed sense etc. Any experienced pilot could have landed without hesitation.

About five years ago I commented on the fact that a low approach was a very messy job requiring close coordination between pilots. Never was this more clearly presented to me than during the Andrews work. My "view" consisted of Capt. Walt Jensen, "co-pilot," and Capt. Bruce Cretell and Capt. Eric Lurie as observers. Here is what we learned.

By the time the approach lights were clear enough to permit visual flying it was already too late for the pilot who was "in instrument" to look out, correct heading and control the landing with ease. The pilot who had been looking out the windshield on the other hand had, even before the mid visual barrettes came into view, picked up considerable information concerning displacement, attitude, drift, speed etc. But there was not enough time left for him to transmit this information to the other pilot. Again we have great agreement that best operating practice may be to have one man act as "visual pilot" the other remains "on instruments." The visual pilot will take over the controls at some point to make the landing, or pull up if required.

British Conclusion

It is interesting to note that our British colleagues have come to similar conclusions regarding this division of duties and are beginning to use this method.

Considering the fact that the Captain is charged with making some sort of a weather observation prior to touchdown, as well as with the landing itself, it seems to follow that he should be the visual pilot and also over all matters. The co-pilot will stay "on instruments," or monitor the approach master, and, until the Captain takes over.

There is also the question of the need for a visual training device. Generally pilots do not have much opportunity to experiment in this type of weather and to work out an agreeable system. Since most of the success in the transition to visual flight rose a reliable runway track as the British say) might help prepare crews for the actual coordination. If the fact about the ground is so late to begin deciding who is going to do what.



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Airline Traffic — 2nd Quarter 1957

	Revenue Passengers	Revenue Passenger Miles	Load Factor	U. S. Mail	Expenses	Profit	Total Revenue Passengers	Per Cent Revenue to Aviation Fuel-Miles
DOMESTIC TRAFFIC								
American	5,514,370	1,343,424	48.8	2,000,149	1,312,829	25,591,641	127,380,368	41.1
Boeing	215,497	322,855	48.8	792,012	247,256	1,094,071	23,458,794	48.8
Capital	1,037,669	420,287	40.1	1,241,747	411,858	1,054,508	41,215,341	48.8
Continental	337,670	47,473	33.2	234,190	52,726	419,622	5,195,277	48.8
Delta	2,025,469	128,264	48.8	1,205,249	822,249	2,321,249	16,849,473	48.8
Eastern	2,036,611	1,138,432	43.77	2,834,739	1,541,438	4,479,287	118,450,143	48.78
National	350,384	395,107	48.8	77,771	146,449	1,204,483	26,448,146	48.8
Northwest	485,472	48,729	48.8	48,729	181,773	2,031,773	26,448,146	48.8
Northwest	333,333	348,418	48.8	1,201,667	499,578	2,402,792	26,448,146	48.8
Trans World	1,185,412	181,669	48.8	2,007,716	1,286,882	2,541,207	381,147,384	48.8
United	1,409,828	1,174,395	48.8	2,064,912	1,535,321	1,400,813	198,196,875	48.8
Western	333,733	119,848	48.7	792,510	321,442	740,792	17,844,426	48.2
INTERNATIONAL								
American	64,170	25,993	48.4	179,239	1,784	883,292	3,402,617	48.2
Boeing	10,468	3,572	38.6	46,171	327,483	2,779,914	36.8	
Continental-Alitalia	32,444	3,472	38.6	4,502	8,389	401,418	37.48	
Delta	14,246	20,158	40.9	22,279	179,734	9,481,484	52.8	
Eastern	85,714	52,477	48.78	102,774	34,819	11,481,140	42.4	
National	19,848	19,817	48.8	33,588	47,815	1,203,341	47.7	
Northwest	31,271	46,267	50.2	2,407,470	22,314	1,912,708	98,043,894	48.8
Trans American	31,884	22,910	48.4	134,212	929,474	3,392,182	48.8	
Alitalia	397,875	418,428	48.8	2,207,898	1,126,202	92,889,746	48.1	
Latin America	276,120	218,120	48.8	1,113,184	10,377,118	42,377,327	38.4	
Pacific	70,610	985,719	35.7	2,894,434	4,418,128	27,280,341	48.8	
Panama	38,618	12,743	37.3	179,779	1,644,247	5,869,414	48.8	
Trans World	75,532	192,226	38.9	3,166,740	3,113,199	84,244,880	71.2	
United	35,217	45,183	48.8	20,930	22,277	4,499,677	48.8	
LOCAL SERVICE								
Allegany	179,796	29,369	48.8	25,632	43,632	48,203	2,646,748	48.8
Allegany	34,742	8,528	48.8	12,568	4,320	22,333	915,915	48.8
Capital	31,381	4,140	38.8	19,264	11,073	30,409	938,700	21.2
Frontier	21,764	14,471	48.8	20,732	31,807	116,132	1,481,684	48.8
Latin America	110,437	30,730	49.1	74,897	521,670	246,793	1,441,802	48.8
North Central	144,894	38,844	49.1	81,722	72,018	2,708,187	47.1	
South	163,127	16,847	48.8	41,894	41,894	1,745,919	48.8	
Panama	110,447	22,840	38.7	40,488	34,704	42,307	2,262,848	48.1
Seaboard	36,221	15,024	40.1	27,317	24,649	1,014,494	48.8	
Seaboard	63,838	27,914	48.8	48,527	15,561	21,569	1,780,234	48.8
Trans-Pacific	61,234	16,477	38.3	41,444	19,641	70,262	1,272,622	48.8
West Coast	47,210	11,819	31.30	13,114	4,321	18,574	1,506,995	38.10
HAWAIIAN								
Hawaiian	121,738	12,764	37.7	12,268	4,208	247,774	1,741,141	38.8
Trans-Pacific	22,382	7,207	34.8	4,208	—	25,119	399,870	37.4
CARIBBEAN								
American	38,470	145,792	48.8	11,244	49,167	90,844,441	84,223,023	48.1
Boeing	8,381	49,332	180.2	—	—	2,729,477	18,404,723	78.4
Embarcadero & Western	11,201	85,381	85.38	242,109	117,284	15,414,998	89,718,497	79.95
MEXICAN								
Chicago-Wallpaper	11,230	171	32.2	75,917	—	39,948.1	29.4	39.4
Los Angeles Airways	8,618	234	24.14	12,407	4,720	40,413	47.45	47.45
New York Airways	19,111	345	24.2	5,197	2,641	1,700	48,486	42.4
ALASKA								
Alaska Airlines	1,280	2,038	47.4	54,710	—	588,806	779,499	48.4
Alaska Central	14,822	1,262	40.9	11,477	—	12,321	120,871	47.7
Central	18,765	834	47.5	5,440	—	9,170	112,023	47.9
Pacific Northwest	37,074	27,039	34.9	247,448	—	687,914	6,572,328	47.8

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TAXI-DOWN TEST of Trident II checks out one of the two engines of SRPE rocket. Check is a standard preflight procedure.



SRPE rocket engine auxiliary (above, left) is a complete preflight unit attached under curved fuselage between bulkhead. Twin auxiliary set in tail (below). Nevertheless well (above) was engineered to take fragment loading prior to Trident II. Longer one (below) is one of design changes in pre-production series. Small side is a nose compartment ahead of wheel well. Cockpit hood is one-piece steel forging.



TRIDENT PLUS missile is advanced secondary development of this school preflight intercepter. Plus has extended Mach 1.5, only try requirement. Concept called for ground control to target vicinity, target acquisition and firing by pilot.

Trident Blends Simplicity, Performance

By David A. Anderson

Paris-Sud Aeronautics' Trident (named preflight intercepter) is probably the simplest high-performance aircraft ever built.

With component parts numbering only in the hundreds—the wing, fuselage, engines fit different parts—the Trident has excited great interest as a possible "common production" aircraft to be built by an inter-European coalition.

Meek 2 Candidate

Its performance makes it a serious candidate for the job of European local air defense. It has a top speed above Mach 2 and is being limited by law to 10,000 ft. in about two minutes and thirty seconds after takeoff are required at the start of the takeoff run.

Simplicity of the design and its rugged construction make the Trident capable of field operations with minimum equipment and facilities. Its development program, which encompasses

a transition from a piloted intercepter to a ground force platform capable of being passed through its planned phases on vehicles.

In spite of all this, the future of the Trident is uncertain. In this respect it shares status with its cousin in France, technically brilliant in concept and design, but with little hope for a future as high-volume production aircraft. Lack of funding in the end will always defeat technical ability.

Under a batch of ten preproduction Trident II interceptors is underway at Sud Aeronautics. There are six follow-on orders. These ten, designed to replace military interceptors, are to be used for service testing by the French. Another 144 is a variety of simulated missions and operations in the field.

If all goes well, a production order for the Trident II or a later Trident III could be the outcome.

Recent sides for testing of the Dassault Mirage (AW July 22, p. 32) plus preproduction aircraft may mean that the piloted version of the Trident

will never see service and that the non-pilot will skip over any intermediate models directly to the ground-to-air mode.

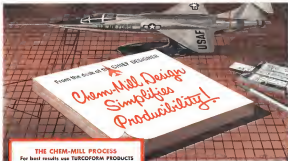
Regardless of the technical differences between the two airplanes, both have been designed to the same basic concept of the named preflight intercepter.

Longer Turboprop

The Dassault airplane, with a propeller-driven engine, turbojet and turbofan engines, can double as both intercepter and light bomber. This cannot be done by the Trident in its present form.

Versatility can have been a major reason for giving the nod to the Mirage instead of the Trident.

Regardless of the outcome of the test program and the fate of the plane, the Trident is unusual and worth detecting from an engineering viewpoint. As first planned it was built as a single-seater with a cockpit added as an afterthought. But a clear look under the smooth skin shows an integrated design



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of a simple airplane designed for a single mission.

Original concept of the Trident missile was that of the "out-shot" interceptor, a light aircraft with a terrific rate of climb and high aggressive speed coupled with the ability to maneuver at extreme altitudes. The mission profile was a simple one—meeting head-on with the target.

The Trident was to take off and climb toward the target, vectored by ground control. It was to get to its altitude preferably above the bombers and on spotting, they would pull out one and go after it. With a single outburst missile, it could only strike once, and then return to base.

Engine Program

The chosen formula for this task was that of the atom-powered concept, a combination of turbojet and rocket power. Trident would be on the combined concept. Once in the air, the rocket would be shut down and the turbojet began on turbojet power alone. At an altitude near 50,000 ft., the first chamber of the rocket would be cut in, followed by the second and third in a programmed firing sequence usually optimized for maximum performance at altitude. After jettison of the target and firing the return missile, the rocket could be shut down and the Trident would return to base either in a glide or at low cruise thrust on its turbojet power. Landing would be by turbojet power, or could be with no power at all.

During childhood years of the basic considerations as a philosophy, Douglas or Lucien Servotte began work in 1948 on the Trident system. His aim was to develop the interceptor through a well-defined, step-by-step program, which first phase was to be the Trident I, a development prototype with a speed of Mach 1.5 and a target date of 1954. From that point, the design team would progress through phased studies in various in a piecemeal mode, should first be the optimum altitude stage.

Trident Layout

Servotte's original layout is almost duplicated in the Trident II today. The plane is simple. Its geometry can be drawn almost entirely with compass and straightedge.

Forelegs of the Trident is combination of optical profile with a linear rise at about 15°.

All sections through the fuselage are circular. Overall length of the Trident II is 42 ft. 5 in.

The pointed nose contains a small radar unit and, during the final phase of the attack. Behind it is the nose landing gear well. The cockpit is covered with a one-piece steel forged hood of exceptionally rugged construction to



DEEP TEST MODEL of Trident was used to check nose section system and sensitivity at tests of engine escape scheme. Capsule also has been abandoned, replaced with simple engine unit.

take the large pressure differentials at extreme altitude.

The whole forward fuselage section, from just aft of the cockpit was designed originally to separate in an escape capsule. Tests demonstrated this idea and, since the Trident has a conventional ejection seat.

All of the cockpit, the fuselage is full of fuel and accessories, linkages for controls, the hydraulic system etc. The S2FR-511 rocket capsule is in the full crew, and its lines, valves and other components are in, the rear fuselage belly under a shallow curved large oval-shaped bulkhead which serves to raise the fuselage wing sections. The rest of the belly contains a simple a non-lubricating lining.

Straight Wing

Trident II's wing is of constant chord and has low aspect ratio, approximately 2.5, but this low figure is achieved aerodynamically because the turbojet engines mounted in nacelles at the wing tips act as an end plate. First span is at about 175 in. aft of the nose and the rear span is about 75 in. Wing thickness is a constant 6%.

Trident wing load is about 75 psf. The figure decreases to 41 psf for the land wing.

Engine wing section, between the span is a full frame-cowling structure. Its double end fuel lines and electric conduits for outboard services run along the front face of the front spar. Eight air-intakeable leading-edge suction tubes on the front spar, those of solid honeycomb.

There are no ailerons, but the trailing edge of the wing is a full-span flap. Wing is structurally not pure

straight through the fuselage. Its total span including the tip is about 22 ft. 6 in. or 6 ft. that, only about 17 ft. is actually wing.

All stability and control about the front axis is furnished by a pair of dual surfaces, one vertical and the other two at about 75 deg. below the horizontal. These surfaces operate together or differentially to provide maneuverability. They are also low-aspect ratio surfaces, and are tapered in plan form with swept leading edges. The surfaces are 6% thick at the root and 4% at the tip.

Immense Jacotet-Leduc servo-control are used.

Speed brakes are located on the upper fuselage between the vertical tail and the subhorizontal surfaces.

Landing gear is built by the French Snc-Mestre.

Powerplant Growth

In the first prototype Trident, turbojet used SO 5000 G, the wingtip turbojet were Turbomeca Marboré rated at 600 lb thrust each. The rocket engine was the three-burned SEPR 40F which developed 9,500 lb thrust or about 5.5 times the turbojet thrust. Total thrust available was about 11,000 lb when approximated the takeoff gross weight of the Trident I.

Reengineered with Dassault built Armstrong Siddeley Viper of 1,870 lb thrust each, the prototype had its present climb performance and growth improved ability for approach and land use.

It also had single-engine performance, something that was almost lacking in the original configuration. Rate of climb to intercept threat de-



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Four Trident Versions

Basic Trident design now includes three versions:

Trident I (SO 9084), the prototype development aircraft. This plane is credited Mach 1.6 in level flight. Its primary purpose was to develop new design data through flight tests for the Trident design and to get each other flight test data as possible.

Trident II (SO 9090), the pre-production aircraft. These are designed to a military requirement and will be used in fly simulated missions and in other development work. This version is not now scheduled for a cruise, but several models have been considered and some flight-tested on the Trident II. Design changes from the Trident I include a modified canopy, lengthened landing gear, lengthened fuselage and replacement of the SEPR three-barrel rocket by a two-barrel unit. Speeds in excess of Mach 2 are expected from this plane.

Trident III, an advanced Trident II. Some difference here is that the subjets are to be more powerful and will have afterburners. This will require more internal fuel, so that the fuselage might be expected to lengthen again. Landing gear is to be strengthened for operations from unpaved fields.

Trident IV, reported as a pilotless version. This description is a provisional one and probably refers to the development of the Trident in which the jet is replaced by a guidance system.

crucial to put below three. Thrust weight ratio improved somewhat.

With the change to the Trident II prototype, the technology because of vacuum Duralumin heat Viper development 1,760 lb thrust each. Also on the Trident II, the rocket powerplant was changed to the SEPR 531, basically the same engine but with two thrust chambers instead of three. This dropped the rocket thrust to 6,600 lb. Ratio of rocket to turbojet thrust decreased to less than 1.9 Total thrust was 10,820 lb, which is under the current thrust weight of 11,355 lb.

Future Tridents will have still more turbojet power, using the Turbomeca Canard speed at 2,470 lb each in the new Trident III. Ratio of rocket to turbojet thrust there will be less than 1.4. Later versions of the Trident III are expected to have Canard with afterburning rated at about 3,510 lb thrust each. This will drop the rocket-to-turbojet thrust ratio to 0.9. The thrust weight ratio should increase to about 1.8.

The SEPR Rocket 531 are powering the Trident II is barbed to Sub A-100 as a complete guide. Adding up the full engine section of the fuselage

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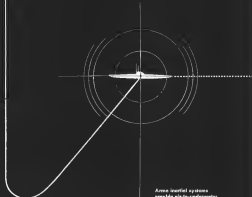
* Superbolts such as A-256, Inconel X, 2-6-6, In-52, #1570 Inconel 700 and Hycron 25 are now available in the Hi-Torque configuration. Precipitation hardening stainless steels—AM 358, 17-4PH and AM 355—are also available.



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up a pre-shaped construction of the fuselage along the belly. Lanes, valves and other fittings run along the fuselage belly as this has completely isolated from the fuselage structure. This design layout was adopted as a safety consideration to ensure the pilots who were not pleased in the thought of loose and leaking in the pressure structure.

The 431 engine is basically a modification of the engine in the prototype Trident. That engine had three combustion chambers and developed a thrust of 3,150 lb. per chamber. The test could be repeated in one two or three stages to develop programmed power between 3,300 and 3,900 lb.

The 431 engine was cleared for flight in 1954 after an intensive ground test program. Subsequent testing in flight 120 hot runs during flight and more than 3,700 static ground runs. The combustion chambers were ignited about 2,700 times during these tests.

Reckless Change

The 451 concept was terminated when the Trident II began its period of the development flying with the 431 engine. But the prototype Trident had in then exceeded a speed of Mach 2.5 at about 53,000 ft. altitude.

Proposals for the SR-71 engine are more and in the engine is 95-98% concentration and further in fuel. In addition to fatigue for these propellers, space must be made for water and seawater which are injected into the combustion gases driving the turbo-jump turbine.

When Scavorty started with the Trident program in 1948, his project team had completed the design of the 50-6038 "Popcorn" (Scavorty) and were looking for a new engine. The Trident concept began to take shape on the drawing boards and in calculations on paper. The French Air Ministry accepted Scavorty's concept of an experimental aircraft fast to prove certain principles and reinforced the work on the Trident I. But there was no experience with methods available in France to get the kind of aerodynamic test data that Scavorty thought necessary.

Wind tunnel tests were made in the United States on the first Trident layout using the Navy Ordnance Laboratory tunnels at White Oak, Md., near Washington.

These were steady state tests, and in addition to giving the main parameters of lift, drag and moments were also made with deflected control surfaces to get complete long-term curves for stability and control calculations about all three axes.

Actually no transient data became available before the flight test program, and the Trident flew through Mach 1

MICRO-BEARING ABSTRACTS

by A. M. DANIEL, President
New Hampshire Ball Bearings, Inc.

BEARING FITS AND FITTING PRACTICES



As shown in Fig. 1, the fitting of Micro-Bearings (the fitting of large ball bearings, clearly involves the clearance between the inside diameter of the housing and the outside diameter of the bearing, the face of the bearing and the shaft diameter.



The achievement of the desired fit by dimensioning is illustrated in Fig. 2. The bearing ID as represented by the top line and the shaft OD as represented by the lower blocks. Such a kind diagram would also be applied to housings and bearing outside diameters. In this kind diagram, it will be noted, bearing ID is represented by a 0.0015 tolerance with a positive clearance for the shaft. A tolerance fit of less to 0.0005 least is shown.



1. An interference fit not tighter than 0.0005 in. is suggested in the following cases:
 - a. Difficulty in assembly. This is often most noticeable when the assembly operation and may result in local bearing distortion.
 - b. Reliability in radial slip.
 - c. Greater or lower misalignment to provide proper geometry of mating shaft or housing.

TOLERANCE DISTRIBUTION

The maximum 0.0005 least condition shown in Fig. 2 may be necessary in some applications. This fitting problem then involves itself in making this extreme, and get maximum the maximum tight fit of low fit. The tolerance may be reduced by reducing the shaft to 0.0005/0.0015 as shown in the block diagram, Fig. 3.



FIG. 3

If the frequency distributions of shaft and bearing ID and OD are statistically normal, the spread of all parts would be 0.0015 zone. Accordingly, an improvement percentage of parts would be noted in the extreme values, and the practical purposes could be improved.

With regard to housing's outside diameter and bore, however, percentage of the distribution curve cannot be assumed. During the preceding operation, the "best match" tendency tends to show the frequency distributions for bearing ID's and OD's in the direction of most match.

In grinding and finishing shafts and housings, uniformly skewed distributions could occur.

Operation on a modified probability distribution of tolerance is possible if the volume of parts is available. But the approximate distribution of shaft and bearing sizes must be verified if that method is to be used.

MATERIALS AND SURFACE FINISHES

The ease of assembly is also affected by materials and finishes. The following factors must be considered:

1. The grinding characteristics, hardness and ductility of the materials involved.
2. Finish job patterns produced by grinding and tool marks and surface finish.
3. R. R. R. surface finish values.
4. Geometry of shafts and housings as regards self-assembly.

The possible combinations of these elements in any single application give no measure that their effect can only be ascertained by trial and error, or by a detailed study of operations on individual applications. A more complete discussion of fitting problems, including methods and mating, is found in our cheap handbook.

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without the benefit of corroborating observations. Much later word arrived which has been done in French factories, however.

Test Vehicles

Some of the basic components of the Trident were tested in prototype form on the Republic prototypes. The complete power control system for Trident, using its reversible fuel-injection system, was flown on the SO 6021. The original prototype SO 6021 was modified to carry Turbomeca Malbecs at its wing tips to check out the engine functioning and engine response under such a layout.

The SEPR rocket was flight tested on two of the prototype Republics, the SO 6025 and 6026. The former had a single-chamber SEPR 25 rocket in a pod and its belly. The latter used an SEPR 251 engine mounted in the fuselage tail.

Overhead dynamically similar models of the Trident were made and air-dropped to check separation, trajectory and recovery of capsule nose-section.

The first prototype Trident I flew on March 2, 1957, with Jacques Cougnaud at the controls. Only the Malbecs propellers were used during the early flight test program. On August 30 that year, Cougnaud had just gotten the Trident off the ground and was beginning to climb when the turbojets cut out.



Fan for Jetliner

The lightweight, high efficiency fan, developed by General Dynamics Corp.'s Electric Fan Division for installing the modern powerplant in the jetliner, will also see service in the air conditioning system of the jetliner being built by General Dynamics' Control Division. The modern fan incorporates detailed blades for maximum efficiency. Unit, including a 18 hp electric motor, weighs about 22 lb and has a 74 in. diameter. The new fan is now being supplied to the Hamilton Standard Division of United Aircraft for the EPR or condenser air system. Patent shows fan being checked for smoothness operation with a vibration probe while riding on an isolated ball-bearing steel disk.



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USAF Aircraft in Group Portrait

Major combat and support aircraft of the Air Force are photographed together at the Air Force General Command, Eglin AFB, Fla. Clockwise around the outer circle, starting with the Convair F-105 intercepter (center, bottom) are North American F-100, Martin B-27 bomber, Douglas B-66 bomber, Boeing B-47 bomber, Boeing KC-135 tanker, Boeing B-52 bomber, Douglas C-124 transport, Boeing KB-50 tanker, Lockheed C-318 transport, Convair CV-440 amphibian, Northrop F-40D intercepter, McDonnell RF-101 reconnaissance aircraft. In the inner circle, starting at the lower left, are North American F-54D intercepter, Lockheed F-94C intercepter, Lockheed KC-130 reconnaissance (tailer), Convair CV-440 amphibian, Convair F-106 fighter, Boeing KC-97 tanker, Republic F-84F fighter, North American F-86 fighter. In the center are a Sikorsky H-34 helicopter and a Cessna T-37 jet trainer.

Guessed not to be out to a crash and was seriously injured.

Charles Gorman took over as project pilot and made the first rocket-powered flight on September 4, 1954. The prototype took Mach 1 in a climb in April 1955.

Meanwhile, the early success of the flight test program and its single turbojet engine produced two effects. First, the French Air Ministry became convinced of the value of the concept and ordered the Trident II to be a military aircraft, specifying a top speed above Mach 1.5.

Second, the prototype was re-engineered with more powerful turbojets, the MD 30 (Dassault-Breguet) Vixens.

The Vixen-powered craft made its first flight on May 22, 1955. The Trident II was converted to 14 turbojets and made its first flight on its turbojets only on July 18, 1955, with Gorman flying. Its last rocket flight was later that year, on December 27.

Guogard, recovered from his early accident, got back into the Trident program again, only to lose another glow before he was shot on Jan. 5, 1956, while he was making a landing. The Trident landed short of the runway, and another

Guogard saw the plane was badly hurt. Nineteen days later the Trident II went supersonic for the first time.

Seven days after it had topped Mach 1.5 with the Vixen-powered engine, leaving no doubt that the top speed of the Trident III coming along will be over Mach 2.

The spring Gorman was killed during a test flight. The official investigation is still going on, waiting, among other things, for the launching of the field trials which the bulk of the airplane fell after having a major fire. And such should produce a surprise of the rest of the wreckage.

And another was that it was not the rocket-powered engine that caused the tragedy.

Trident III

Current status of the Trident program is that the Aviation is completing the third and fourth phases of the test program, including on-site. Following these could come the Trident III, estimated to be the earlier model but with strengthening Gorman engine at the wing tips, more fuel and improved gear and engine fuel capability.

The third Trident II is now on the

floor having Gorman engine without afterburners (not installed) at its wing tips. Thus it will become a reconnaissance aircraft. It will be used in a parallel test with the Trident II to get some comparative data.

Flight time in the program is well over 250 hr., a large figure for a rocket-powered aircraft which requires a great deal of maintenance. Most of this time was taken up in the Trident I before its flight test and development program was terminated as soon as the Trident II.

Whether or not the Trident III gets a production order, or whether or not the missile development goes ahead depends on the political and fiscal situation of the United States. At this stage, French engineers could have developed the world's first complete weapon system and not build a single aircraft.

So it is with the Trident. Versatility, of course, is not a drag on requirements, and it is difficult to build into an aircraft of this type. Scraps and the old engines and the old engines in the project team have accomplished a considerable design program, but its future is cloudy.



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Convair Converts Test F-102As To Latest Tactical Configuration

Ft. Worth—Convair is in the midst of a modernization program here in which early models of the F-102A intercepter are converted from test versions to the latest tactical configuration.

In this Test-to-Tactical program, Convair takes F-102As which were built for test purposes and converts them to the current tactical version. This conversion includes all production changes made since the test F-102As left the factory.

First delivery of a revamped F-102A was made last month to the 435th Fighter Interceptor Squadron at Kirtland AFB, Michigan. Work on the initial batch of 15 aircraft is well along toward completion.

Program Features

Convair and the Air Force are negotiating contracts on further groups of F-102As to be phased into the program. Test-to-Tactical models occasionally evolve into a regular IRAN program for the Convair aircraft.

These are the major changes made in the F-102As going through the Test-to-Tactical program:

- Larger tail is installed
- Radar system is modified and improved
- Missile bay doors are modified to handle larger rockets
- Air inlet canopy is extended to eliminate cockpit noise problem
- Automatic light control system is replaced with a newer version

Although these features in the modernization program all coincide with changes made in the F-102A production line since the ships left the factory, some of the features were actually produced in the Test-to-Tactical program before they appeared on the production line.

The program covers some slight variations in speed and in certain flight attitudes which existed before the aircraft were modernized. The larger tail (AW April 8, p. 128) and the "boom" on the air inlet cone also improve the stability of the aircraft.

Test Equipment

F-102As sent to Convair's Ft. Worth loaded with test equipment for the run on programs they were assigned to when they left the factory. Convair removes all the test gear—measuring equipment, recorders, strain gauges—and puts the aircraft through a program designed to produce an aircraft similar to the latest production model.

Test-to-Tactical program is very similar to the SAMSAC program recently

phased out at the Convair-Ft. Worth plant.

In SAMSAC, Convair modernized B-46s to keep them on a par with the latest production version.

Installation of a new tail is the major structural change in the Test-to-Tactical program.

The old tail is completely removed, leaving only the span which are cut off 22 in. above the fuselage. Shattered

spans in the new tail are then spliced on to the old spans, giving the F-102A a new tail which is 36 in. taller than the old one, including fiberglas tip for antennas.

The new tail includes a larger radome and larger air brakes. Air brake surfaces are about 40% larger than the older version, and the brake area needs a signal to the automatic flight control system which automatically raises the elevators, converting a warning for the loss of the F-102A to drop when the air brakes are opened.

Some beefing of the aft fuselage is done to handle the stresses and strains of the larger tail assembly, including



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de Cadets, a Chateau of the Legion of Honor, he is representative of the men who have helped build the great tradition of French aviation. Today, he and the men who serve under him continue to contribute to that history by keeping dependable air transportation in the grasp of 13 continents.



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Originally the skin of the tail wheel beam on the F4D was made of 4130 steel. By switching to 17-7 PH, Douglas engineers gave the part added strength at high temperatures and made substantial savings in both production time and costs. Stainless formability and low temperature hardening of Armco 17-7 PH greatly reduced distortion eliminated much costly straightening after heat treatment.

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use of bigger rivets and additional clips and fasteners.

Since the design is designed to remove a high frequency noise which has bothered F-102A pilots. An inlet ramp is moved forward a few inches, removing the noise from the cockpit area. The bars fix bulging bulging problems under certain conditions, but it is mainly a cost fix.

Missile Doors

Missile bay doors are modified to handle a 275 in. rocket in place of the older 2 in. model. Exhaust deflectors are added to carry off rocket gases. F-102A can carry Falcon missiles, rockets in bulk.

The aircraft gets a new radar bay to handle additional equipment. The F-102A comes in with the Hughes MG-1 system.

This is modified, and more equipment is added so that it provides three additional modes of operation, making it an MG-10 system.

Each production model of the F-102A goes through the program for modification of these components and the control system. And the old Hughes light control system is replaced with a newer, more advanced system that is directly integrated with the radar and fire control system.

Test-to-Tactical program is providing the F-102A with fittings under each wing to accommodate external fuel tanks.

These fittings have also become standard on production F-102As, and the wing tanks will be used for long range ferry missions.

Slater system in the F-102A is changed in the modernization program, replacing the old winch drive system with a combination type for easier status. Some operational F-102As have had trouble with loss of starting.

Major Inspection

Most of the aircraft going through the program are new to major inspection, and this is done as part of the Test-to-Tactical program. If a F-102A comes in with a 17-4 PH engine it is removed and replaced with the 17-33, which is the current production engine and old aircraft are replaced with new engine.

Test-to-Tactical program changes the interceptor to a fighter engine system for the pilot, and in categories internal release for the canopy is required.

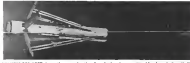
The F-102A mine at Ft. Worth for the modernization program with all kinds of joint jobs, depending on the bases and programs to which they have been assigned. Various sensors all point and given them the standard Air Force gun, joint jobs with standard engines.



TWENTY-THREE INCH LONG cylindrical projectile enters the KCMV nose cone model out of the KCMV gun just shown. It is to be used in KCMV test methods development project by the GE Missile and Ordnance Systems Dept., Philadelphia, Pa. The dotted line at the nose cone was added by Aviation Weekly artist. Note segments of nozzle that behind cylinder.



NOW 70 IN. from the nozzle the dog hooks have been popped and separation between model and cylinder accomplished. Note the springs which caused the dog hooks to pop. Velocity is approximately Mach 2.5.



SIXTY-SIX INCH from the nozzle, the dog hooks have automatically closed the 40 in. cylinder down while the 36 in. diameter nose cone goes on in fire light ahead.

Nose Cone Vehicle Tests Planned

Preparation for a series of free flight model tests of ballistic missile nose cones have begun at General Electric Missile and Ordnance Systems Department.

Tests will be conducted at NACA's Patuxent River Research Station, Wallops Island, Va.

Part of the GE preparation has included development of beam rifles which will separate from the nose cone after they have been heated up to test speeds. A simple way to achieve this separation is by popping heated dog hooks, actuated by inkjet system, into the airframe. Sequence photos showing the results of a launch, tests of this method were made by GE at the ballistics range, Aberdeen Proving Ground, Md.

The 60 inch diameter model of the test setup was fired from a special gun

and the result recorded by a series of cameras along the trajectory.

In the actual free-flight tests at Wallops, one of the latest models developed for Mach 10 and shown in NACA (A-51) (p. 52) would be subjected to the fire each gun. The nose cone model will be over one foot in diameter, large enough to house the equipment to measure back, information on temperature, pressure and noise. With these tests, GE will be able to obtain more reliable data on the surface conditions on the attachment of its nose cone designs.

Though it has been possible to obtain laboratory information on nose cone aerodynamics free flight tests are needed for the airframe. In shock tunnels the aerodynamic of the nose cone will and the pressure of the airframe shock have prevented GE from looking about

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Abdoly Rev. The limited distribution and concentrated nature of the planes jet "Fares" has made interpretation of its results for shareholders tedious at best.

GE would not elaborate on the type of free flight being planned at Woburn. It was not stated whether the dog bites would be sprung on the way up or whether the least trouble would come the more up above the earth's atmosphere layer to insulate security from true ballistic missile threats.

The nose of the next cone model and in those Mach 2.6 preliminary tests has been scratched not to show a flat "sawtooth" shape. The shape which remains has had a typical blunted cone shape dotted in by Aviation Week's artist.

Though it would appear that the reduced diameter of the afterbody is a mechanical compromise for reducing the nose cone into the main missile, a GE spokesman said that this step down in diameter aids in some cone stability by providing a flow separation point. Obviously it is important that the afterbody of such a shape store the least heat resistant nose directly into the atmosphere upon hypersonic re-entry. Though cone stabilization or perhaps rocket nozzles might be used to keep the blunt nose pointed forward, a nose cone expert has told Aviation Week that the short support from a rocket afterbody is able to keep the nose cone properly steered. Other nose cone heat had typical afterbodies after the diameter step-down so that they resemble stable, occupied re-entry cones. It is possible that their tip-like shapes have purposely been used to move the nose cone to a stable and distribute the re-entry heating over a larger surface.

Afterbody cone is being studied because the high frequency plasma flow stream could be shielded by turbulence has been found to affect both internal electronics and to cause fatigue of the surface materials.

U.S., Canada Join In Defense Staff

Washington-New York U.S. and Canadian staff to organize our defense operations and planning will be set up at Colorado Springs, Colo.

Gen. Earl D. Partridge, Commander of USAF Continental Air Defense, will head the staff. His deputy will be Air Marshal C. Roy Skowron of the RCAF. Work of the new command will be confined to operational control of all defense forces in North America and the preparation of joint plans and procedures.

USAF and the program will have no

effect on the information or presentation of equipment. Present cooperation, under which aircraft in the F-86 is built under license in Canada, will be continued.

Warheads of Glass Arm Nike Hercules

Ames, Calif.-High explosive warheads with bodies made of glass fabric covered with steel are being produced for Nike Hercules ground-to-air missiles by Aerojet General Corp. under a \$1,465,500 Air Materiel Command contract.

Accepted designed, developed and tested the warhead, which it said embodies new principles of fragmentation as well as construction. Use of glass materials and techniques not used in this purpose before allowed considerable weight savings.

Nike Hercules has greater capability for apparently will be in secondary or high explosive configuration. Future Aerojet liquid propellant main engine in Nike Hercules has been changed to a Thiokol solid propellant motor to eliminate storage and handling problems. Thiokol also makes booster motors for Hercules. Aerojet continues to make the liquid motor for Nike Apis.



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"NO-MAG" cable is made from type 303 stainless steel. It resists non-magnetic alloy severe cold working—in resistant to standard stainless steel aircraft cable which shows a pronounced increase in resistance after swaging, wire drawing or similar operations.

The non-magnetic property of "NO-MAG" cable eliminates instrument interference from cable magnetism.

CORROSION RESISTANCE...

New "NO-MAG" cables have corrosion-resistant qualities similar to, but slightly better than, cables made of standard stainless steel.

GOOD THERMAL CHARACTERISTICS...

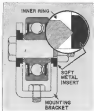
The thermal expansion characteristics of new "NO-MAG" cable are much closer than those of standard stainless steel or carbon steel cables.

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ACCO



Bearing Development Reduces Looseness

Aircraft ball bearing development which cuts axial looseness due to clearance between balls and bore in the inner race is being incorporated by The Talley Bearing Company on balls being used on Cessna's B-18 Hustler super-sound biplane.

Cessna needed a control system that was as tight as possible and with little or no play. To help achieve this goal, Talley developed a series of ball bearings on whose inner rings were no chisel edges, wedge shaped edges on ceramic balls and slightly larger in diam. than the ball hole. The sharp edges extend outward and dig into soft metal inserts in the bearing mounting bracket so as to be tightened down on the bearing support bolts, thus eliminating any slope between bearing and bracket.

Design is used when radial movement of mounted bearing must be at a minimum as in low friction control systems using push pull rods or wire linkages.

Japanese Kappa 3 Reaches 2,420 Mph.

Two-stage rocket—the Kappa 1-3s has been successfully test-fired, giving the way for forward triggering of the supersonic Kappa 4 rocket in the 1st test series. Completed Your observation program.

The Kappa 1, 3s, from Michikawa Beach in Akita Prefecture, northern Honshu, toward the Japan Sea, climbed to a height of 17,500 ft with a maximum speed of 2,420 mph. The rocket is 10.5 in. long and weighs 174 lb.

Experts of Tokyo University's Institute of Industrial Science who fired the Kappa 3 believe the more advanced Kappa 4 will be able to reach an altitude of 197,500 ft in testing at speeds up to Mach 5.

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Skill and knowledge and experience, especially experience, are decisive factors in juggling aircraft designers to turn to Fafnir for help in solving bearing problems. Through continuous research plus collaboration with aircraft design engineers for nearly 30 years, Fafnir keeps in step with aircraft developments . . . and produces bearings to meet the needs. Super precision jet engine bearings are typical examples. The demand for these Fafnir "specialties" has skyrocketed. The Fafnir Bearing Company, New Britain, Conn.

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FIRST . . . at the turning points
in aircraft design



C-46 Changes Begin to Meet CAB Ruling

By Craig Levin

San Antonio—Long-awaited modification of the C-46 to meet Civil Aeronautics Board requirements is finally getting under way as C-46 operators begin large scale installation of Super 46 modifications kits developed for them by the Aircraft Engineering Foundation.

Howard Aero Service has launched an installation program for C-46 operators here at San Antonio International Airport, and several others are installing their own in an effort to remove the stigma of CAB deadlines which have been hanging over their heads for four years.

Kit Prices

Aircraft Engineering Foundation has engineered a kit which meets the CAB's requirements and which was certified in April, 1956. Fleetfoot Aircraft Co. manufactures the kits, and they are sold through the foundation for \$32,500 to members and \$37,500 to non-members. Howard is charging \$36,000 to install them.

Basically, the Super 46 enables the airplane to meet modern fire protection and fire extinguishing requirements, and it improves performance by increasing power but through short-circuiting in the standard C-46 engine installation.

Foundation has secured a supplementary certificate covering all existing C-46 models A, B, C, D, and F. Present certification will accommodate either the R2800 B or the R2800 M-1 engine. L. B. Smith Aircraft Corp. of Miami is currently in the midst of a program to recertify the Super 46 kit for the R2800-C engine. When this version of the kit is certificated, L. B. Smith will have exclusive sales rights to it.

Other Modifications

Before getting into the Super 46 program, L. B. Smith had sold several CW-207s, a major overhaul and major expensive modification of the C-46. Radley Aviation has also developed a modification designed to bring the C-46 to CAB requirements.

Foundation Super 46 kit is available to any C-46 operator. Although Howard Aero has turned out the first production modification and is working on more, kit purchasers are free to work it themselves or have the job done by any field base operator they choose.

Major problems on the C-46 is inadequate engine cooling, especially in emergency conditions. Kit solves this problem with a number of changes, including a smaller cowling, a new pro-



FIRST production modification from C-46 to Super 46 was done by Howard Aero Service for All American Airways. Changes are smaller cowling, new propeller spinner, external air scoop on nacelle top for carburetor and on bottom for oil cooler, and the exhaust tubes from the exhaust system protruding at an angle from the sides of the nacelle (below).



pellor spinner and a new exhaust system.

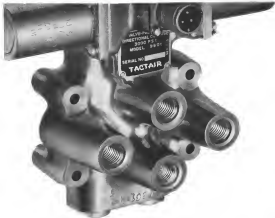
New cowling is mated with external air scoop for the carburetor and oil cooler in place of the standard egg shaped cowling with its built-in air scoops. New cowling, with the new spinner and afterbody is designed to provide the correct amount of cooling air to the cylinder banks in a stream line flow.

Once the air passes over the cylinders, a new auger-type exhaust system provides a vacuum effect to enhance the stream-line flow. The auger-type system uses auger-type tubes with Ramjet exhaust stacks in place of the old oil cooler ring system. Left of the auger

tor stacks, made by Ledette Co. in Foundation specifications is substituted at 5,000 lbs., compared with the 1,000 lb. life of the old exhaust system.

New air scoop on top of the cowling for the carburetor system channels the turbulence of the old built-in scoop to stall out at higher angle of attack. It also increases the optimum cruising altitude about 1,500 ft.

Structural problems with the old cowling in the old nacelle is solved by installation of an external scoop on the bottom of the new cowling which gives a constant flow at all angles of attack. Another major problem with the airplane has been inadequate fire protection. Super 46 kit solves this prob-



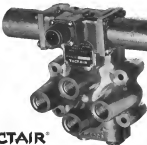
Note: to precision-minded men at MARTIN

It's one thing to design valves of unusually light weight—another to engineer them for high reliability under extraordinary conditions. At Tactair, we've learned to do both, not occasionally, but consistently. Result: a growing reputation among leading aircraft designers that they can insure their toughest valve problems to Tactair.

Case in point: the 4-way, actuated-operated, pressure-reducer valve for a rocket engine mechanism. To assure its rapid, dependable operation over a wide range of operating pressures, we considered a number of steel and brass design principles and individually in other metals. And to minimize weight, we made the valve a pilot-operated unit.

Result: an unusually wide pressure range of 508 to 3,000 psi at altitudes from sea level to 70,000 feet. Extremely high flow capacity for a valve this size—actual flow factor of 2.4 gpm. Low leakage—5 cc per min. of free air. Rapid operation—15 sec. max. (optional). And with this, a weight of only 1.9 lbs.

Reminder: on standard or special components, we welcome the opportunity to assist you with your most precision valve problems. Every job we do is done on a personalized basis. It has been that way for 18 years. Tactair Valve Division, Aircraft Products Company, Bridgeport, Pa. 18809 or 3-1006.



CONTROL, SELECT, ISOLATE, RESTRICT, CHECK... with

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with new flow for detection, extinguishing and protection systems.

A secondary firewall is installed between the engine and the accessories section in the nacelle. This provides a sealed accessory section since in the first area, such critical areas as the exhaust system, the secondary firewall, ducting for the oil cooler pump and the bleed air ducts, are stainless steel. Zone 2, the accessory section, is completely compartmented with stainless steel.

Third zone, the wheel well compartment, is protected with stainless steel sheeting on the wheel well doors.

Lower Temperatures

Howard Aero found that nacelle temperatures in the first lot installations were lowered considerably. Takeoff temperature of the engine heads is 220 deg., compared with 250 deg. on the standard C-46. Cruise temperature is 175-180 deg. substantially lower than the old range of 210-215 deg.

These new lower operating temperatures, and isolation of the various nacelle zones with stainless steel add significantly to the fire protection capability of the modified C-46. Howard also points out that protection of accessories and lines from heat sources is no longer a cost on the engine forward of the firewall to about 25%.

New fire detection and extinguishing systems were designed for the Foundation. Individual fuel warning lights for all three nacelle zones are installed on the instrument panel along with a master fire-warning bell. Fire detection operates on both rate of rise and maximum temperature.

Each nacelle has a new high discharge, two-shot extinguisher system built for Foundation specifications by Walco Kolls & Co., Inc. Agent is Brown/Baranowski, and the system will absorb oxygen eight times faster than CO₂. It is also less toxic than CO₂, and the new system is 100 lb lighter than the current CO₂ system.

Ground Handling

Hubel Anti-Shock Brake System is installed to improve braking and ground control handling qualities of the C-46. Super 46 is the only tail wheel assembly certificated for this braking system.

Hydraulic brakes is modified to make both the brake and main accommodation accessible to the pilot without leaving takeoff. This reduces maintenance time to 11 sec. from the former time of 10-20 sec.

Antistrut power line indicators are installed on the instrument panel next to the corresponding feather switches to help the pilot determine whether an engine should be feathered in a partial failure. All emergency controls are centrally located on the panel to pro-

vide more rapid emergency procedure.

As the Foundation's president, Monte H. Stauffer, points out, improvements come from the application of numerous engineering improvements well within the state of the art, rather than from any new technical breakthrough.

Improvement in performance comes from a number of factors. Changes in the nacelle area allowed Foundation engineers to increase about 250 hp in each engine, and the new cowling reduces the static drag about 1.5 in. ft. Finer gear retractors reduces drag and improves takeoff performance. Rate of climb increases about 34%.

Super 46 kit increases cruise speed 12-15 mph to about 212 mph. Gross weight with the R2500B engine is 37,100 lb and gross for the M-1 engine is 47,400 lb.

Kit installation

First production installation was made by Howard Aero on an All American Aeronautics C-46 which was certificated June 29. Howard has modified a second All American ship, but current status for two more from All American.

and not from Western. Other contracts are under negotiation.

According to Jack Brink, Howard's chairman in charge of its engine modification, the Super 46 kit reduced empty weight of the first All American ship from the annual C-46 average of about 12,000 lb to 9,512 lb. Combined with a higher gross, this means a net gain of about 3,500 lb available for passengers, cargo or fuel in the operator's choice. More dual capacity means greater economy because fewer ships are required, and added passenger capacity means more revenue.

Operational Test

Foundation's Super 46 prototype was operated in the Flying Tiger Line system for eight months to test its sea service aspects. During the period of scheduled cargo service on the Pacific, San Francisco-Seattle run, the Super 46 operated at a block-to-block speed of 168 mph, compared to 173 mph for three standard C-46s operated in the same period. In 1,358 hr of operation in the Flying Tiger pattern, Super 46 direct operating costs were 516 1/2¢ per hour less than that of the standard C-46.

Development of the Super 46 kit



Stretch Press Forms DC-8 Skin

DC-8 conventional jet wing skin is formed in large 4-direction stretch press at Douglas Aircraft, Long Beach, Calif. The Shearola stretch press, described in the layout of this ad on page 1, is used to form single piece wing skins from 10,500 to 12,000 lb aluminum plates. As shown, stretch clamps cross-over to the short hold the skin while the press on the right and left draw the skin skins over a stretch die which runs under the skin to form a double curved shape. Holding fixture in the foreground supports the aluminum plate which has been tapered from 0.239 to 0.193 in. before delivery. After forming the part is once again, dented, surface treated and painted before going into the DC-8 wing assembly.

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Over the years Bendix airborne and ground-based antenna devices have been successfully designed by our highly specialized staff of radar and servo-mechanisms engineers to meet a wide

variety of existing requirements. And, of course, world-famous Eclipse-Pioneer precision instrument components are standard in all Bendix radar antenna devices.

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Lightweight, ground-based, all-transportable tracking radars preferred for monitoring 8 to 100 miles; a hand-operated portable reflector and radar for accurate highly accurate 3 speed data systems in elevation and azimuth.



Lightweight, ground-based, all-transportable, dual reflector, multi-band, high gain search systems and control systems. 3 speed data systems and magnetic display devices in azimuth and elevation.



Airborne reflector mounted and also with the antenna antenna 2 speed, dual-band systems in azimuth, elevation, 40° and 60° sector scans, remote control, line of sight indication.



Airborne weather radar antenna with line of sight indication for a or a band, multi-band with line of sight signal beam system for mapping in ground based for storm detection.

Eclipse-Pioneer Division

TELEPHONE, N. J.



grow out of application action taken by the CAB four years ago after the Board had taken a look at the C-46 accident record. Special stipulations was issued regarding that the transport must transport category standards in order to be eligible to carry passengers. Excessives have been granted since then, and the stipulations have been lifted the accident record would be a satisfactory modification because available.

When the CAB changed down in 1953 C-46 operators got together and formed the Aircraft Engineering Foundation. Foundation President Seiderer's job was to find ways of bringing the transport up to CAB's new standards.

Safety Record

When the Foundation launched its program in 1951, the C-46 safety record was under fire. The transport had been manufactured during World War II for the military. After the war, war plus C-46s were turned loose for civil operation without the normal engineering help and guidance outgrowth provided by the manufacturer of a commercial transport.

Looking over this safety record, the Foundation discovered that 45% of the accidents had been caused by a lack of pilot training, and another 30% were traced to poor maintenance. There were shortcomings which could be corrected without modifying the airplane so the Foundation launched a four

point program to cover all aspects of the C-46 situation.

This program concentrated the Foundation to (1) improve operating standards of the C-46, (2) improve maintenance, (3) develop a modified two list, and (4) set an technical sponsor of the transport.

To solve the pilot error problem, the Foundation sponsored the CAA in a program conducted at Tuscon City between September, 1953, and June, 1954. In this program, C-46 pilots went through a two week course designed to train them in the proper emergency procedures for the airplane.

As a measure of the success of these efforts, Seiderer points out that there have been only two fatal accidents with the C-46 since September, 1953.

Fifty Kits Sold

Work on a modification kit for the C-46 was started in 1953, and a certificate was issued by the CAA on April 6, 1956. Fifty of the kits have been sold by the Foundation, and about half of them have been delivered.

With the design and development of the Super modification kit out of the way, the Foundation continues to promote its role as technical sponsor of the airplane. Service bulletins are issued as problems come up, and the Foundation does a continuing job of providing a factory-type technical service for C-46 operators.

■ TUBHUS is industry's most advanced method of testing the true QUALITY of any steel. It's the only method that can be accomplished easily. Not to worry! PERFECTION! When the shape is done and safety, performance, and LIVES are on the line, the kind of testing TUBHUS makes perfect sense all the way and without quality-control department specifications demand.



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TUBE METHODS INC.

REIDSBORO (Hampshire County), PENNSYLVANIA



Stand to Test X-15 Engine

Republic Motor Inc., built the test stand at Lake Donnell, N. J., capable of testing rocket engines in the million pound thrust class in any flight attitude. It will test the 50,000 lb. thrust engine for North American X-15. Engine system for test including tanks are mounted on a large testing base which can be used or lowered by hydraulic cylinder.

AVIATION WEEK, August 18, 1957



LOW NOISE CHARACTERISTIC of atomic amplifiers can be used to extend the range of mobile and satellite tracking radars as indicated partially above. Conventional Maser is shown at bottom of the diagram.

How the Maser Operates: Part I

Maser Shows Promise, Some Drawbacks

By Dr. Richard W. Darrow

Recent developments in atomic physics have produced an entirely new type of high frequency amplifier which holds promise of great gains in the range and performance of radar, electronic countermeasures and scatter communication systems. Another application of the same principle is the atomic clock, an oscillator with remarkable frequency stability (AW Oct 22, 1958, p. 101).

The term Maser frequently is applied to the new device. This is an acronym derived from the principle of operation, i.e., Microwave Amplification by Stimulated Emission of Radiation.

The name sometimes is applied to a very recent development in atomic amplifiers which was termed a giant maser. It is a short name, the newer device is not only a Maser because its operation is based on somewhat different principles—principles not yet understood as fully as those of conventional Masers. However, these newer low-noise amplifiers have operational characteristics which are almost as desirable as those of Masers and both will be covered in this discussion.

Low Noise

These new atomic amplifiers have generated tremendous interest in military and industry circles primarily because of low characteristic extremely low noise.

Ultimate sensitivity of any radio communications or measurement receiver is set by the noise level of the input stage. Noise level of atomic amplifiers can be as little as 1/1,000 that of presently used klystrons and traveling wave tubes. The reason is based

conventional tubes operate by virtue of electron flow, that is, a noise of noise. The atomic amplifier uses no electron flow, exhibits internal energy from unchanged atoms.

The solid state atomic amplifier consisting of a small crystal suspended in a vacuum, cooled and exposed to a magnetic field and source of microwave energy, is basically less complex and inherently should be less costly than traveling wave tubes or klystrons.

The devices provide attractive bandwidth, noise as high as several hundred megacycles. They are electrostatically tuned and can be designed for frequency ranges from 100 mc to 100,000 mc.

These advantages are not obtained without certain penalties. For example: •Extremely low-temperature operation. Most atomic amplifiers, such as the Maser as far as it is also known) per-

formed at near absolute zero (-273°C) temperatures, which require heavy, costly cryogenic equipment. This tends to limit their use to ground based systems. However, the recently developed ferromagnetic amplifiers can operate at room temperature and higher, although lacking the superior noise performance.

•Low power capability. Most atomic amplifiers developed to date have relatively low power output of about a fraction of a watt. Amplifying properties are destroyed at higher power levels. Atomic amplifiers may eventually be used primarily as preamplifiers with conventional vacuum tube amplifiers for output stages. Most agree, however, ferromagnetic amplifiers may overcome these power limits.

•Low noise design. Extremely low noise level of the atomic amplifier in receiver system design problems because all other sources of noise within a system, particularly those of the atomic amplifiers, must be minimized to achieve full advantage of the device.

Hardware Emerging

The atomic amplifier has just begun to move out of the laboratory into hardware development, but intensive research work is continuing. Principles of operation have been established and the feasibility of building useful devices has been experimentally demonstrated.

One indication of the current interest in atomic amplifiers is the number of companies and research institutions now active in the field. These include Air Force Cambridge Research Center, Bell Telephone Laboratories, Columbia Institute of Technology, Columbia University, General Electric, Harvard, Hughes Aircraft, Lincoln Laboratory, Massachusetts Institute of Technology, Princeton, Radio Corporation of America, Stanford University, Varian Associates and Westinghouse Electric.

Out of these widespread research and development efforts, several new types of atomic amplifiers, some of which may be suitable for airborne equipment, much as the original prototype invented in 1948 has spawned a vast family of semiconductor devices with far greater versatility and usefulness than the original.

The familiar sodium street-light illustrates some of the fundamental properties of atomic systems similar to those used in the Maser. In both, no external source of energy is used to create the system of atoms. The atoms release this absorbed energy in the form of discrete quanta—radiation whose frequency is characteristic of the particular atoms used. In the sodium lamp, the atoms receive energy from an electric arc and give it off immediately as yellow light whose wave-



Author of Exclusive Series

Atomic amplifier, or Maser as it is sometimes called, is one of the most significant developments of the post war period, comparable to the discovery of the transistor one year ago. It is a principle of operation that allows amplification without the vacuum tube or transistor, making use of electron spins within the atom itself.

This article is the first of a two-part series intended to give Aviation Week readers an understanding of the operating principles of the new atomic amplifier and some of their remarkable characteristics as well as their limitations.

The author, Dr. Richard W. Darrow, is a physicist in General Electric's Research Laboratories, currently engaged in research in the field of magnetic resonance phenomena and their application to atomic amplifiers. During the author's doctoral dissertation on ferromagnetic resonance, at Harvard, he discovered the important behavior of ferromagnetic materials which forms the basis for operation of one of the newest types of atomic amplifiers.

applies to the characteristic frequency of the sodium atoms.

In the Maser design also is fed into atoms constituting the heart of the device. But the atoms in materials used for Masers have a characteristic frequency that is a lower value than the visible light spectrum. There is no emission, and crucial difference between the Maser and the sodium lamp the Maser atoms do not radiate the excitation energy spontaneously as sodium does.

The Maser requires a means of excitation which must be applied at the characteristic frequency of the atoms. Under such conditions the energy stored in the atoms is released at the characteristic frequency, adding to the strength of the signal radiation and thereby amplifying the signal.

Thus the three basic properties required for Maser operation include:

- Source of energy
- Characteristic frequency of atoms is natural and need be within range of frequencies at which amplification is desired.
- Energy must be stored in atoms until

excitation is stimulated by radiation at their characteristic frequency.

Basic principle of Maser operation can be illustrated by the use of a mechanical analogy. The particular system described is a pulsed solid-state device using a paramagnetic salt, but with energy storage the same could be used for the gaseous type or for CW operation.

Electronic Spin

About 30 years ago scientists discovered that electrons, in addition to moving in fixed orbits about the nucleus, also spin about their own axis. This causes the electron to behave much like a small gyroscope and many of its properties can be explained using the classical theory of gyroscopes.

However, the electron is a tiny compared to a gyro. In addition to its spin, the electron also has an electric charge. This spin and charge is equivalent to an electric current. Since every electric current produces a corresponding magnetic field, the spinning electron behaves like a small magnet.



Directing the talents of outstanding people. Staying ahead of potential aggression requires quality of people as well as equipment.

How General Electric is trying to help meet the increasing challenges of defense

Today Americans are being forced to think in a totally new way about national defense. The United States can no longer expect to build military strength after an attack, but must be ready at all times to discourage aggression and maintain peace.

Yet, at the same time that a stable pattern of research, development, and production is constantly devoted to defense, we as a nation are striving to continue to advance our living levels.

Security with solvency

The resources of the nation are not limitless. Maintaining security with solvency presents a challenge to business and government to make sure that every citizen is getting the most for his defense dollar.

In helping meet this challenge, General Electric is:

- Devoting the talents of nearly half

of the company's scientists, engineers, and technicians to defense activity.

- Bringing to bear its large-scale resources to pioneer new and complicated defense projects... and then breaking down the big jobs into tasks to which thousands of other businesses contribute their specialized skills.

- Trying to conduct defense work in a business instead of an interruption of business.

Toward greater defense values

Meeting defense requirements is a continuing duty of responsible business. General Electric believes, however, that even fuller values from industry participation can be gained by infusing into defense work the same free enterprise incentives that keep the civilian economy vigorous and able to supply good values to customers.

One way is to encourage maximum

flexibility for cost reduction in which both the taxpayer and the producer share in savings; another is to stimulate risk taking by making possible returns on defense accomplishments that warrant greater private investment.



As General Electric sees it, fully utilizing the incentives of a free society will deliver to every citizen greater defense value for his tax dollars... and at the same time continue to provide Americans with the highest living levels anywhere in the world.

To help further public understanding of key defense problems, this message is appearing in THE NEW YORKER, U.S. NEWS, AND FOREIGN AFFAIRS, LOOK, LIFE, TIME, and other leading magazines. For more information, contact your nearest General Electric office.

Nearly half of G.E.'s technical personnel is engaged in defense work, even though it is only about 20% of the company's total business.



Bringing to bear large-scale resources. Typical of complex jobs undertaken by General Electric in development of atomic reactors for submarines (like the Sargo, above).



Mobilizing the skills of businesses of all sizes. In taking responsibility for complicated defense projects, and breaking them down into jobs smaller than one could, General Electric brings together the specialized talents of many businesses. (Here are a few representatives of more than 800 firms which help General Electric produce large order units.



The revolutionary J79 jet engine powering the new F-4 Phantom II fighter-interceptor was developed by General Electric. The J79 is the most powerful jet engine for its weight yet built.

Progress Is Our Most Important Product

GENERAL ELECTRIC

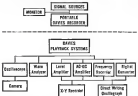


General Electric Company's Charles A. Wood, left, makes analysis of aircraft gas turbine operating signals by using "blast" action on one of two Davis Division magnetic tape units at the Flight Propulsion Laboratory Dept. Data Reduction Center, Eindhoven, Ohio.

how G.E. data center analyzes dynamic jet engine operation

G.E. wants to know a lot of things about an aircraft gas turbine before it leaves the ground. Not just thrust, pressure, temperature, ... but vibration and stress characteristics too. While suitable equipment has long been available for recording and measuring the "static" operating characteristics, only the last few years have seen reliable techniques developed for measuring, recording, and reducing significant information from dynamic signal sources.

A pioneer in the application of these techniques, General Electric's gas dynamic data handling installation at the Flight Propulsion Laboratory Department, Aircraft Gas Turbine Division, Eindhoven, stands as one of the most complete in the country. Originally conceived by the Laboratory, the facility has contributed substantially to the outstanding progress of aircraft gas turbine design at G.E. Data recorded at the test site can be played back through two tape systems at the laboratory into electronic data analysis equipment and analog and digital displays for a fast, detailed picture of gas turbine performance.



The substantial part played by magnetic tape in the system reflects its growing importance as the common denominator of dynamic data handling. Dynamic range and frequency range are several times greater than

characteristic other recording media, and the data is stored "live". The original electrical transducer signal can be accurately re-created whenever desired for observation and processing.

Five portable magnetic tape data recording systems are used by G.E. to collect data at the engine test cells. Signals ranging in frequency from DC to 20 kc. are recorded by Direct or FM carrier techniques. Wide 150" tape carries 20 tracks of data, plus reference frequencies, timing signals, and voice, as desired. Monitoring oscilloscopes permit visual observation of the recorded signals. It is interesting to note that the five recording systems were originally designed by the Davis Laboratories Division for the limited noise and severe environments of aircraft field testing. The strength built into them has proved invaluable to G.E., despite the fact that they have never been operated in the air.

With the completion of a test, the reel of tape is transferred to the data reduction center. Two complete Davis Division laboratory near-production systems permit almost flexibility in feeding tracks of data serially or simultaneously into reduction and display equipment. Oscilloscopes permit a quick look at results through rapid visual inspection; wave analyzers provide a record of frequency components; level analyzers measure amplitude; direct-writing recorders and oscilloscopes examine period; a complete record of wave shape and phase relationships.

Davis Division magnetic tape data recording installation is large and complex as this one at General Electric Company can easily be confirmed with standard package equipment. But for the smaller installations, recently introduced Davis Universal Magnetic Tape Systems are an ideal choice. You can get a good introduction to magnetic tape data recording techniques in general by requesting our Bulletin 1000, Universal Systems are covered in Bulletin 2701, Minneapolis-Dwight, Regulator Company, Dayton Laboratories Division, 10221 Main Street, Beltsville, Maryland. Or call Wilshire 5-2200.

HONEYWELL
DAVIS LABORATORIES DIVISION



BLOCK diagram pattern direction of incoming signal by tone.

attached to a free gyroscope. This model will be used to describe the basic principles of Maser operation.

Gyroscopic Action

If the gyro-susp, representing the electron, were placed in an external magnetic field, like a compass needle it would seek to align its north pole (S) with the north pole (N) of the external field. (Fig. A) This is called the equilibrium position and corresponds to the lowest energy state for the system.



However, because the electron exhibits gyroscopic behavior, the torque exerted by the external field causes it to precess at right angles about the field (see Fig. B). The precession rate, like that of a conventional gyro, is determined by the applied torque and the angular momentum. For an electron, this works out to approximately 2.83 megacycles per gauss of field strength.



This precession rate corresponds approximately to the characteristic frequency of the paramagnetic atoms used in the Maser since the paramagnetism arises from one electron spin on each atom.

Thus a Maser operating over the quantity of 100 mc. requires an external field of approximately 35 gauss, or 35,000 gauss for operation at 100,000 mc.

As the gyro-susp precesses, it develops magnetic energy associated analogous to the way a conventional gyro oriented in a fluid experiences viscous damping. Most of this magnetic energy is dissipated in the Maser crystal in the form of heat, a small amount is given out as electromagnetic radiation at the characteristic frequency. Thus the gyro-susp possesses single degree of freedom and it returns to a state of equilibrium in which its north-south axis is aligned with the external magnetic field.

Energy Absorption

If a gyro-susp is in a state of equilibrium it is exposed to a source of radio frequency energy applied at right angles to the external field, the oscillating magnetic field from the radio frequency source will exert an oscillating torque on the north pole (N) of the external field. (Fig. C) This is called the excitation position and corresponds to the highest energy state for the system.

The wall between the gyro-susp's precession angle (from equilibrium position) and the power absorbed from the oscillating field just equals the power dissipated in the previously described damping mechanism (Fig. C).



This is the process of resonance absorption by a system in equilibrium, a technique that has been applied to a chemical analysis of materials. Vibration does not provide amplification because the gyro-susp starts from a position of lowest energy and therefore has no means energy to give up to the oscillating field.

Getting Amplification

If, however, the gyro-susp can be excited so that its north pole points toward the south pole of the external field then it will be in its highest energy state, an unstable one, and amplification can be obtained.

(Number of different techniques for creating the gyro-susp in its excited position will be described in the second part of this series.)

When a radio frequency source and its oscillating magnetic field are applied to the excited gyro-susp, precession occurs as in the previous example. How-

ever, because the gyro-susp has excess energy which is applied to the radio frequency circuit, it is able to provide signal amplification, and it finally is in a state of equilibrium. Its north pole is aligned to the static field's north pole.



Now the gyro-susp has excess energy which is applied to the radio frequency circuit, it is able to provide signal amplification, and it finally is in a state of equilibrium. Its north pole is aligned to the static field's north pole.

Thus the key to achieving amplification in a Maser is the achievement of the gyro-susp (electron) into the non-equilibrium state. To accomplish this, either power must be applied constant only to the system to overcome the losses which are not being to return it to equilibrium or else the system must be periodically disturbed and then held only during the brief interval (usually a few milliseconds) before equilibrium can be re-established by the damping mechanism.

One of the reasons for operating the solid state Maser at extremely low temperatures is to delay this return-to-equilibrium process.

The technique used for determining the system constitutes the essential difference between pulsed and CW Maser which will be described in a subsequent article.

Paramagnetic Amplifiers

The heart of a solid state Maser is a crystal containing paramagnetic atoms — in fact, atoms having a net electron spin which makes them behave like

Father of the Maser

Discovery of the Maser principle of amplification (Microwave Amplification by Stimulated Emission of Radiation) was credited to Dr. Charles H. Townes of Columbia University. Townes and associates first demonstrated the principle in 1954 using ammonia gas.

Discovery of the Maser, like that of the transistor, did not arise from a search for a new type of amplifier. Rather it resulted from the idea of using microwave instead of visible light in spectroscopy to study the structure of gas molecules.



Messner, General Manager, Engineering, and W. P. Moore, General Manager, Maintenance — engineering and maintenance personnel constant to guarantee top operating efficiency of United's 100 aircraft. All old friends of mine, close three men and their dedicated staff are a driving force here at the SFO base.



MIKE MESSNER

I spent several days at this 125-acre maintenance base housing on San Francisco's International Airport. There wasn't a section of the million square feet of floor space I failed to see — hangars, overhead docks, shops, test cells, storage and stock areas, service and training headquarters, office buildings, even the aircraft repair pads — all of it comprising a "push-button" maintenance base that's recognized as one of the most modern and complete in the nation.

From a small, split operation at Oakland Airport and Cheyenne,

Wyoming, in 1941, United has mushroomed with almost equal steps in the aviation industry. With today's maintenance operation centered at San Francisco, total UAL investment in buildings at year-end stood at \$17,210,900, with 5,677 employees on a \$31,789,602 payroll — just at San Francisco alone.

Current maintenance-hangar construction will boost United's San Francisco investment to \$24,483,900 by June of 1958.

Each year, hundreds of skilled mechanics, technicians and craftsmen work with engineering specialists to overhaul 250 to 300 aircraft and more than 1,500 engines. Maintenance standards and procedures are rigid.

On a precise schedule of preventive maintenance, each UAL Martinair and Caribbeair is tested in San Francisco for progressive overhaul. This is planning at its best. It is one of the most perfectly synchronized operations I've ever seen.

A Mainliner is assigned a hangar. Ten minutes after being squired into position, it is literally caged by cat-

walks, timeworks and platforms. Powerful jacks lift the plane. Overhead cranes move in to support engines while they're dismantled. Control surfaces, instrument panels, wheels and brakes, seats, buffets and all other removable components are "pipelined" to their respective shops. Metal workers, X-ray specialists and inspectors move into the striped hangars to probe for metal fatigue, corrosion and wear.

Meanwhile, the giant engines have gone into the Powerplant Section, where they're steam-cleaned, torn down, and part-by-part chemically washed and minutely scrutinized, in strict conformance to United's rigid specifications.

After engine assembly, valves are adjusted, timing is set and test cells are readied. Then, with the deft care of surgeons, specialists check each engine pale to deathly assure perfection before mating the engine with first liners to form the safe and dependable power package of a UAL Mainliner.

"Dependability is of such great importance," Red Messner told me, "that we cannot afford to use a

second-rate product any place in our operation.

"Take spark plugs," said the top man in Engineering and 24-year UAL veteran. "We've been using Champion as standard equipment on all of our airplanes for more than 10 years. We've tested many different types of spark plugs during that time. We've found, however, that Champion has developed its product more rapidly than anyone else. Result is — since we've used Champion we've enjoyed the best overall ignition experience of our entire history."

As Ben Goodart, a maintenance expert in Line Service Engineering, put it: "We have got to find one engine failure caused by a Champion Spark Plug. With these DC-7 engines costing about \$120,000 each, the importance of product reliability is obvious. Sometimes a spark plug is blamed for trouble when actually it is not at fault. We've found that Champion is the best we could possibly use."

Harry Taylor, Superintendent of Powerplant Engineering, said: "We've always emphasized depend-

ability, long life and economy. We feel that no other spark plug has been so far provided us with a full measure of these requirements."

Frank H. Gregg, buyer in United's Purchasing Department for 19 years, calls it "business integrity and forward thinking on the part of Champion."

Just as a good airplane is the cooperative idea of designer, builder and user, so is development of a spark plug largely the result of cooperation between manufacturer and user.

One of United's technician engineers, Thomas Prentiss, told me that "United has attended Champion Ignition Conferences every year for the past seven. There, airline operators get together with Champion engineers to discuss their problems. The result of such meetings is a spark plug that meets the needs of the airline operator, a spark plug offering greatly improved performance. . . ."

Dependability the Champion way at 1,000 hours of service. New spark plugs are installed in UAL's Pratt

& Whitney and Curtiss-Wright engines and, after 500 hours of operation, they're removed by line maintenance people and returned to the base for reconditioning. Then they're put in service for another 500 hours.

"In terms of operating costs," Bill Pitt, Manager of Line Service Engineering, told me, "Champion has been able to extend the time between spark plug replacements so that we are now realizing as high a time in any airline in the industry. This represents a saving in direct maintenance and is less frequent need for overhauling or service." Mr. Pitt began his career with United 27 years ago as a cleaner and has progressed through all levels of mechanical maintenance. He speaks with authority, then, on economy and dependability.

"Actually, the spark plug is one of our best tools for evaluating maintenance check procedures, such as burned pistons and valves," Mr. Pitt said. "By careful study of the electrode and of a spark plug removed from a malfunctioning engine, we've determined actual cause

Pratt & Whitney extending flight testing of turbocharged aircraft with (left) Robert M. Moore, Test Pilot, W. B. "Doc" Wheeler, Flight Engineering Manager, and Robert C. Cohen, Test Engineer.



"Usually, when material data Messner spark plugs for ignition tests trouble, we find the spark plug is the cause and not the cause," reports Red Messner, General Manager.



"Eye of Fire" spark points up safety, passenger comfort, flight performance, economy, economy. Reported by Red Messner, General Manager, United States, William, Powerplant Manager, Bailey Quinn, Maintenance Manager Assistant.



Describing types of all kinds of spark plugs and reasons for using Champion exclusively the past 10 years are (left) Harry Taylor, Powerplant Engineering Superintendent, and Frank H. Gregg, Purchasing Department Buyer.

Planning precise schedules for aircraft overhaul are (left) W. B. Moore, General Manager of Maintenance, and G. W. Cook, Assistant V.P. of Engineering and Maintenance.



Bill Prentiss, UAL Champion Representative, gives technical instruction in Line Service Engineering Department and A. C. "Doc" Wheeler, General Manager of Spark Plug Maintenance.





A Mustang is craned toward right down to the test unit, bolt and spark plug.

of the trouble and avoided some rather destructive engine failures."

General Manager of Maintenance — Mr. Howe — emphasized UAL's success with Champions: "United's experience has demonstrated Champion's consistently dependable performance under widely varying operating conditions," said the 30-year UAL veteran.

Those who occupy the "front office" of the airline itself, those who bear the final responsibility for safe operation of the aircraft in flight, have the deep personal appreciation of the meaning of reliability. United's Flight Test Section employs 15 experienced pilots and flight engineers. During a hanger flying session with these test pilots, I learned just what Champion does mean to them.

Robert M. Melver, a Test Captain for UAL, said: "My experience dates back to 1933 and includes some 10,000 hours' flying time, 6,000 of a flight testing. We have a tremendous sense of moral responsibility toward each airplane we release, since we realize each plane will cover millions of miles under all kinds of conditions before returning again for overhaul.

"Part of our job, then, is to devise exhaustive tests for each component of the plane, and those tests we carry out religiously. The work of Flight Test is the end point in a long line of maintenance and inspection procedures, and we have authority to fly each plane as much as necessary to insure the ultimate in quality. Each component must fall within very tight tolerances be-

fore it can be considered satisfactory. Naturally, the engines and their accessories are subject to very close scrutiny, and spark plugs fall into this category.

"During such flight test, our flight engineers, electric spark plug performance constantly by means of ignition analysis, and pilots do likewise indirectly by means of the instruments on the power panels.

"Using this instrumentation, we've been watching the performance of Champion Spark Plugs for 10 years. The excellent characteristics of these spark plugs have become not just a matter of personal opinion, but a matter of statistical record. We can substantiate that record by saying that Champion meet the tough requirements of an airline operation and are backed by a good service organization . . .

Captain Melver's comments were seconded by UAL Test Pilot Robert C. Collins, who told me that so far as he knew no other spark plug has ever come close to replacing Champions at United.

UAL's record of safety and dependability reflects the reliability of the heart of its aircraft ignition system, the Champion Spark Plug.

Red Carpet means the floor. To millions of passengers it means the ultimate in reliability, service and comfort. In a Mustang, it means the ultimate in engineering and maintenance.

When you fly United, you're flying Red Carpet.

by HERBERT G. FISHER

effective "Q" (capacitance for a system that emits power) which can be coupled to the waveguide system and substitution can then be carried out as for any other notch attached to a waveguide.

Finally, for the straight waveguide, the material is described in terms of a negative attenuation constant and the net power change can then be calculated in a straight-forward manner.

For the straight guide device, for example, the net gain is given by the following equation:

$$G_{net} = (1 - \alpha) \cdot \frac{1}{2} \cdot \frac{1}{\beta}$$

Where:

- α = Waveguide attenuation constant
- β = Waveguide length
- γ = Dissipated emission gas coefficient

For a paramagnetic sample the stress

related emission gas (a) is established from the following:

$$a = \frac{2.8 \times 10^{-10} \cdot N_0 \cdot f}{\beta \cdot T}$$

Where:

- N_0 = Total density of paramagnetic atoms
- f = Operating frequency
- β = Guide width
- T = Absolute temperature of crystal

Representative values for a device



TYPICAL DeW Line Early Warning (DEW) Line station showing complex of methodology, access roads and intercommunication facilities surrounding the three shaped housing of the main search antenna. DEW line officially became operational this summer.

DEW Line Goes Into Operation



DIRECT and receiver stake rods that show subsidence of ground built and coastal surface early warning system as described in a talk by Brig. Gen. C. B. Gibson.



PARABOLIC microwave antenna 10 ft high set for DEW Line microwave receiver system. Due to severe winds, antenna must be heavily secured against prevailing high arctic winds. Site was installed by USAF by Western Electric Co. will be operated by International Telephone and Telegraph.



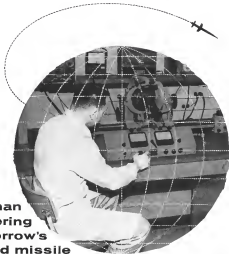
Cabin view of Cessna 440B twin base, which electronically duplicates flying characteristics of DC-8B Mustang, shows United crew selection flight conditions "simulated" by instructors. UAL has series of these — largest number ever for an airline — for advanced pilot training.



Robert McCoy collects with Republic company instructor.



United with United in two-speed briefing room with a 1,000-mile view. Specialists from company instructors in Denver every morning on operations in-line operations for post-24 hours — gives a forecast of the unit 24. This contributes to accuracy improved service.



This man is steering tomorrow's guided missile

His uniform is a laboratory coat, his cockpit a dust-free room with carefully controlled temperature and humidity. He's a skilled General Mills gyro technician—as much a part of the defense of his country as the jet pilot. Results from his work, and from work in other R & D labs, assure us that tomorrow's guided missiles will be even more accurate than today's. ♣ At the Mechanical Division of General Mills, the gyro lab is part of a

talented, well-equipped guidance and navigation systems development group. The group also includes top men and facilities in infrared, microwave, electronic and mechanical design, and overall systems engineering. We train these men and facilities with a complete precision production plant to handle systems, sub-systems and major assembly problems for the world's most exacting customers. Possibly you too can benefit from our capabilities.

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operating at a frequency of 10,000 mc., with a bandwidth of 10 mc., using a paramagnetic crystal coating (10^{-10} cm² cm⁻² (N₂), operating at an absolute temperature of 4K (-269°C), works out to give a modulated oscillator gain coefficient (a) equal to 0.05/cm. With negligible waveguide losses (α_{wg} = 0), this would provide an overall gain of 10 in a waveguide one meter long.

The properties of several different types of Masers, including the new ferromagnetic amplifiers, and the operating techniques peculiar to each, will be described in the second article of this series.

OPTICAL FILTER CENTER OCCURS

- **Usability in Reverse-MATS** Fife, Dewart is studying incorporation of the failure of Lorenz as considered cause for guidance errors on long overwater routes. Old-fashioned celestial navigation is thought to be too reliable, superior in proving that it should be regarded only as a backup in electronic navigation aids. Tests on operational flights checking celestial navigation against Lorenz found:
 - Celestial navigation is less accurate than is desirable.
 - High cloud cover interferes with co-

operation right now frequently.

- **Complete Is** is possible only at night with celestial navigation. During daylight with a single line of position from the sun can be obtained.
- **Is autostep or automatic** in which it is slow of lock, a celestial fix is impossible because manually piloted aircraft does not provide needed stable platform.
- **New Doppler System**—Doppler navigators, specifically designed to meet near-terminal landing requirements, will be developed by Bendix Aviation, according to announcement at a recent meeting of the AFEG Self-Contained Navigation Aids Subcommittee in Los Angeles. Prototype deliveries will coincide with delivery programs for the Douglas DC-8 and Boeing 707 jets. System will be frequency modulated and continuous wave operating on X-band and will feature guidance of all electronic elements in plug-in ATR units.
- **Transmitter Market Growth**—Possible billion dollar market for transmitter and other semiconductor within ten years has been predicted by a General Electric executive. Sales of these devices have consistently outstripped even the most optimistic market projections according to J. H. Swanson, manager of marketing for GE's Semiconductor

Products Dept. at Schenectady, which this year sales volume for the industry is expected to exceed \$140,000,000, with 40% or over the production figure and 82% over the industry sales for 1956.

• **Vanguard computer**—Autocue Recording and Reduction Facility (ARR) to process flight information from each of the three stages of the test and satellite knowledge checks will be installed by early fall at Air Materiel Test Center near Cocoa Beach, Fla. Facility, which operates on the same principles as a high speed digital computer to produce vehicle performance data within 72 hours, is a joint development of the U.S. Naval Research Laboratory and Lockheed, Inc.

• **"Pence Remote Cals"**—Autocue tape recorder, which must select plan to install in jet engine to provide high-governed engine, may also be used to give pre-recorded warning announcements in event of unsatisfactory development at high altitude. Warning message would be delivered automatically when decompression altitude pressure switch

• **No Change of Heat-Flux**—Chrysler at Kenilworth Company's Asheville, N. C., plant have rejected union for 1956 base year 1953, with vote of 225 to 77 against representation by



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Write or wire Dept. 220A for specifications and complete information.



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INC.** *Dynamic Mechanical Design*
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San Anselmo, CA, California

INC. 455 W Washington Blvd,
San Angeles 12, California

Color: printed, manufactured by Amco (see Electronic prints) AF 301 (G) Maxima Series, Germany; Golden
Ruler: Japan; Adhesive and Paper: Gerd in, Elmer's (Gerd in)
Notes: Nuclear Instruments and High-Speed, Los Angeles, California

International Brotherhood of Electrical Workers

► **LFE Builds Test Radar**—Navy has accepted first of 10 AN/GPN-6 airport surveillance radar systems being built by Lantech Inc. For Electronics. Reimburse are scheduled for delivery within six months.

► **Staved Wins—Staved Engineering, Inc., Plattsfield, N. J.,** has beaten out 11 competitors to receive Navy contract to develop airborne radar for use on the AN/ASB-4 boom director system for attack bomber aircraft.

NEW AVIONIC PRODUCTS

Components & Devices

* Subminiature fixed glass equipment, Type WL, features radial high temperature soldered leads for direct connection to printed circuit boards. WL-4 measures 0.5 by 0.3 in. and is available in values to 1,000 ohms. WL-5 is 0.3



by 0.5 and is available to 2,200 mesh. Both types are less than 0.1 in. thick. WL's are rated at 300 volts to 800°C. Corning Glass Works, Corning, N. Y.

• Since 16 synchros capable of operating for 100 hours at 450° and for 1,800 hours at 170° have been developed for direct takeoff magnetic mounting. Units are available in transmitters, transducers, differentials, and resolvers. With rotor as primary, transmitter characteristics are 26 volt, 0.145 amp input, 11.8 volt output, 206 ms./deg. zero drift, and 3 deg. phase shift. Developed by Chittara Products Co., Inc., 9014 West Chester Pike, Upper Merion, Penn.

*Electronic counterstat for airborn telemetry applications. Series ETC-18.

is obtained by the manufacturer to be the best commercially available unit of its kind. Converter is offered with an operating rate from 75 to 900 power per second, and accepts all IRIC in quantities for PAM and PDM controlled techniques and magnetic tape systems under MIL E 5773A envelope. Errors due to drift, constant, and nonlinearity are less than 0.005%. Power required is 150 watts DC at 120 vac.

Size is 3 in. in diameter by 5 in. long. Weight is less than 2 lb. One developed by Arneux Corp., 11924 West Washington Blvd., Los Angeles 65, Calif.

- Heat sinks, Models TBS10 and TBS20, permit higher levels of dissipation for transistors and provide a mounting clamp for transistors suspended by their leads.



CT addressed case and HS-10 for the GT version of the JETEC 10 case. Both models are made of beryllium copper. Manufacturer is General Transistor Corp., 91-27 115th Place, Jamaica 35, N.Y.

• TransistORIZED electrohydraulic valve amplifier has 50 db power gain and will operate at temperatures to 175F. Capable of operating with 50 cycle to 400 cycle excitation, the unit includes variable gain, balance, and quiescent controls in a package measuring 1 in. by 2 in. by 1 in. Manufacturer is Control Systems Division of Kelsey Hayes Co., Detroit, Mich.

• Subminiature potentiometer called Mini E-Max weighs 10 grams and is capable of operating to 125 C. Unit weighs 0.5 in in diameter by 0.5



is long and can be solder-mounted. Unit is produced by General Scientific Division of San Fernando Electric Mfg. Co., San Fernando, Calif.

there's a
heat barrier
on the
ground, too!



With so much risk about the various in-flight "barriers," it should be mentioned, there is a "best barrier" problem for jet aircraft on the ground, too. Ground checking of instrumentation and other parameters requires efficient and dependable cooling. That's why manufacturers like Douglas Aircraft Co. use American Electronic's MA-5 ground support air conditioners as the PDA-1 SKYRAY and other aircraft.

The cart shown here has a "design load" capacity of 100 lb./hr. of 43F air against 0 to 3 positive static pressure with controllable output temperatures ranging between -43F and 57F. The unit is designed to provide a wide range of operating conditions including ambient temperatures ranging from -43F to 137F. America's MFA's quality design, engineering, and build is also available with electric motor driven units. One of a series of designs and capacities available. Contractors and instrumentations are such that satisfactory operation and maintenance may be obtained with minimum electrical personnel.



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Other products manufactured by the Thomas Machinery & Equipment Division of American Electric include Motor Alternators, Switch Power Supplies, and Magnetic Amplifiers.

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First in Fluid Connections

Weatherhead is pleased to announce the appointment of Pacific Automotive Corporation, Burbank, California, and Airwork Corporation, Millville, New Jersey, as distributors for its aviation products... hydraulic and fluid system components.

With these new appointments, Weatherhead products are now available to aircraft owners and operators across the nation through factory-trained personnel of the two companies. And in the Southwest, Weatherhead products will continue to be available through Associated Aircraft Supply Co., Dallas, Texas.

These distributors provide prompt, reliable delivery at lowest possible cost, to airlines and suppliers. Through their networks of nationwide local dealers, who also maintain adequate stocks of Weatherhead products, these distributors offer fast service to aircraft owners and airport service operators.



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Distribution and sales in a 17-state area on the eastern seaboard, through its spare parts division... branches at Newark, N. J.; Washington, D. C.; and Miami, Florida.

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NEW AVIATION PRODUCTS



pressure on absence of fuel flow is piped down to one inch diameter. Control units use a single electric transfer and are powered by the probe when it is inserted in fuel.
Academia Associates, Glenwood Landing, N. Y.

Fuel Transfer Pump

Model RC1140 reversible fuel transfer unit, weighing 1 1/2 lb., consists of a positive displacement rotary vane pump, an explosion proof motor and relief valve. Its flow rating is 60 gph with 5 psi gauge discharge pressure. Rating is based on pumping MIL-9-5572 fuel at 600-1000, inlet at 50



Missile Coupling

Flexible coupling designed for missile and aircraft applications meets MIL-C-25014 and allows a total of one quarter-inch axial motion and a 30 degree arc. Standard AND10693 and AND10690 bonded tube ends are used for fitting attachment. Standard O-rings are used for seals, standard gland dimensions allow space for seal swell and constant motion of tubes. Part number is 64501.

On Mark Couplings, On Mark Engineering Co., 260 N. Ave. 64, Los Angeles 41, Calif.



Device Controls Metal Fatigue

Designed to lessen problem of metal fatigue in vital aircraft and aircraft components, metal tension monitor will electronically control and test material for uniform strength while it is being formed on a radial draw former. Unit will virtually eliminate overstretching or buckling in materials such as aluminum, stainless steel, magnesium, titanium and copper-nickel alloys.

Unit was furnished recently to Army Ballistic Missile Agency, Huntsville, Ala., and will be used by Comets for work on the B50 jet transport.

Unit is made up of standard stock, no special tooling by local electronics representatives and is adaptable to full automatic operation.

Cyral Bath Co., Solon, Ohio.

gph with same gauge discharge pressure at 15,000 ft.

Continuous duty motor operates at 27 v. d.c., external wiring to the electrical connector provides for reversing the direction of rotation through control switch connection.

Learn-Rotor Division, Lear, Inc., Elkhart, Ohio.



Modular Autors

Line of modular component aircraft engine sections can be assembled from 68 standard interchangeable parts to meet a wide range of requirements. New line is slated to meet MIL-A-5004A.

Components consist of motor, gear reducer, output ratchet (dividing line or capstan), accessory drive or transducer and screw jack. Three broad application classifications comprise the series: grapples: 1.12, up to 510 lb., 1.16, up to 1,500 lb. and 1.30, up to 1,700 lb. Airborne Accessories Corp., Hillside 5, N. J.



Tough Magazine

Cordmaster record magazine for recording cardiographs is designed to take 100G shock and 1,000F forces, in addition it is explosion proof in accordance with MIL-E-5400A (ASG).

Magazine is eight inches wide x 12 1/2 in. high x 17 in. long, with full load of paper it weighs 80 lb. Consolidated Electronics Corp., 100 N. Stone Mountain Villa, Pasadena, Calif.



Giant Template Camera

Camera being built to Canon specifications weighs over 10,000 lb., measures 35 ft. long and 5 1/2 ft. high, enabling reproduction of templates up to 5 ft. x 12 ft. within 1/2,000 in. or

Keeping a fatal rendezvous in 4-D

Any hunter who's fired at a fast-flying duck knows you have to figure time—the 4th dimension—if you want the shot to meet the duck in flight...

Now substitute an airplane taking full evasive action at 40,000 feet or more, and conventional ways of aiming become obsolete. Yet this problem is relatively simple, as was dramatically proved at the first tests of the Nike-Hercules, jointly developed by Bell Telephone Laboratories and Douglas.

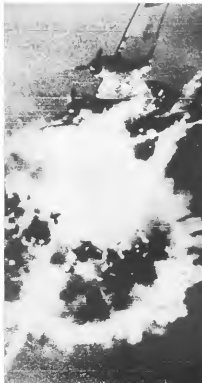
Even more complex than ground-to-air marksmanship is air-to-air gunnery, where opposing planes top 5000 mph, and fire bursts that move twice that fast. It's an entirely new science, based on principles first expounded by a Douglas scientist, and proved in repeated successful tests—even against unseen targets.

Depend on
DOUGLAS

first in
Aviation



Dramatic moment in aviation history, a Douglas-built Nike-Ajax bursts in on a dense burst.



A split second later from the launcher in flames, as Nike-Ajax scores a direct hit.



Here is Nike-Hercules, soon to take over defense of U. S. cities from the earlier Nike-Ajax. Trapping the range of its performance, Nike-Hercules can be armed with a nuclear warhead, to knock out entire Soviet with a single blast.



Douglas engineers load the rocket pod of an Ajax Sigma. The control problems at supersonic speed, severely more greater than encountered at subsonic speeds, are solved on principles developed by a Douglas engineer. Constrict lift can be moved by planes approaching air another at close to a mile per second, and even when unseen.

LOW-TEMPERATURE PROCESSING

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Equipment size 42 in. dia or plate can be housed manually or electrically.
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WHAT'S NEW

Reports Available:

The following reports were sponsored by Office of Technical Services, United States Department of Commerce:
Studies in Respiratory Physiology: Third Series: Chemistry, Mechanics and Calculation of the Lung—by H. Sachs, The University of Rochester School of Medicine and Dentistry for Wright Air Development Center—\$1.25, 96pp. (P.B. 121883)

A Study of Permanent Magnets of the Reman Fourth Type—by K. J. Setaas, The Indiana Steel Products Co. for the Wright Air Development Center \$1.50, 39pp. (P.B. 121893)

The Effects of Interdiffusion Contaminants on the Neutron-Torque Properties of

Titanium and Titanium Alloys Part 2: Alloy Titanium—by E. P. Klay and N. I. Ploch, Syracuse University for the Wright Air Development Center. \$5.50, 162pp. (P.B. 121775).

Surface Hardening of Titanium with Metallized Elements—by Arvid R. Research Foundation for Walworth Arsenal, U. S. Army. \$4.00, 151pp. (P.B. 121821)

Heat of Formation and Entropy of Titanium Tetrafluoride—by W. F. Kline and others, California Institute of Technology for Ordnance Corps, U. S. Army. \$3.50, 17 pp. (P.B. 121750)

Transistor Feedback Amplifier Design—by G. L. Roseberry, University of California for the Office of Naval Research \$7.50, 25pp. (P.B. 121556)

Lowvoltage A Low-voltage Triggered Gap Switch—by E. E. Callaghan, W. C. Chase, and B. L. Morgan, Air Force Cambridge Research Center \$5.00, 39pp. (P.B. 121063)

Effects of Moisture Sorption on Weight and Dimensional Stability of Alkyl Inorganic Foam Cores—by V. C. Schramm and E. W. Kucera, Forest



Model Shows Plant Layout

Three dimensional model depicts new jet engine fuel control plant which will be placed in full production during October by Hamilton Standard Division of United Aircraft Corporation. The facility located at Windsor Locks, Conn. adjacent to the main plant will employ between 2,500 and 3,000 persons and have 400,000 sq. ft. of floor space. Structure will house a self-contained unit which includes fuel control engineering, experimental shops and production installation of machinery and equipment now in being this month during the annual Hamilton Standard vacation shut-down. Model will be used by plant engineering to facilitate changes between of machinery and general plant layout.

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Shimnuts are of NAS 618 alloy. Permits use of single length screws in tapered joints. Available for bolt sizes from 10-32 to 1/2-13, spacer heights from 1/16 to 1/16. Nuts are cadmium plated steel with nylon insert (NENA 887) or all metal (SPS-71100). Spacer is 7075-T6 Aluminum Alloy. Bolter is Anodized in various colors to indicate height of spacer.

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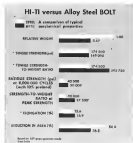
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SPS Hi-Ti titanium aircraft bolts are regularly manufactured in four standard configurations: internal wrenching, external bolts, 12-point external wrenching, tension bolt (through head external) wrenching, shear bolts, and flush head internal wrenching shear bolts. Wrenching 40% less than steel bolts of equivalent dimensions, they can save you as much as 8000 lb per aircraft—while not loss in strength.

SPS Hi-Ti titanium bolts help you build lighter airframes without sacrificing strength



Hi-Ti vs. alloy steel. Tensile, elongation and reduction area properties are based on the performance of bolt type specimens. Endurance limit was determined by subjecting bolts to minimum load stressing between maximum and 10% overload for a total of 1,000,000 cycles without inducing failure. Significant improvement in the strength-to-weight ratio at endurance limit.

SPS Hi-Ti titanium bolts help you solve the problem of reducing airframe weight without compromising strength. They weigh 40% less than alloy steel bolts of the same size. One lb. of them can do the work of 1 1/2 lb. of steel bolts. Yet in tensile-strength-to-weight ratio and fatigue resistance, they outperform steel.

Once considered a laboratory curiosity, titanium bolts have for some time been standard production items at SPS. Hi-Ti bolts are found in many advanced design operational aircraft. This is because SPS—producer of the first successful titanium aircraft bolt—invested over \$500,000 and several years of high priority research in learning how to deal with this promising but sensitive metal. Today SPS has the most extensive facilities in the industry for the production and testing of titanium fasteners. As a result, we can give you both the technical assistance and the delivery you need to unlock fully the advantages of titanium bolts in your current aircraft projects.

For more information on Hi-Ti titanium aircraft bolts, write Aircraft Products Division, STANDARD PRESS STEEL CO., Jenkintown 3, Pa.

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C-130 Cockpit Stresses Visibility

Interior of Lockheed C-130 Hercules cockpit emphasizes stress on good visibility. Glass area amounts to 40 sq. ft. in the crew area. Besides upper panel, window for downward view can be seen over area of Lloyd Harris (left), chief pilot, Georgia Division.

Products Laboratory, Forest Service, U. S. Department of Agriculture, for the Wright Air Development Center 51-05, 49pp. (P.B. 121880)

The Effects of Guided Impedance to Turbulent Air Flow on the Pattern of Breathing and Absorption Can Composition of Man—F. W. Zickman and F. D. Hall, Dair, Virology School of Medicine for Wright Air Development Center 51-25, 49pp. (P.B. 121507)

Evaluation of Porous Materials for Boundary-Layer Control—by D. E. Debus, Battle Mountain Institute for Wright Air Development Center 51-05, 169pp. (P.B. 121851)

The Relationship of Hydrogen Measurement to the Tensile and Compressive Flow Curves—by R. E. Leshart, General Electric Research Laboratory for Wright Air Development Center 51-30, 49pp. (P.B. 121140)

The Silver Oxide-Zinc Alkaline Primary Cell Part 4: Alkaline Characteristics of Zinc Alloys—by C. M. Shepherd, Naval Research Laboratory 51-75, 25pp. (P.B. 121744)

Automatic Flame-Out for Landing—by D. Matheson, B. McLane and G. Foxworth, Minneapolis-Honeywell Regulator Company for Wright Air Development Center 51-35, 169pp. (P.B. 121633)

Flight Test of an Autopilot Installation on a Lateral Gust Allevator on a PT-35

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Velocity Center 51.08, 16pp (P.B. 121715)

Performance Standardization for a Turboprop Engine Equipped with a Variable Area Nozzle Controlled by Engine Speed—by R. L. O'Neal, Air Force Flight Test Center 510, 17pp (P.B. 121668)

Permanent-Magnet Generation Part I: Theory—by D. J. Homburg and D. S. Toffolo, Naval Research Laboratory 510, 19pp (P.B. 121662)

Hinman, A New Permanent Magnet—by E. Adams, W. M. Haskard, and A. M. Sykes, Naval Ordnance Laboratory 510, 19pp (P.B. 121716)

Permeability of Burner Materials to Volatile Carbonic Inhibitors at Various Humidities—by A. S. Mohaupt and J. P. Hild, Forest Products Laboratory, U. S. Department of Agriculture for Wright Air Development Center 5135, 9pp (P.B. 121593)

Publications Received:

A Case Study in Corporate Acquisition—A Presentation by R.C. Anderson, Inc. 5175 (51 30 424), 44pp

A proposed comprehensive, step-by-step presentation of acquisition, control, integration, and integration of different plants, in brief that full details can be most helpful.

Post-War International Civil Aviation Policy and the Law of Air—by H. A. Wesselsch—Pub. Martinus Nijhoff N. V., 5 Lange Voorhout, Netherlands 11-50, 600pp (approximately \$5.00) 183 pp

A review of the many problems which arise when attempting to reconcile the individual requirements of the world's differing nations. A valuable sourcebook for the serious student of air law.

Electrical Construction Materials List—Pub. Underwriters Laboratories, Inc., 161 Sixth Ave., New York 13, New York, 453 pp

A list composed of all the materials needed for electrical construction work.

Electrical Appliances and Utilization Equipment List—Pub. Underwriters Laboratories, Inc., 161 Sixth Ave., New York 13, New York, 360 pp

A list composed of all the appliances needed for electrical appliance work.

Household Location Equipment List—Pub. Underwriters Inc., 161 Sixth Ave., New York 13, New York, 105 pp

A list composed of electrical equipment for household locations.

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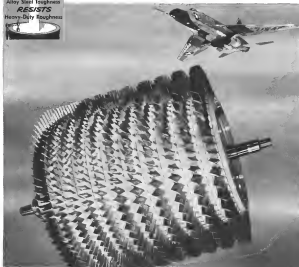
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For in these fine steels are found the highest strength values—plus an exceptionally high strength-to-weight ratio that permits the design of thinner, lighter sections to save weight and hold down size without sacrifice of strength or safety.

These essential qualities of AMS 6415 were the basis for its selection for use in compressor rotor discs in Pratt & Whitney Aircraft's J-57 jet engine. The discs are machined from forgings by the Jet Division of Thompson Products, Inc. Forgings are supplied by Wyan-Gordon Company.

Still another reason for selection of this fine steel is its exceptionally good deep-hardening characteristics. Undersize sections to less than .001 in. diameter—plus the production of hard, wear-resistant surfaces around tough cores.

Production, processing and development of aircraft alloy steels requires extremely close cooperation among the metallurgists of the part manufacturers, forge plants, and steel producers. The collaboration of these combined efforts provides the designer with a material for operating and structural parts that is free from excess weight, yet tough and strong to withstand shock, impact, stress and fatigue.

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USAF Contracts

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Helicopter Test Fires Rockets

Facility of many rockets and machine guns fired from helicopter at ground targets is used by Army H21 at Ft. Rucker, Ala. (Aircraft Below) includes eight 80 mm. Cannon rockets, two 16 mm. and two 32 oz. machine guns. Various public have been prepared for use on various types of helicopter by General Electric (AM May 20, p. 32).

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New Westinghouse J34 engine powers first Navy all-purpose jet trainer

North American Aviation has chosen the new version of a proven power plant for this first Navy all-purpose jet trainer. Designed as a complete pilot training system for operation from both land and carrier bases, the T2F-1 is capable of taking students from primary through advanced phases to carrier qualification.

Engineered and designed after Pearl Harbor, the J34 turbojet saw rugged service in Korea. Its solid reputation for combat reliability and durability has now become almost legendary after more than two million hours of combat and operational flight time. It is less than half as vulnerable to foreign object damage as other engines. No wonder the Westinghouse J34 was selected to power this new all-purpose jet trainer!

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Compatibility tests of the Wernigerode J84-WF-46 engine and the North American T731 turbine inlet duct were carried out using the rig at the Naval Air Station, Orlando, Fla. The engine is shown mounted in an outdoor thermal coffin. Completion of the first T73 was expected late this year. Engines will be manufactured at Wernigerode's plant in Kernen, Germany.

Doc ID: 23499 Date published: 2011-01-11
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Army Contracts

Following is a list of modified contracts for \$25,000 and over as released by Army contracting office:

ARMY AIRCRAFT VEHICLE DIVISION, INC.
 1001 N. 10th St. 1001 N. 10th St.



Atlas Plant Takes Shape

New \$40 million plant in San Diego for production of Orion Atlas antiaircraft missile under development. The new structure is integrated with engineering building (left) and administration. Figures at upper left. (Wendell Wright photo)

108

AVIATION WEEK, August 18, 1957



Exciting new things are happening at HOFFMAN ELECTRONICS

Out of advanced electronics research and development at Hoffman comes an *exciting new* concept in ground radar: *skill* under wraps by the Military, this ground and ground radar development applies revolutionary techniques to solving complex problems.

Hoffman's extensive experience in designing, developing and producing radar for airborne and shipboard installations laid the groundwork for this exciting new electronics development. In radar, as in other vital military and commercial fields—communications, navigation, weapons and communications systems—Hoffman continues to produce the pattern for progress in electronics.

In a dynamic industry where only change is permanent, exciting things continue to happen at Hoffman.

PHYSICISTS • ELECTRONICS AND MECHANICAL ENGINEERS

Significant new developments at Hoffman in the fields of radar, VLF, HF, VHF, UHF, radar, ECM, weapons systems and communications have earned important positions for scientists and engineers of high caliber. Please address inquiries to Chief Engineer.

Hoffman Electronics

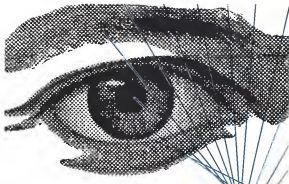
HOFFMAN LABORATORIES, INC.
 2761 South Hill Street
 Los Angeles 4, California

AVIATION WEEK, August 18, 1957



HOFFMAN LABORATORIES, INC.
 2761 South Hill Street
 Los Angeles 4, California

RESEARCH
 DEVELOPMENT & PRODUCTION
 WEAPONS SYSTEMS
 COMMUNICATIONS
 NAVIGATION
 ELECTRONIC COUNTER MEASURES
 RADAR
 SURVEILLANCE APPLICATIONS



endowed with sight ...and insight

Inquisitive eyes denote the inquisitive minds that made possible today's rocket powerplants. Sharp eyes and minds that have the power to visualize challenging problems . . . and the insight to solve them.

Spurred by such long-range sight and insight, RMI has blazed the trail in rocket power for over fifteen years. And today, with new developments in manned and guided flight creating vast new propulsion problems, RMI will continue to lead the way. For RMI engineers and scientists form a talented, far-sighted team, discerning and producing advanced powerplants for the vehicles of tomorrow.

Engineers, Scientists—Perhaps you, too, can work with America's first rocket family. You'll find the problems challenging, the rewards great.

Power for Progress



REACTION MOTORS, INC.

DERVILLE, NEW JERSEY

4071

Certificates of Necessity

Washington—Office of Defense Mobilization has awarded The Martin Co., Orlando, Fla., three certificates of necessity totaling \$4,039,215 covering the manufacture of guided missiles. All three certificates were allowed 65% of the rapid rate. Other certificates awarded:

Strom Laboratories, Beverly Mass., aircraft equipment for military use, \$675,018 with 10% allowed.

American Tooling & Manufacturing Co., Warren, Ohio, aircraft tooling and equipment, \$400,000 with 10% allowed.

East Rock, Inc., Cambridge, Mass., \$700,000 for engine parts, \$100,000 with 10% allowed.

Continental Aircraft Corp., Cincinnati, Ohio, \$1,000,000 for engine parts, \$100,000 with 10% allowed.

CAA Contracts

Washington—Following is a list of contracts as released by the Civil Aeronautics Administration:

U.S. Civil Aeronautics Co., Philadelphia, 127,000 for development of a derivative engine, \$100,000 for 10% and production of 10 (10/10/10).

West Electric Co., Kansas City, Mo., 114,000 for manufacturing equipment, \$100,000 for 10% and production of 10 (10/10/10).

Boeing Co., Seattle, Wash., 114,000 for 10% and production of 10 (10/10/10).

Boeing Co., Seattle, Wash., 114,000 for 10% and production of 10 (10/10/10).

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SECRET POWER PROGRESS REPORT

Igniting Liquid Propellant Rocket Engines

by Leonard Danabas

A research physicist in the Physics Department, Component Development Division, Reaction Motors, Inc., Mr. Danabas has recently been investigating ignition phenomena as applied to rocket engines. A graduate of Rutgers University, where he received a B.S. in Physics, he has been with RMI since 1951.



Rocket engines operate with a wide variety of liquid propellants having various chemical and physical properties. These properties, in conjunction with the application of the igniter, dictate the type of igniter system that will be most effective. Although the exact mechanism involved in the ignition process is not completely understood, it is generally believed that, under specific conditions, there exists a minimum threshold energy requirement below which the combustion wave does not propagate. Rocket engine igniters are generally designed to provide the energy on a "margin that is enough" basis.

Many liquid propellants require an external source of energy to achieve ignition. Hydrogen peroxide, for example, may be catalytically decomposed in the presence of silver or platinum catalyst, and many hydrocarbon oxidizers react spontaneously merely on contact with such catalyst.

Non-spontaneous propellant combinations must be ignited by some form of energy introduced from an external source. In early rocket experiments, ignition was achieved with burning kerosene-soaked wick held at the end of a long pipe. After several engine starts, the trend shifted to the use of electrically fired pyrotechnic charge—a technique successfully used in the V-2 and Viking engines and still employed in many single-shot rocket applications. Later ignition techniques employed ultrasonic, catalytic agents, pyrotechnic fuels (which react spontaneously in the presence of an oxidizer), electrothermal devices (operating with arcs, sparks, and heated filaments), and torch or flame-type igniters.

The advent of jetted rocket research aircraft required lighter, more compact igniter systems capable of operating at high altitudes and possessing unusual physical problems. The experimental X-1, X-1A, and the Douglas Skyrocket, powered with RMI engines, had four thrust chambers which required starting during flight. To insure smooth and positive operation, low-type igniters, which operate with sparks and alcohol, were developed. Small quantities of these propellants were injected into a pre-combustion chamber, ignited, and the resulting flame projected into the rocket motor. This unique device provided high ignition energy, operated at all altitudes, and assured smooth, reliable starts.

Preventing the nonignition of liquid propellants in the combustion zone is the most important factor in rocket motor starting, regardless of the ignition technique employed. An insignificant ignition delay—in the order of milliseconds—or a sudden propellant surge during the ignition period can result in a buildup of pressures that may explode with destructive violence.

A successful rocket motor start, therefore, is a function of numerous variables. Propellant flow rates must be accurately controlled and the propellants intimately mixed. The mixture ratio must be suitable, and the igniter action must be adequate. These factors are continuously being investigated at RMI for the purpose of advancing rocket science.

Solving some of tomorrow's specific rocket problems—by conducting experiments pertaining to the development or investigation of theories, principles or techniques concerning operation of rocket engines and related equipment—is the general work of the team of specialists comprising RMI's Component Development Division. One door is open to qualified men interested in this challenging work.

If you desire one or more reprints of Mr. Danabas' article, or would like to receive further information about employment at RMI, write to Information Service, Reaction Motors, Inc., Ford Road, Derwent, N.J.

Power for Progress



The new CURTISS-WRIGHT ZEPHYR JET ENGINE

So quiet...

You won't know it's a jet!

An transportation must become quiet not only for its passengers, but for the communities in which it operates. The development of the Curtiss-Wright Zephyr Turbogun was based on the premise: 1—that its technical excellence must provide a worthwhile performance experience; 2—that its advantages must be noted in the final analysis from the economic point of view; and, 3—that it must have a high order of public acceptance, not only from the viewpoint of the passenger, but the communities in which the airport is located and over which the airplane is operated. Moving airports further out of town with its streamlined expense and the inconvenience of time lost in additional trip to airport transportation is not the answer.

Curtiss-Wright Corporation has completed sufficient testing with the "Zephyr" to warrant the statement that the jet engine transport can be quiet and have a high order of public acceptance.

New Testing Procedure

The testing procedure for the Zephyr is new and novel. Curtiss-Wright designed the standard jet engine test cell—a mass of concrete, steel, sound absorbers, water piping, etc.—and transferred the testing of the Zephyr to an open air test stand in which the engine is run with no soundproofing or sound absorber equipment under a specification calling for a maximum of noise, as well as a minimum of fuel consumption, just as the engine would run when it is in an airplane at the airport.

This outdoor testing equipment consists of two mobile vehicles, one on which the engine is mounted for tests, the other housing the control room and instruments required to measure thrust, fuel consumption and other equipment. Through the use of these mobile vehicles, the Zephyr is immediately being tested and demonstrated at airports and at off-highway locations to record and analyze its actual level of noise, taking and fuel throttle conditions. This practical method of testing enables engineers and technicians to change new noise suppressing devices being developed and other equipment quickly and with ease.

The engine may be held at a given set of conditions so that technicians can observe via lanterns, record and analyze noise levels for extended periods of time, even, while they are doing this, engine operating conditions can be changed instantly for purposes of demonstration. This method is preferable to listening to an airplane passing overhead

where the noise is reduced or muffled, where the time for measuring and recording the engine of noise is limited and where it is difficult to fix the distances and conditions with any degree of accuracy.

Curtiss-Wright plans to demonstrate the mobile engine plant and the Zephyr engine at airports and other locations in the U.S. A request has been received to demonstrate the unit at a major airport in Europe.

Quieter Than Piston Engines

A jet engine transport unit is quieter than a piston-engine airplane. For example, Curtiss-Wright has been able to measure the compressor whine of the jet engine. This is the loudest of the two principal sources of noise from a jet engine, but, in a practical sense, its frequency and characteristics are something like a buzz and more objectionable to some people than the jet noise which is of a lower frequency. Curtiss-Wright has also reduced the jet exhaust noise to acceptable levels because basically the engine is designed so that the mass of air passing through the jet to produce thrust is moving at a lower speed than that caused by engines developed for military use at supersonic speeds. For the foreseeable future airlines will be substantially, and, therefore, there is no need for wing tip extensions, special jet engines, with their disadvantages, for jet engine operation.

Sound Research Program

Engine noise suppression, while the major element, is only one phase of the Curtiss-Wright program to lower noise levels for aircraft and airport communities. The mobile control room used in the Zephyr tests is loaded with C-40s, a new sound absorbing, multi-cellular, laminated material designed by Curtiss-Wright. This enables engineers to conduct concurrent tests of new methods of soundproofing aircraft passenger cabins. The same C-40s can play an important role in absorbing ground noise at airports where jet and other aircraft are operated. For example, C-40s can be used along between runways and buildings will substantially reduce noise at the building. And C-40s, in the form of wall film, ceiling and floor coverings, can further absorb sound and result in quieter air terminals and offices than ever before.

There are but a few of the technological developments of the Curtiss-Wright Corporation designed to make the introduction of jet airplanes to the world airports quieter, more comfortable and economically sound. Others will follow.

This advertisement is published in response to the public interest in the Zephyr Jet engine reflected by numerous inquiries received from individuals and governmental authorities at every level.

CURTISS-WRIGHT



CORPORATION

WOOD-RIDGE • NEW JERSEY



Engineering and production of precision electronic equipment and systems for the UNITED STATES AIR FORCE is the daily, day-out concern of many hundreds of people and many specialized departments in the Reeves organization. We take personal pride, therefore, in solving this great Service upon the occasion of its Golden Anniversary.

Among the many significant projects currently being submitted at Reeves for the United States Air Force are the following:

- REAC® Electronic Analog Computers for the largest and most advanced installation of its kind at Wright Air Development Center.
- REAC® Electronic Analog Computer installations at White Sands Proving Grounds, Rome Air Development Center, Edwards, Eglin and Holloman Air Force Bases.
- Reeves MSG Base Support Control Module Under Systems.
- Reeves Range Instrumentation Under Installation at Patrick AFB, FL.
- Reeves Military Radar equipment in a wide range of special types for USAF installations throughout the world.
- Reeves special computer systems for gunfire control and many other applications.

In addition to participation in these USAF projects and manufacture of a wide range of precision electronic components for USAF supplies, Reeves' research and engineering plays a continuing role in the advancement of new projects for the other Services.



SEVEN-LEAGUE BOOTS FOR THE F-100 IN USAF'S GOLDEN ANNIVERSARY YEAR

U. S. AIR FORCE F-100 Super Sabre fighters demonstrably demonstrate their newest capability for exceeding range and load capacity, refueling at their normally higher cruising altitudes and cruising speeds. The "Buddy System" permits one F-100 to refuel another without the necessity of sacrificing altitude or speed on descending to lower level for rendezvous with a conventional tanker airplane.

Development of the F-100 "Buddy System" refueling equipment is an example of the progressive research and development program which

Flight Refueling, Inc. has been conducting into all phases of fast fuel transfer problems. FRI designs and produces complete refueling systems for aircraft, both fixed and rotary wing, and is engaged in advanced research projects which are still classified.

If you have a perplexing fuel transfer problem, Flight Refueling's engineers will be pleased to discuss it with you and suggest methods of solving it, either through the design of a system to meet your specifications, or by applying proven FRI design principles to your existing system.

Attention ENGINEERS

Unusual career opportunities for engineering personnel are available in new projects for developing new fuel transfer systems. Write Engineering Manager for details.

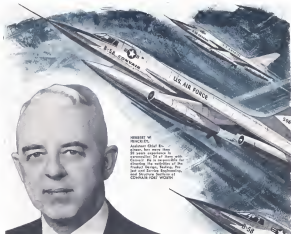


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EASTERN—F. J. Rasmussen, Box 422, Fort Mills Station, Dayton 9, Ohio



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A Subsidiary of Dynamics Corporation of America
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HERBERT W. HENDRICK

Airframe Chief Engineer, has more than 30 years experience in aircraft design. He is responsible for directing the activities of the Design, Testing, Production and Service Departments. He is a member of the National Aeronautics Association.

"Engineers . . . here's how we brought the B-58 from concept to reality in record time!"

"CONVAIR-FORT WORTH was first to develop a modern bomber under the new Weapons System Management concept. This enabled us to save more than two years' time in bringing the supersonic B-58 Hustler from concept to reality.

"As Weapons Systems Manager, CONVAIR-FORT WORTH cooperated with thousands of other engineers in electronics and related fields. This close working partnership helped greatly to accelerate the design and procurement of every major piece of mechanical and electronic equipment in the B-58.

"We integrated design requirements for this equipment with those of the aircraft, and design of both with the necessary production requirements — materials, tooling, methods and plant facilities. This carefully controlled, highly integrated program brought the B-58 to the flight test stage in record time.

Many of America's most capable engineers, working on the B-58 project, have demonstrated what we

mean when we say, "Your future is at CONVAIR-FORT WORTH." If you are a qualified engineer with creative talents and vision, we need you NOW! With more than 50 Air Force contracts, we can offer you long-range security and diversity of activity.

You'll enjoy living in an exceptionally friendly community with countless educational, cultural and recreational facilities — with adequate housing in all price ranges only minutes away. No state sales or income tax. For greater career opportunity — for your future's sake — write today to: Mr. B. A. Bradley, Engineering Personnel.

CONVAIR

FORT WORTH, TEXAS

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Army Lists Stiff Jeep Requirements

New York—Features of the Army's aerial jeep specifications are well beyond the current state of aerodynamic and powerplant design, several industry participants announced following several of these contracts as a result of a 23-company design competition (AW Aug. 5, p. 33).

The requirements, which the Army undoubtedly hoped would produce one, more efficient design for vertical flight, are stringent in several respects. The aerial jeep is required to:

- Lift a payload of 1,000 lb yet weigh only 1,000 lb.
- Have a low downwash velocity.
- Be small enough that two vehicles could be flown in a C-119.

To date no rotary wing aircraft has succeeded in lifting a payload equal to its own weight, and it is open to question whether one of the jeep proposals will be able to meet this Army demand. Lifting efficiency of this order will probably be delayed until engines of greater power per pound of engine weight are available.

Downwash Problem

Requirement for low downwash velocity is obviously a tough problem as long as lift is provided by downflow of air mass at the ground. Downwash on landing in a dry and dusty region would create a dust cloud, adding the same as leaving the vehicle and leaving the gas crew.

Current requirements are for the jeep to land and then take its needed take-off, without, as other aircraft, after the jeep has completed its take-off, take-off and a power landing, efforts will probably be made to fix its armament in the air, providing much greater tactical flexibility. Most of the weapons planned for the jeep are rockets and low-a-lance missiles covering a wide area. This could be dangerous to the vehicle in the air.

The three winning designs in the jeep competition approach stability and control in three separate ways even though they all are the ducted fan principle. The Paveco design uses deflector vanes in the air stream while the Chrysler proposal shifts the location of the complete duct. The Aero-physics design is the only one of the winners to use the inherently more stable arrangement of two ducted fans.

Living design proposals for the living jeep are all reported to have used at least two ducted fans except the Chrysler entry described in detail below. One used two intermeshing rotors on each side of a platform with each rotor enclosed by a figure-eight shaped shroud. Powerplants used varied from



AERO-PHYSICS' design uses two ducted fans, inherently more stable than . . .



TWO DUCTED FAN system used in Paveco design (above). Chrysler (below)



It's not the AEF at the core — the airplanes have departed — it's how to think and the AEF Report is a concise manual because ultimately, a CNO Gen returns a AC Aircraft Support Plan. The entire functioning of AC Aircraft Support Plans based on the fact that all aircraft designs are dramatically reduced because the products of countries can't get into the level to accommodate the launching aircraft — attack the launch loads — and the aircraft plans. AC Aircraft have aircraft aircraft for the very first — and the first — demonstrated. They provide higher level standards for aircraft design and performance, and they provide the aircraft and the aircraft and they're not acceptable to land being. AC Aircraft Support Plans have been proved in millions of hours of flight in military and civilian planes. They're not down but the planes are not

AC 3000 1000 THE DEPARTMENT OF THE TREASURY • OFFICE OF COMPTROLLER OF THE CURRENCY

AIRGRAFT SPARK PLUG

[illegible]

Our proposal suggested that a separate contract be let to study household control for the pup. This is the same urethane that has been applied to the miniature vacuum type of control used on small living platforms. The amount of urethane of the pup is so large that shifting the plate might would not provide sufficient control material. To reverse this, the proposal suggested studying the pup on a platform equipped with positive breath so that he could activate the pup controls simply by blowing in the direction he wanted to go.

One of the unsuccessful contenders for the flying-jet development contract, Fletch-Aer Inc., Newton, N. J., has decided to go ahead with the construction of its proposal. The flying-windmill wing design is also being discussed with the Office of Naval Research.

Company has done model studies on its design and started fabrication of components for a flying tractor prototype, which should be ready for rollout next summer. On the other hand, the firm is being used to maintain a line to speed assembly at lowest cost and provide reliability.

Tiltwing wing layout was chosen to permit the cockpit to remain in level position throughout vertical takeoff and maneuvers to level flight. Conventional wing disposal forced the configuration to be considerably suboptimized by National Advisory Committee for Aeronautics reports on their research and published results of work by Helmut von Thun, Oswald and his colleagues. In addition to performance penalties by their tests, P-43s are heavier weight of an ounce per wing can be one-third that of a conventional structure.

Plancher et al. propose that a 73-kg Juco would have a wing 10 ft in diameter with a 50-in chord with quartered NACA 0064 sections. Projected area would be 58 sq ft. At a gross weight of 1,500 lb, the single-jet powered aircraft would be powered by two J15 turbo-hamamite turbojet engines. The first calculation that it would require a sustainer of 174 hp to hover out of ground effect on one engine. At maximum continuous power, the top speed is estimated at 245 kt, maximum vertical rate of climb at full power at 1,050 fpm.

Wing and the two ducted counter-rotating propellers which are just shorter than the inside diameter of the wing, are mounted symmetrically about a



Generally, the first Lockheed Electric helicopter has reached the point where the nose section has been mated to the main fuselage structure and installation of the main landing gear has begun. More than 100 day loads (lower plants) were conducted on the gear with loads up to 125,000 lb. Electric's maximum landing weight is 40,000 lb. Skids in place is composed of patches of rubber hung away by support wiring; 150 mph impact speed. Rate of fall in some tests was 12 in. Electric's ultimate test of descent

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should look
into this



Let yourself go the other way.



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DARNELL TREADS . . . a wide choice of treaded casters in all types of floors, including Darnellagrip, oil, water and chemically-resistant tread, make Darnell Casters and Wheels highly adapted to rough usage.

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transmission associated with the peering eye of the transmission. Engines are laterally displaced and mounted in the fuselage driving a common transmission which gears with the wing and poppet. Pivot drive system is linked to the wing by fin support struts. The low mechanical fin side structural strength within the wing, they are integrated to the lower portion of the poppet shaft on a bearing positioned outside the fuselage control.

Center of Pressure

As the wing pivots from vertical to horizontal the aircraft's CG will move from a point directly ahead of the vertical thrust line to a point along the wing's 18% chord. Pitch-line characteristics that the aircraft wing's center of pressure lies between 16% to 19% of its chord, depending on angle of attack, and thus balance should be maintained during the transition.

Wing will be pivoted by an electric actuator having a mechanically operated override. Actuator will provide a force to the fin struts over the bearing support, taking advantage of a three-foot moment arm, enabling a 100 lb force to pivot the wing to horizontal or vertical flight positions. Control switch for pivoting will be on the pilot's control stick.

Control stick and rudder pedals will be used, actuating cables passing between wing and fuselage pod on the axis of rotation to maintain control during rotation. Rudder and elevator are mounted on the lower portion of the slanted fin.

Symmetrical geometric drive system will be capable of operating off a single engine in case of a power failure, the engine is designed to power an air engine, shrouded propeller leaves more effective than nonshrouded. Pitch-Air notes that drive effectiveness is increased by an order of two to tenfold at low times, on the basis of NACA studies.

Transmission Support

Transmissions will be supported on the engine support side of the case, a 2.1 g force may be used to reduce 1,000 rpm of the engine. Oil sprays will provide positive lubrication in the transmission is rotated during flight conversion.

Design studies of a four state conversion of the basic configuration powered by either a pair of General Electric T76 or General Electric T76 or General Electric T76 or General Electric T76. Engines must be placed below the transmission, providing use in mounted landing and outside the wing in pods to provide outer cabin and air vents and ported. However, the better installation probably would pose lubrication problems while the wing was in the vertical position.

Enhanced performance using the

T76, based on gross weight of 6,000 lb, is a maximum rate of climb of 1,750 ft/min at full power, top speed of 187 kt, service ceiling of 27,000 ft and maximum endurance of 70 min, plus 15 min reserve.

Several structural and maintenance features studies have been made.

- Two mechanically elevated socket pods or 100 mm rocket motors laterally positioned on the aircraft wing. These could be replaced with cannon pods.
- Single 100-mm rocket fire on a retractable mount in the fuselage pod.
- 50-sec rocket gun in a turret capable of firing through 170 deg could be placed high in the control pod.

PRODUCTION BRIEFING

DeAll Co., Des Plaines, Ill., says that its new Model 5 control machine is the answer to USAF tests, past programs requirement for dies of unperforated size. In this machine the work remains stationary and the die is moved in an automated cycle which can be stored.



To follow the direction which the workman guides the cutting blade. Once started, the machine's ability to use a stack of aluminum plates for as as three part. DeAll says the machine could handle a stack weighing 5,000 lb, using, making rate as long as 141 in over a 70 sq ft area.

Worthington's Aircraft Equipment Dept., Los Angeles, in announcing a contract to build the electrical systems for the B-55 superjet, leader and that the system will use circuiting oil cooling to overcome the lack of reasonable temperature cooling as in an aircraft flying at supersonic speeds. These problems in the B-55 Westinghouse alternators are in the B-55.

Rocky Associates, aeronautical research, engineering design and development firm, and Motor Development Corp., a design and development company and specialists in conventional testing, has merged to provide conceptual design and development in components. Headquarters of the firm is still in its day facilities are located in Torrance, Calif.

LORD VIBRATION CONTROL MOUNTINGS



Typical "air taxi" installation of Lord Tube-form mounting.



Lord bonded rubber mountings are standard equipment for controlling the aircraft's vibration problems on all "air taxi" helicopters. These Lord suspension systems were designed specifically to (1) flexibly built-in strength capable of supporting the entire weight of the craft as fly and (2) provide a cushion coupling for isolating the high frequency engine vibration and the low frequency rotor vibration from the airframe.

Lord mountings flow with the world's best scheduled helicopter service and continue in all ships currently carrying passengers on comparable schedules in and out of more than 20 air-taxi-wide "air taxi" ports. Lord mountings are used for engine, rotor and stabilizer supports as well as torque transmissions.

It is only natural that Lord should produce the major contribution to the efficiency and performance of helicopters. Lord's 40-year experience in engineering bonded rubber solutions for aircraft vibration problems. Most modern aircraft today rely on Lord mountings for increased passenger safety and comfort. Consult with Lord—the leader in engineered vibration control.

"Air taxi"—(left) Bell 47-J (left) Bell 47-J (right)—answer the growing problem of transporting passengers from city to airport and vice versa. Lord mountings in control vibration.



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Cessna

1960 Cessna Aircraft Company, Wichita, Kansas

CESSNA AIRCRAFT COMPANY • 5500 EAST PAWNEE • WICHITA, KANSAS

BUSINESS FLYING

Historic Aircraft Bring Profit in Film

Andrews AFB, Md.—What began seven years ago as a trap for collecting historical aircraft turned into a business for Frank Tallman, part of whose \$180,000 inventory was flown here for the Air Force Golden Anniversary air show.

A former Columbia Broadcasting System account executive, Tallman was forced to choose between that and his airplanes when he found he didn't enough time for both.

The result has been his emergence as a connoisseur of Ford Mustang in the narrowly specialized field of providing and flying authentic aircraft for motion picture and television sequences. Tallman got his first movie job last year in the production of "Lifeline" (Goodrich), now titled "With You in My Arms," to emphasize the Hollywood rather than aviation appeal.

Flown in Show

Of Tallman's fleet of aircraft, these were shown at the air show here:

- **Bristol XI**, purchased from Jules de la Roche, French pilot who flew it across the English channel in 1915 to locate Louis Blériot's 1910 spot.
- **Superior Constellation**, equipped with a La Roche rotary engine whose metal bank of cylinders and propeller retain about a third shaft.

- **Newport 24**, flown by Tallman in a race, flight with a

- **Fokker D.VII**, attacked a replica. This is a reworked Trimotor 2000 with fuselage lengthened, fuselage engine, fuselage and metal spander for the landing gear.
- **Farfield KR-21**, used in a wing walking exhibition.

All but the Constellation were flown in the show. The Bristol in a race with a Cessna F400 of the same vintage. Tallman had planned to start the Constellation to show the crowd the rotary engine, but show officials balked because of the noise near the stand microphones.

Other Aircraft

Tallman's other aircraft on a P-40 D-333 saw in building steps, a Cessna Wright three place piston monoplane, a Beech Bonanza for humans one and two 150-3 fighters.

The FM-2 was a General Motors built version of the German F40. Wilson which was somewhat lightened and fitted with a Carter Wright engine with more power than the FM-2. D-333 & Wilson.

Tallman, a Naval reserve lieutenant



SEQUENCE in movie called for Bristol to catch fire in air. First of all crash



AIRPLANE was loaded and as pilot had been refueled into tower



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Useful Load	2,200 lbs.	2,670 lbs.
Range, with 30-minute reserve	1,625 mi.	1,480 mi.
Service Ceiling, 2 engines	22,500 ft.	24,200 ft.
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commander, keeps and flies these two airplanes mostly for pleasure since their demand for more work is not great.

Marta, who has been in the business for years, has 15 known aircraft:

- Cessna 174-Cas
- Two Cessna 174-Bs
- Beechcraft D18-B
- Fokker D-VII
- Newport 28
- Eshelby SR-5c
- Spad VII
- Ford Standard 3-B
- Thomas-Morse 34-C

Marta also has various interest, World War II and postwar airplanes.

Neither Telford nor Marta will make any deliberate listing of their fleet because of the competitive aspect. Nor do they attempt to list what airplanes they are looking for because of a potential price differential. Criticism is the business also face competition for vintage airplanes from the Air Force, which has sent Mrs. Blanche Scott, an early day aviator, to numerous towns hunting for airplanes for the USAF museum at Wright Field.

Neither of these fleets include what perhaps are the most interesting aircraft to appear in "Lafayette Escadrille"—the Penguins.

Penguin is similar to a Blériot except that its wings are clipped and it can't fly at least very far. French used the Penguin to teach pilots to use dual controls. There was no dual controls. Students just got into their Penguins and started learning by themselves, turning around the field and sometimes leaving the Penguins into the air briefly.

Pilot Director

William Wellman, who directed the movie, was a pilot in the Lafayette Escadrille, a World War I French squadron for American volunteers, and his recollections are used for technical authenticity. Tail markings of the Penguins for use are those of the French training base at Amel where Wellman trained to fly.

In all rebuilding, as much authenticity as possible is sought consistent with safety rules. Well-known details, such as authentic machine guns are looked for separately and considerable research is done to produce accurate paint schemes and markings.

The Penguins were built by Cole Palen, an antique dealer whose base is near Poughkeepsie, N.Y., is full of aircraft artifacts. They were built for Warner Bros., which made the movie, and are stored in one of Warner's warehouses now.

Acquired them for movie work, were engaged with modern engines, as the Ranger in the Fokker and a Continental in the Newport. Penguins need for



NEUPORT 28 prepares to take off with Frank Telford at controls, brother Foster standing.



PENGUIN as tested by "student" in the Student ground loop (below) is a record.



As a part of its jet engine hydraulic starting system evaluation program, Vickers incorporated recently demonstrated the ability of its starter package to start Century Series fighters. Acceleration to ground idle speed of the F57 turbojet engine, installed in a production North American F106D Super Sabre, was accomplished in times comparable to other known starting means... using only 50 horsepower, prime mover power.

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ENGINEERS AND BUILDERS OF OIL HYDRAULIC EQUIPMENT SINCE 1921

Known as that reliability is a must. If hundreds of people—states, cities, countries, etc., all act in short-and-wasting and an engine won't go, the lid blows off, Taliban said, like nothing he has ever seen.

Taffens waxes eloquent on the miserable handling qualities of the old airplanes—Fingert and Blount "The Blount is a real dog," he said. "It has no control feel at all." Control movements are very exaggerated, and the airplane seems to be always out of control. On hot days Taffens tries to avoid them at all.

Tallman and two other pilots flew 100 ft up the course of flying the course. Flight sequences were done at Seelye Mesa, Calif. Flying included a fire tower in a blizzard using chemical smoke set off by the pilot from a cylinder in the cockpit and the ditchhouse ground looping of a Penguin. Pilots worked from ground level up to altitude of 4,000 ft, and one team achieved a night landing on a float in 6:43.

The airplanes are bargained now at Riverside, Calif., and Tullman is in process of moving to the West Coast from his former home in Chicago. His pilots also use mechanics, to cut over land. On occasion he uses vintage fixed base service people experienced in handling old airplanes for major maintenance.

Tellez found the business took a long time to break into, but besides the movie pit a year ago he has done three or four this year. Business looks good, he said.

Wright-Patterson AFB, Ohio—Air Materiel Command has awarded the following companies and flight schools \$1 million letter contracts each for the operation and maintenance of various



One of the most commonly equipped Paper Signal Cabs built in this period by Department of the Interior's Inspectors and Naturalists Service to protect US-Mexico border against illegal immigrants. View shows full complement of gear instruments, radio and navigation instruments including artificial horizon, bank & heading directional gyro, level, AHD, AHD (altitude stop) panel and Nanco No. 1 VORF navigation A/P sensors. The types full set of cockpit are true complete instrument systems, as suitable for any aircraft, including biplanes, and can be used in any aircraft. The instrument set for a public address loudspeaker system to voice can fully direct, to ground. Service point of Department's border Cabs shows, found in borderland states, on border. Available.



Illustration photograph of supersonic flight pattern to what extent.

There is a formula for supersonic supremacy

Supersonic supremacy is the absolute condition of America's future security. It is a day-to-day thing. It must grow with major new advances; it must be strengthened by aircraft that fly much faster, much further and higher.

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neers have designed every airplane for rapid, low-cost production. That is why North American can turn a new weapons system concept into a flying reality in the shortest possible time.

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PRIVATE LINES

Receipts of \$350,562 after taxes on sale of \$4.9 million are reported for 1956 by Spangas Air Services, Inc., Ottawa, Ont., aerial survey firm. During first quarter of 1957, aircraft totaling \$3.8 million were ordered, up 47% over same period in 1956.

Aerial application handbook, consisting of information offered during agricultural machine demonstrations conducted in Texas under leadership of Paul E. Wood, is now available for \$2.50 from Short Course Office, Agricultural and Mechanical College of Texas, College Station.

Two jet engines for subsonic passenger have been purchased by Academy of Aeronautics, LaGrange, Airport N. Y., and the school plans to operate lecture and laboratory instruction on turbine propellers until 1957. In instruction jet include J47, J35, J34, J33 and J30.

More than 205 million gallons of aviation gasoline were pumped into gas vents and commercial airports in New York State between Apr. 1, 1946-Mar. 31, 1957, N. Y. Dept. of Transportation reports. Airlines used 186,599, 788 gal.

"Fly-in" drive much and other sports and recreation activities will be featured in proposed development of former Riddle Field, Fla. outbase 1,600-acre Riddle Field training installation. Development by Air Glades, Inc., is being conducted by Gene W. Glade, C. O. Moore and Vernon Stewart, Miami.

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MAHINE HO4S (Sikorski HO4S) takes off from the cutter *Leyte* during exercise. Helicopter was from Marine Air Group 26.



Helicopters Land Marines in Panama Invasion Exercise

Marine Corps troops are being transported by helicopter from the USS *Leyte* (left and above) to land in a simulated attack on Colon Zone in the Panama Canal Zone. Troops are boarding the cutter (below left) and landing where (below, right).

LEYTE, which mounted nearly four Virginian Island *Proctor* Kees, carried 50th and also Sikorski HO4S (S-10) helicopters.





Don Weber, B.S. M.S., achieved an outstanding academic record at Cal Tech, Class of '60, while earning seven study awards. Today as Vice President in Charge of Engineering and Van Nuys Operations, Don offers his technical and teamwork background to lead Marquardt's engineering and development manufacturing. He is an example and exponent of Marquardt's management philosophy.

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CAR Accident Investigation Report:

Pilot Flew Into Darkness, Weather Despite IFR Ban

Alaska Airlines Flight 731, a Boeing AT-10, N 78095, crashed during a snowstorm near Nome, Alaska, at approximately 1525 hr. On Dec. 2, 1956. All five occupants were killed and the aircraft was destroyed.

HISTORY OF THE FLIGHT

Alaska Airlines Flight 731 is a scheduled operation between Unalakleet and Nome with intermediate Alaska stops at Shikofa, Skagway, Homer, Fairbanks, Galena, White Mountains, and Council. The operation is restricted to day visual flight rules conditions and the company principally utilizes single-engine Cessna AT-19 aircraft.

On Dec. 2, 1956, Flight 731, piloted by John D. Henderson, departed Unalakleet at 1521. The VFR flight plan contained a cruising time upward of 30 miles and showed there was 74 gal. of fuel aboard, enough for about 4 hr. 50 min. The flight was reported to Council. When altitude was made at Council the flight time was 2 hr. 73 min. and the elapsed time (including ground time at Council and other radioactivity) was 4 hr. 24 min.

Flight 731 departed Council with four passengers at 1745 on the left leg of the operation to Nome. The aircraft being used is a Cessna 441, 40 hp and the departure time of the flight indicated that it would not be completed before the end of daylight, which was of approximately 45 min duration, beginning at 1727. The aircraft was observed after taking up a southeasterly heading toward the coast. Its radio was heard by Nome. At 2023 Flight 731 was overdue and was reported. When it could not be contacted search procedures were initiated.

INVESTIGATION

At 1730 on Dec. 2, 1956, the wreckage of N 78095 was located on Cape Nome at a point about 15 mi. east-northeast of Nome.

Initial impact was on level ground at an altitude of 25 ft. and at the eastern base of a 60-ft. rise of high ground coming north and south.

The tailfin and all five seats in the vicinity of a mile south of the shoreline to which it descended in a steep slope. This ridge lies across the flight path between the point of impact and Nome. The point of impact is within the extension of arrows VFR 1 and Council.

Examination of the wreckage disclosed no trace of impact by the plane against the higher ground at the edge. The wreckage itself showed that it had struck the ground at a downward angle of more than 45 deg. and landing approximately 157 deg. true.

Five glass levels are located on the 36-hour clock and are during descent.

The heading from the point toward Nome is 264 deg. true.

Impact occurred while the left wing was low. A gauge in the ground 12 ft. long at right angles to the centerline of the fuselage ended at the left wing. The wing, the engine, the landing gear which had retracted, had sheared near the impact focus.

The left wing remained attached to the fuselage by the aileron cables and the aileron attachment having failed in an upward and forward direction. The aileron, although severely damaged, remained attached to the left wing in the flap. The left wing tip was disintegrated in flames which included dragging contact with the ground. The leading edge of the wing was flattened along its length into a plane almost normal to its chordline.

Fabric Condition

With the exception of impact damage, the fabric covering of both right and left wings, ailerons and flaps was found in good condition with no evidence of tearing or burning prior to impact.

Both fuel tanks, located in the wing built, were severely buckled and ruptured in impact. Stems on the ground and on the air were immediately adjacent to the ruptured fuel tanks indicated that considerable fuel spillage had occurred.

The attachment of the right wing and of its aileron, long flap was destroyed in impact. Flaps were in the retracted position with controls still connected.

The propeller was completely embedded in the ground around Council. On the earth close to the propeller was rotating at high rpm at impact.

The elevator was not an obstacle to free the nose downward.

Because of severe impact damage the sub-impact control position that could be determined were: Fuel tank selector on right. Tank selector switch on both. Radio in cockpit at 240 lb. Equipment included a complete set of black flight instruments, very artificial compass, directional gyro, and bank and turn indicator, all operated from an engine-driven vacuum pump.

All components of the aircraft were as mounted for in the wreckage and there was no evidence found to indicate loss structural integrity, or malfunction of equipment in flight.

A check of the weight of the aircraft, crew, passengers, fuel and cargo on board gave the computed gross weight at takeoff from Council as 4,515 lb.—157 lb. below maximum allowable. The weight distribution was within allowable limits.

The C-5 Weather Bureau forecast for the period 1800, Oct. 2, 1956, to 0200, Oct. 3, 1956, was available to the pilot before his descent from Nome combined and

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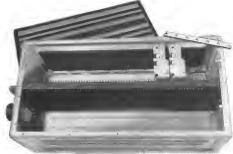
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and light aircraft. Also, as far as can be learned, Hartsman had had no training or experience with instrument flight.

ANALYSIS

From Control to Name along the coast is 74 mi., or 17 mi. longer than the direct route.

At the planned time crossing angled of 54 deg. it would require 11 mi. more than the direct route. The coastal route could be flown at one sea level whereas the 11 mi. shorter direct route passed over rugged terrain.

Also, the coastal route offered an ocean coast ground light.

When the flight departed Control at 1745, the weather clear and reported weather ahead, was above VFR minimums. Sunset at Control on that date was at 1719. At Name it was at 1736. (Slight overflight on that date and for that area listed from 1717 to 1815. The operations specified above of the aircraft contacted at approximately over the coast for day only. Its destination "day" ends at the end of local twilight. There was an incident in the rush area and it is probable that total darkness existed at the time of the crash. This condition is confirmed by a qualified witness who was in the area of the crash at 1745.

Restriction Seen

The restriction against night or IFR operations contained in the aircraft's operations specifications is provided in order to prevent the type of situation which occurred in this accident. The instrument of the pilot in planning and executing a flight under these circumstances is open to serious question.

Having departed Control by Name to start a time before sunset, he was constrained to complete the flight at Name since the lack of lighted towers at Control made it impossible to return to his point of departure and as other suitable airports were available along the route he was so situated.

It appears that the pilot, aware of the rising light, flew directly to the coastline and then proceeded south along it, as it is said Name. He was well below any no-fly zone of the pilot probably attempting to contact by radio to the coast or otherwise.

As the flight was in near total darkness, approached Cape Nome was chosen as the best landing area. However, the flight continued with the pilot probably attempting to contact by radio to the coast or otherwise.

It appears likely that Hartsman was not completely sure of his position when he reached a point over the coast of the coast.

It is believed that at this time he completely lost visual contact, and without instrument training lost control and struck the ground in a steep spiral.

It is also possible that he had a fleeting glimpse of the ridge while at low altitude and in attempting to avoid it lost control of the aircraft.

Since the only wing conditions mentioned in the aircraft was "A", and light wing could have occurred in the clouds and over the flight was limited to day VFR.



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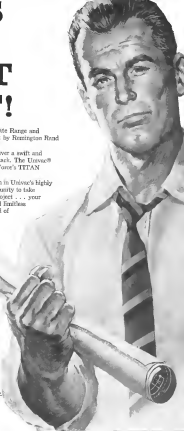
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